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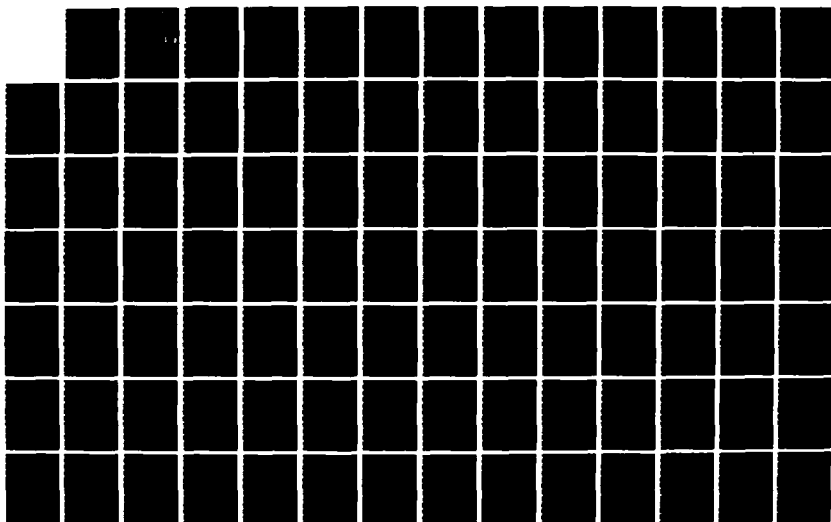
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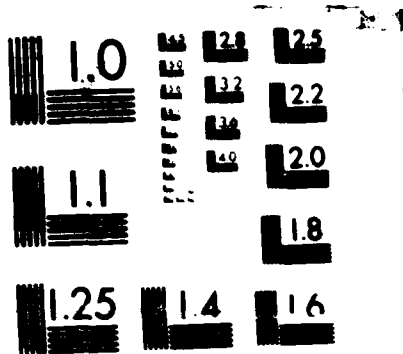
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LASER BALLISTIC
SENSOR DEVELOPMENT

BOEING AEROSPACE COMPANY
MS 8C-64
PO BOX 3999
SEATTLE, WASHINGTON 98124

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JANUARY, 1987

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| <p>A prototype instrument to determine the orientation of a projectile in free flight (i.e., roll, pitch and yaw angles) was developed at Boeing and used to perform tests at the U. S. Army Ballistic Research Laboratory (BRL) at Aberdeen Proving Ground, MD. The instrument incorporated two visible gas lasers (to provide a two color collimated illuminating beam) and a pair of galvanometer beam deflectors to generate a circular angular dither of the output beam. All three angles can be determined from the detected light reflected by a retroreflector/hologram combination mounted on the projectile. The prototype instrument was designed to make three-axis angle measurements on in-flight or in-bore spinning or non-spinning projectiles, and to make two-axis angle measurements of gun tube motion.</p> <p>Tests were performed by both Boeing and BRL personnel at the Aerodynamics Range at BRL during a two-week period in August 1985. The test instrumentation was then left at BRL for additional testing by BRL personnel. A summary of the test results follows:</p> | | | | | |
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19. Abstract (continued)

- 1) All three angles of a spinning projectile can be measured with a fixed beam and two detectors, where one detector senses fringe amplitude and the other detector senses roll,
- 2) The dither rate (3 kilohertz) was too low for the non-spinning projectiles tests,
- 3) No in-bore data was obtained due to obturation,
- 4) Dynamic measurements of both angle and position of a gun tube were demonstrated.

As a result of the test, the following problem areas have been identified:

- 1) The test results show that a system with a higher dither frequency must be developed,
- 2) The use of two independent lasers complicate the alignment procedure. It would be more reasonable to use a two-color laser (e.g. argon ion laser),
- 3) Some of the data contains an intensity variation at the fundamental of the dither frequency. This could be due to the projectile lying at the edge of the "overlap" of the dithered beams. A larger beam should eliminate this problem and simplify the alignment procedure,
- 4) The reflection coefficient versus angle was assumed to be sinusoidal for the present data reduction algorithm. In practice, the reflection coefficient is not sinusoidal, and the shape of the function changes with relative humidity and laser wavelength. Additional development is needed on an algorithm based on a more realistic analytic model.

ACKNOWLEDGEMENTS

We would like to acknowledge the excellent support by Mr. Donald McClellan, Mr. John Carnahan and Mr. William Thompson (range personnel) during this program. The technical results obtained were due in large part to the cooperation and assistance we received.



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SUMMARY

The prototype instrument to determine the orientation of a projectile in free flight (i. e. roll, pitch and yaw angles) was developed at Boeing and used to perform tests at the U.S. Army Ballistic Research Laboratory (BRL) at Aberdeen Proving Ground, MD. The instrument incorporated two visible gas lasers (to provide a two color collimated illuminating beam) and a pair of galvanometer beam deflectors to generate a circular angular dither of the output beam. All three angles can be determined from the detected light reflected by a retroreflector/hologram combination mounted on the projectile. The prototype instrument was designed to make three-axis angle measurements on in-flight or in-bore spinning or non-spinning projectiles, and to make two-axis angle measurements of gun tube motion.

Tests were performed by both Boeing and BRL personnel at the Aerodynamics Range at BRL during a two week period in August, 1985. The test instrumentation was then left at BRL for additional testing by BRL personnel. A summary of the test results follows: 1) All three angles of a spinning projectile can be measured with a fixed beam and two detectors, where one detector senses fringe amplitude and the other detector senses roll, 2) The dither rate (3 kilohertz) was too low for the non-spinning projectiles tests, 3) No in-bore data was obtained due to obturation, 4) Dynamic measurements of both angle and position of a gun tube were demonstrated.

As a result of the test, the following problem areas have been identified: 1) The test results show that a system with a higher dither frequency must be developed, 2) The use of two independent lasers complicate the alignment procedure. It would be more reasonable to use a two color laser (e.g. argon ion laser), 3) Some of the data contains an intensity variation at the fundamental of the dither frequency. This could be due to the projectile lying at the edge of the "overlap" of the dithered beams. A larger beam should eliminate this problem and simplify the alignment procedure, 4) The reflection coefficient versus angle was assumed to be sinusoidal for the present data reduction algorithm. In practice, the reflection coefficient is not sinusoidal, and the shape of the function changes with relative humidity and laser wavelength. Additional development is needed on an algorithm based on a more realistic analytic model.

INTRODUCTION

This document is the final report on Phase 1 of contract DAAK11-84-C-0095. The contract is the first part of a program whose goal is to develop hardware for measuring the angular orientation of projectiles in-flight and in-bore, and the angular response of gun tubes during firing. The technology, originally developed for measuring the pitch angle of models in the Boeing Transonic Wind Tunnel, was shown to be applicable to spinning projectile yaw angle measurement in a prior test program (Reference 2) at the U.S. Army Aerodynamics Range. Significant improvements to the measurement techniques were made in the present contract.

A two-laser breadboard system was assembled at Boeing for evaluation at the Ballistic Research Laboratory. The breadboard was designed to make 3-axis angle measurements on in-flight and in-bore non-spinning or spin-stabilized projectiles, and 2-axis angle measurements on gun tubes. For spin-stabilized projectiles, three-axis angle measurements were demonstrated. Signals were obtained from non-spinning projectiles, but could not be converted into meaningful information due to the inability of the breadboard optical system to measure the orientation of projectiles possessing high yawing rates. No in-bore data was obtained due to degradation of the incident laser beam quality resulting from the leakage of hot combustion gases around the projectile. Dynamic measurement of both angle and position were demonstrated for the gun tube tests.

The report contains a description of the breadboard system, the modifications required for the various measurement tasks, the test conditions, and the test results. Data reduction algorithms developed by Boeing are in Appendix B. Preliminary design information for Phase II hardware, based on the results of this evaluation, are included.

OPTICAL TEST SET-UP

The test setup consists of an optical unit and supporting electronics located in the range together with data acquisition and processing equipment placed in the control room. The optics will be discussed first.

Figure 1 shows the arrangement of optical elements on a standard 0.6x1.2 meter setup board. Red and blue lasers are used to provide orthogonal angle measurements on the projectile. The beams are combined in a dichroic cube beamsplitter and then directed to the filter-splitter. A lens in the red beam is used to adjust the beam diameter at the filter-splitter so that the red and blue beams have the same diameter when they exit the unit.

The filter-splitter is a Boeing device (U. S. Patent No. 4,329,059) that spatially filters the outgoing beam while transmitting most of the return beam energy to the photodetectors. The coaxial red and blue beams at the input to the filter-splitter are focused by a microscope objective lens onto a small elliptical mirror on a tilted glass plate which spatially filters the reflected beams and insures that the red and blue beams are collinear. The elliptical mirror is small compared to the diffraction pattern of the return beams from the reflector on the projectile, so more than 90% of the return energy passes on to the photodetectors.

A two-axis galvanometer type beam deflector generates a circular dither (or conical scan). The relay lens forms an image of the filter-combiner mirror in the focal plane of the large collimating lens. The function of the relay lens is to reduce lateral motion of the spot at the projectile. (The translation is zero in the plane where the combination of the collimating and relay lenses forms an image of a galvanometer mirror.) The output beam is collimated and reflected into the projectile path with a pair of mirrors (not shown).

The reflector assembly consists of a hologram and a cube corner reflector, as described in Reference 2, with modifications that allow 3-axis angle measurement. The hologram has a central circular holographic grating surrounded by an annular grating with equal area but orthogonal fringe orientation. Red and blue spectral filters that match the hologram profiles are sandwiched between the hologram and the reflector.

The circular dither on the outgoing beam produces orthogonal sinusoidal components to match the axes of the dual hologram. Amplitude and phase of a pair of dither frequency harmonics (for each color) are used to find the orientation of the projectile body axis. Roll about the body is proportional to the phase difference between the deflector driver and the fundamental component of the return signal.

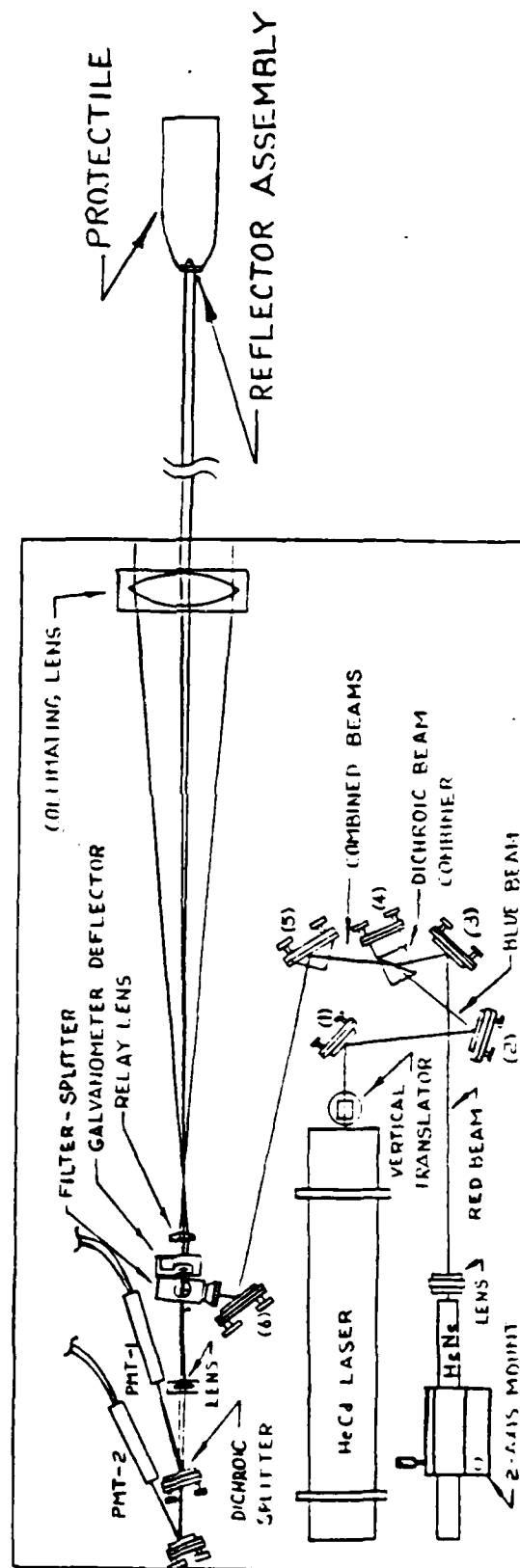


Figure 1. BRL Optical Breadboard.

DATA ACQUISITION ELECTRONICS

Figure 2 is a block diagram of the data Acquisition Electronics (DAE). The DAE was used to collect range or calibration data. The photodetectors were Hamamatsu R761 photomultipliers (PMT), a head-on type with a half inch (13mm) diameter photocathode. The PMT has ten dynode stages, typical current gain of 5.8×10^5 , and a spectral response from 185nm to 850nm, peaking at 420nm. The light entering the PMT's was optically filtered for the two optical wavelengths before application to the PMT's.

Both the red and blue signal channels are identical with the exception of the overall gain. Load resistors for the PMT's were 10k ohms. At the optical assembly each PMT output was amplified by a HP465A amplifier with a gain of 20db and filtered by a Krohn-Hite 3342 filter/amplifier. One section of the Krohn-Hite filter is used as a low pass filter with a cutoff frequency of 100Hz and a gain of 20db. The low pass section drives the high pass section resulting in a bandpass with cutoff frequencies of 100Hz and 24KHz. The high pass section of the Krohn-hite filter drives approximately 100 feet of terminated coax cable between the optical assembly and the control room. The variable gain capability of the HP467A was used to adjust the input amplitude to the tape recorder and analog-to-digital converter to approximately one volt RMS.

All timing for the system is synchronously derived from a very stable frequency source, an HP8460B signal generator operating at 576Khz. Quadrature (i.e. the two signals are out of phase by 90 degrees) signals at 3Khz drive the elevation and azimuth galvanometers.

An EMI7000 tape recorder was used to record all test data. Each signal output from the HP467A's was applied to three tape recorder channels and one channel of the ADC/Computer Interface. The gain settings of the three tape recorder channels were adjusted to accommodate signals that were the expected nominal and plus and minus 50% from nominal. Other sources of data recorded by the EMI7000 are shown in Figure 2.

The direct memory access, DMA, of the HP9836 computer was used as a transient waveform recorder to collect real time range and calibration data for immediate and future data reduction. All four channels of the ADC/Computer Interface are sampled synchronously at a 36Khz rate and data transferred to the computer by DMA at 144 kilobytes.

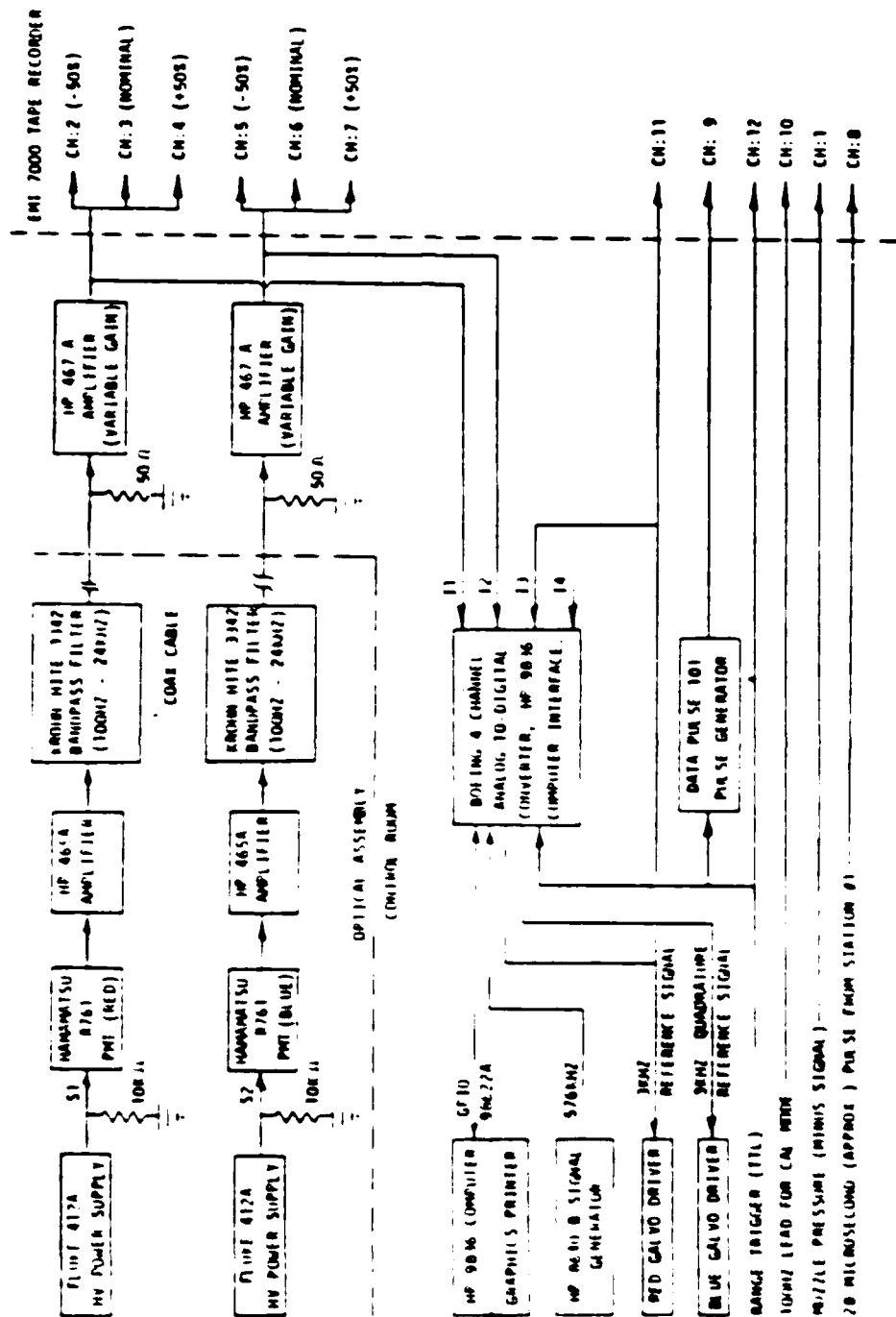


Figure 2. Block Diagram of Data Acquisition Electronics.

Initiation of range data collection was synchronized to the first positive crossing of the 3Khz signal driving the red galvanometer after a range trigger is received. This synchronization was incorporated to facilitate the phase measurement required in processing the data. When range data was collected, the red, blue, and 3Khz reference signals were sampled and stored by the computer.

Each hologram/retroreflector combination was calibrated by placing it in a test fixture which allowed it to be rotated in two orthogonal planes, simulating pitching and yawing motion. During this procedure the red and blue signals and the x and y positions of the gimbal positioning device were also sampled and stored by the computer. Appendix A contains the schematics for the data acquisition electronics.

DATA PROCESSING

Since data processing is dependent on the characteristics of the holographic grating and corner reflector combination, the reflector assembly will be described first.

The reflector assembly, Figure 3, consists of a holographic grating and a cube corner reflector. The reflector assembly is small and rugged with typical dimension of 6.4 mm diameter and 5 mm high. The input light beam is diffracted by the hologram into the zero and higher order beams. For clarity, only the zero order and one of the signed first order beams are shown in the figure. These two beams are retroreflected by the cube corner and again diffracted by the hologram. Note that the zero and first order beams are again diffracted into zero order and first order beams. The emerging composite zero order consists of a number of parallel component beams. These component beams follow different optical paths, consequently they interfere constructively or destructively, depending on relative phase difference. Figure 4 shows the modulation of the reflection coefficient due to optical interference as a function of the angle of incidence.

The resulting fringe angle of the hologram corner cube combination is dependent on the fringe angle of the hologram when it is produced, the index of refraction and height of the corner cube reflector. Figure 5 is a diagram showing the optical set up for producing the hologram. The diffraction angle for the hologram is

$$\theta_a = 2 \sin^{-1}(\lambda/2s)$$

where λ is the wavelength of the laser source and s is the grating period.

The fringe angle for the hologram and corner cube reflector combination is approximately

$$\theta_r = n\lambda/2h\theta_a$$

where h is the height and n is the index of refraction of the cube corner reflector.

If the change in angle of the retroreflector is unidirectional then the amplitude of the reflection coefficient could be used to measure the angle of the retroreflector. In reality the amplitude does change due to effects other than changes in the angle of the cube corner reflector (e.g. noise, changes in laser power output). To determine the direction of change of the retroreflector angle, quadrature signals are required. With quadrature

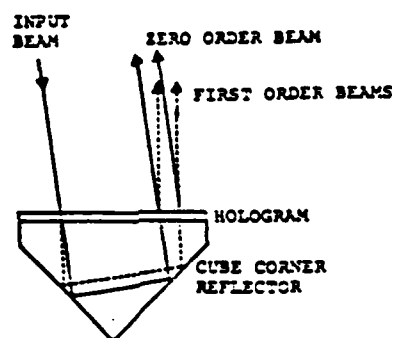


Figure 3. Reflector Assembly.

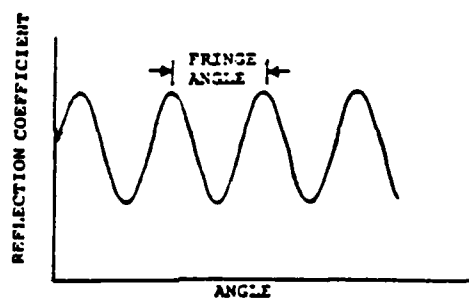


Figure 4. Reflection Coefficient.

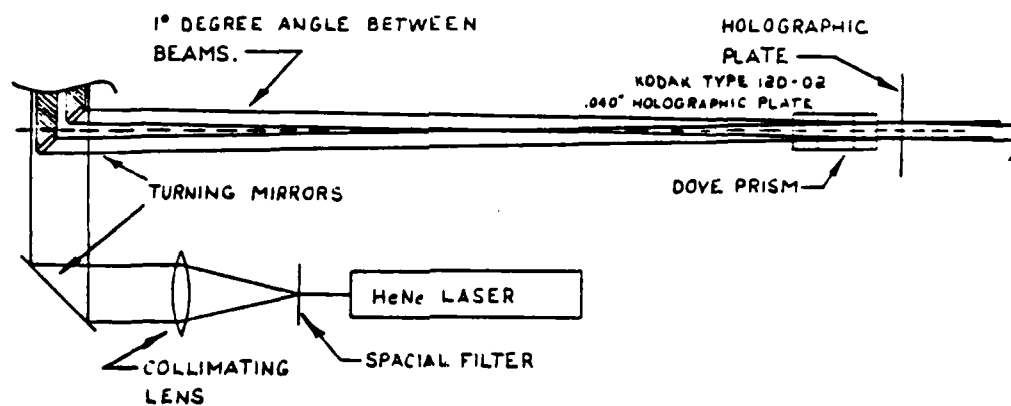


Figure 5. Technique used in making a 1° Diffraction Angle Hologram.

(i.e. signals are out of phase by 90°) signals phase can be measured, direction of phase change determined, and the amplitude effects which are not related to angle can be eliminated.

The spin of a spinning projectile provides the modulation which enables data reduction. For a slow spin or non-spinning projectile the laser beam is dithered (angle modulated) in order to change the incident angle of the input light beam at the retroreflector. This is accomplished by the azimuth and elevation galvanometers and beam directing optics.

Angle modulation of the beam across the fringe pattern shown in Figure 4 results in quadrature relationships existing between frequency components of the photodetector current which are harmonically related to the modulating frequency. Figures 6 through 8 support the analytical description to follow from which the software algorithm was developed. This analytical description is for one axis only and is not as rigorous as the description for the analytic model which was developed at the end of the program.

Figure 6a represents the photodetector output (equivalent to the reflection coefficient) and Figure 6b the angle modulation at the dither frequency. Figure 6c is the photocurrent for the angle modulation shown in Figure 6b. I is the photodetector current output as a function of the angle of the corner cube reflector and $\theta_r = 2\pi$ (yaw angle)/(fringe angle). I_p is one half the peak-to-peak value of the photodetector current and t is time. m_p is the peak deviation of the angle modulation in radians and ω_m the modulation frequency of the dither in radians per second.

The equations that describe the modulation effects of the dither are

$$\begin{aligned} I(\theta_r, t) &= I_p \sin(\theta_r + m_p \sin \omega_m t) \\ &= I_p \sin \theta_r \cos(m_p \sin \omega_m t) + \\ &\quad I_p \cos \theta_r \sin(m_p \sin \omega_m t) \\ &= I_p \sin \theta_r [J_0(m_p) + 2J_2(m_p)\cos 2 \omega_m t + \\ &\quad 2J_4(m_p)\cos 4\omega_m t + \dots] + \\ &\quad I_p \cos \theta_r [2J_1(m_p)\sin \omega_m t + 2J_3(m_p)\sin 3\omega_m t + \dots] \end{aligned}$$

where the values of $J_n(m_p)$ are determined from the graph of Bessel Functions, Figure 7. Note that the amplitudes of the odd and even harmonics above are quadrature functions. The amplitude of the odd harmonics is proportioned to the cosine of θ_r and the amplitude of the even harmonics is proportional

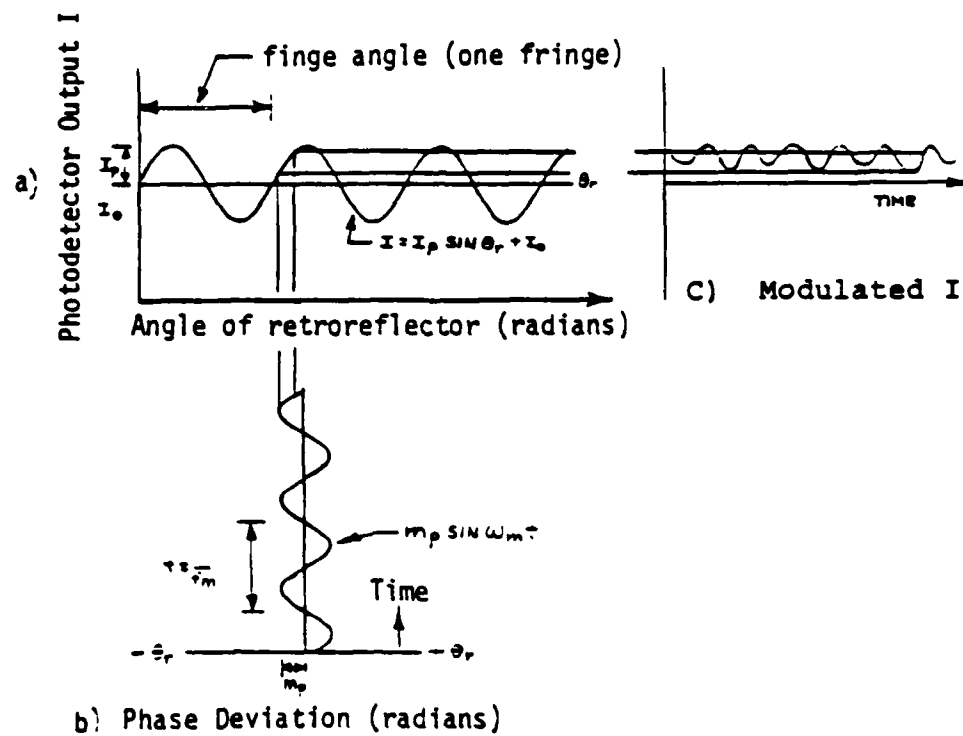


Figure 6. Angle Modulation of the Fringe Angle.

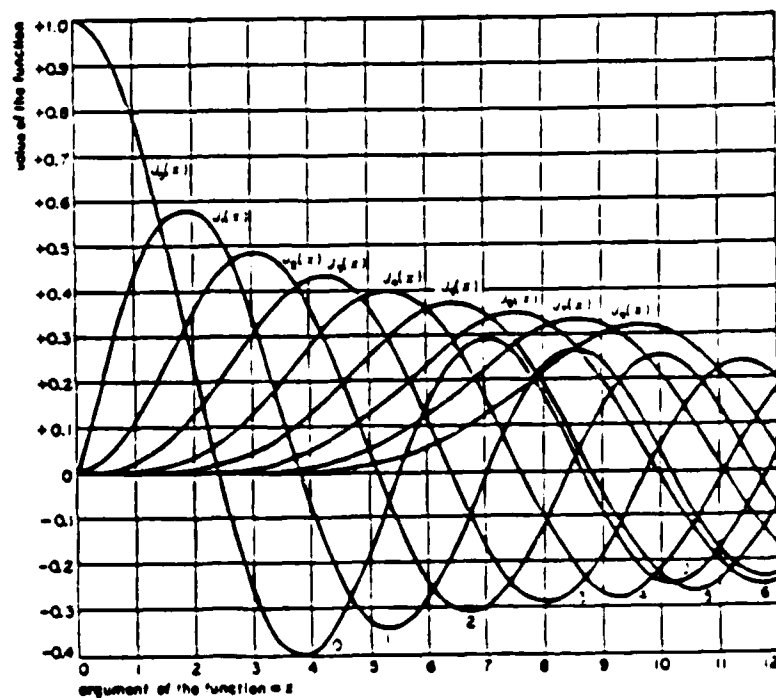


Figure 7. Bessel Functions for the first 8 orders.

to the sine of the θ_r . The fundamental and second harmonic were used in the breadboard system.

Figure 8a is the photodetector output with a graphic representation showing the relation between the fundamental and second harmonic components of a reflected beam for a small value of m_p . The figure shows the relationship between the reflected signal and the dither signal, when $\theta_r = 0, \pi, 2\pi, \dots$. The fundamental frequency dominates the signal when $\theta_r = 0, \pi, 2\pi, \dots$. The second harmonic dominates the signal when $\theta_r = \pi/2, 3\pi/2, 5\pi/2, \dots$. Note the π radian phase reversal between the positive and negative slopes for the fundamental and the positive and negative peaks for the second harmonic. Figure 8b and c are the fundamental and second harmonic filtered from the photodetector current by bandpass filters. The amplitude factor m_p is chosen equal to approximately .27 resulting in equal peak-to-peak amplitudes for the fundamental and second harmonic. It is important to note that the waveforms of the fundamental and second harmonic are in quadrature.

Figure 8d shows the desired quadrature terms as a function of cube corner reflector angle. To arrive at these outputs a phase comparison must be made between the fundamental and the modulating frequency for the fundamental, and between the second harmonic and twice the modulating frequency for the second harmonic. Devices such as balanced mixers used as a phase detector, followed by a suitable low pass filter, would produce the desired output. For the computer algorithm the envelope amplitude and sign is determined for each cycle of reference phase.

After amplitude and phase detection the quadrature signals of Figure 8c and d can be written as

$$E_1(\theta_r) = I_p(2J_1(m_p))\cos \theta_r$$

$$E_2(\theta_r) = I_p(2J_2(m_p))\sin \theta_r$$

where $E_1(\theta_r)$ is the detected envelope of the fundamental and $E_2(\theta_r)$ the detected envelope of the 2nd harmonic. Then

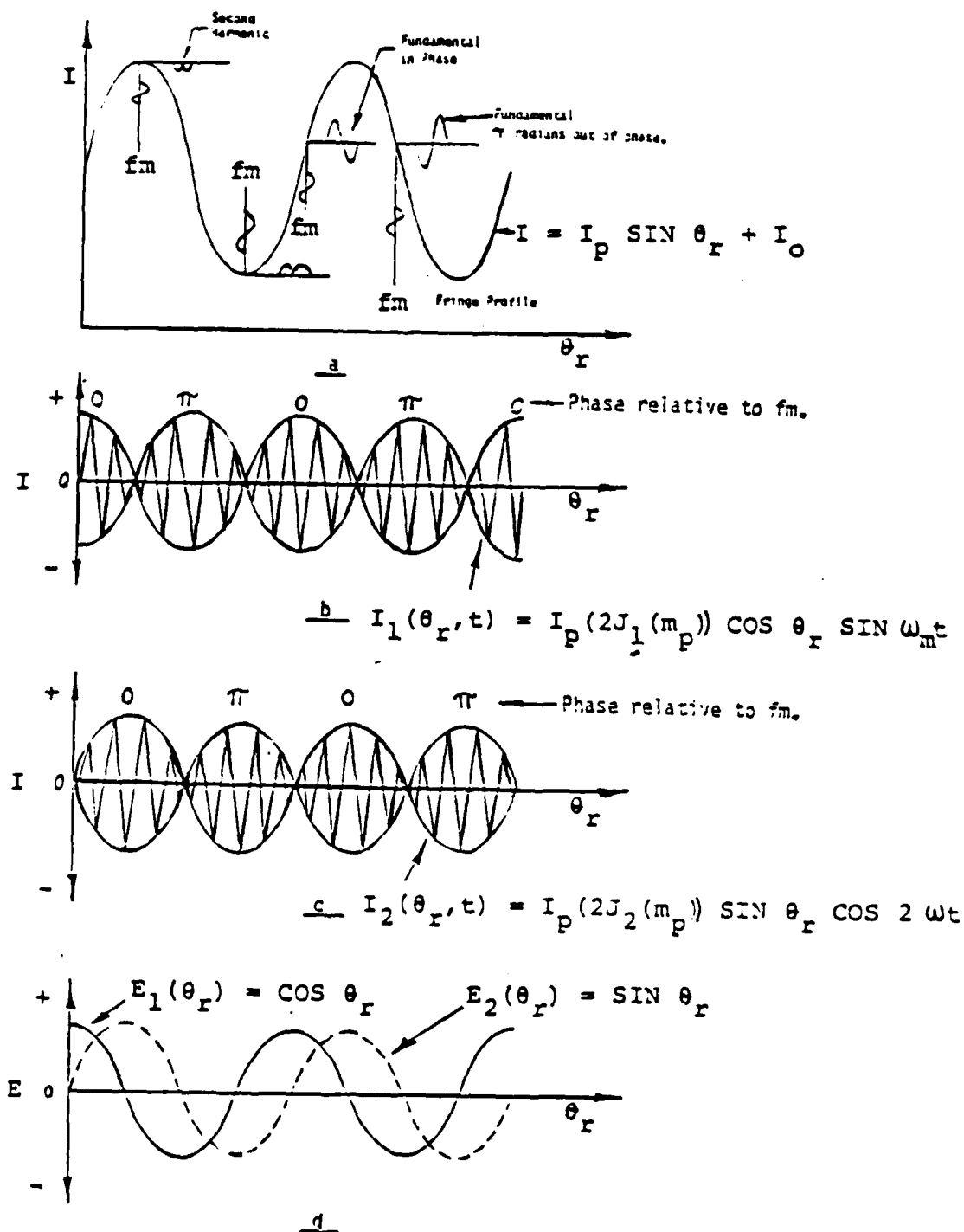
$$E_2(\theta_r)/E_1(\theta_r) = I_p 2J_2(m_p)\sin \theta_r / I_p 2J_1(m_p)\cos \theta_r$$

$$= J_2 \sin \theta_r / J_1 \cos \theta_r = (J_2/J_1) \tan \theta_r$$

and

$$\theta_r = \tan^{-1} [J_1 E_2(\theta_r) / J_2 E_1(\theta_r)]$$

The sign of the argument of the arctan identifies the quadrant for each quarter-fringe cycle. Comparison of the signs of the numerator and denominator of the present and previous measurements are used to identify quadrant boundary



crossings. Consequently, fractional fringe quarter cycles can be measured, and the boundary direction crossing used to sum quarter fringe cycles.

SOFTWARE DEVELOPMENT

Software was developed using the preceding analytical description. It is important to note that the software was a development effort intended for diagnostics and for the future determination of the requirements for complete data reduction by a HP9836 computer or the range computer.

The two programs (written in the BASIC language) are included as Appendix B. One software program is used for collecting and processing range data when a range trigger is received and the other is used to collect calibration data for a given hologram cube corner reflector combination. The calibration program was derived from the range program, consequently parts of the program are not integral to the calibration process. Figure 9 is the flow chart for the range data collection and processing program.

Range data consisted of the outputs from the two photomultipliers and a 3Khz reference (the 3Khz signal driving the elevation galvanometer). For each cycle of the 3Khz reference there are twelve data samples for each channel of the ADC/Computer Interface. Transfer of data was done using the Basic TRANSFER command. Data was transferred to a DMA Buffer in real time for a predetermined time after receiving a range trigger. At the end of this predetermined time, determined by the DMA Buffer size, the binary data was converted to decimal and placed in a program array for use by the program.

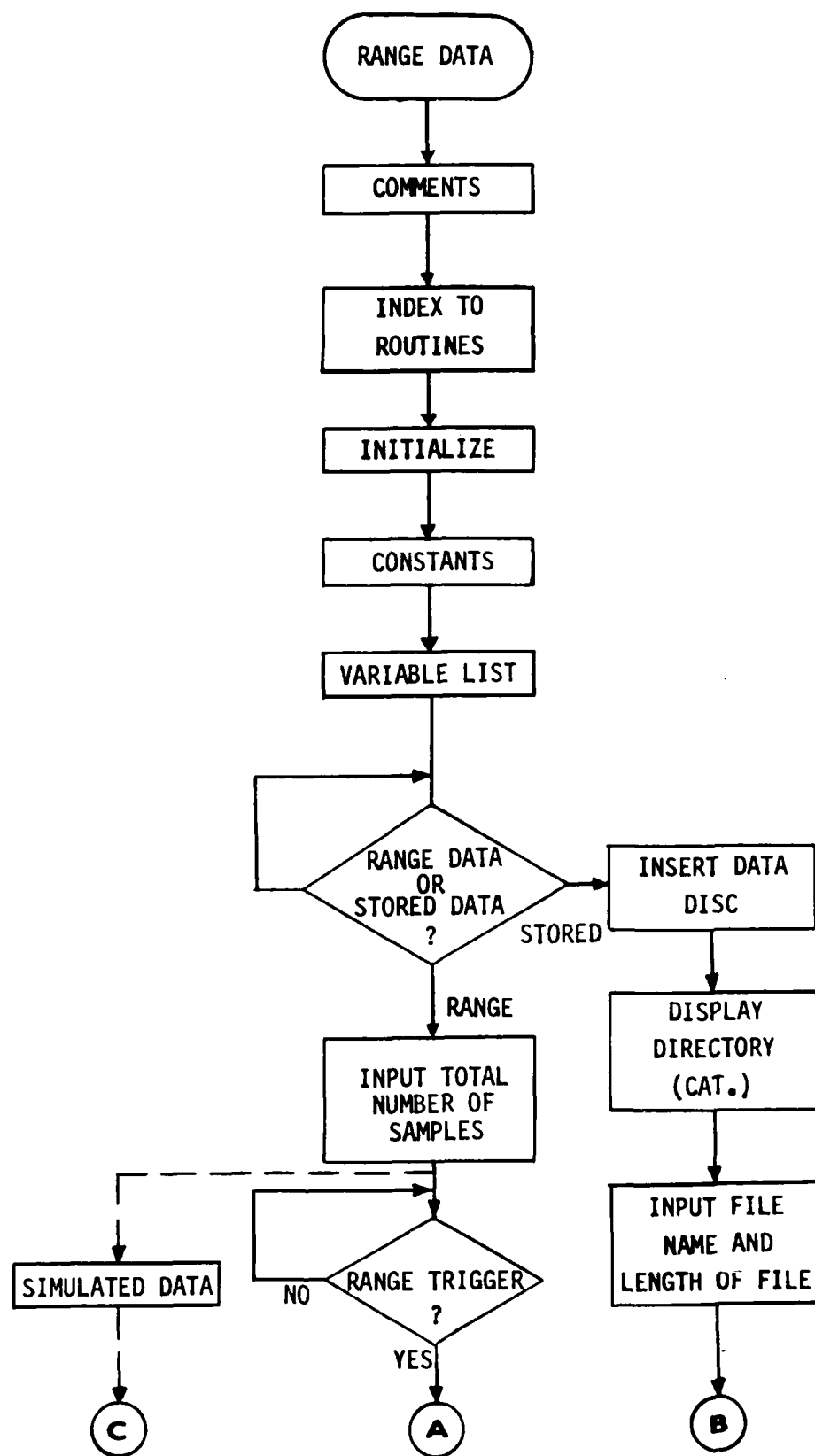


Figure 9. Computer Program Flow Chart

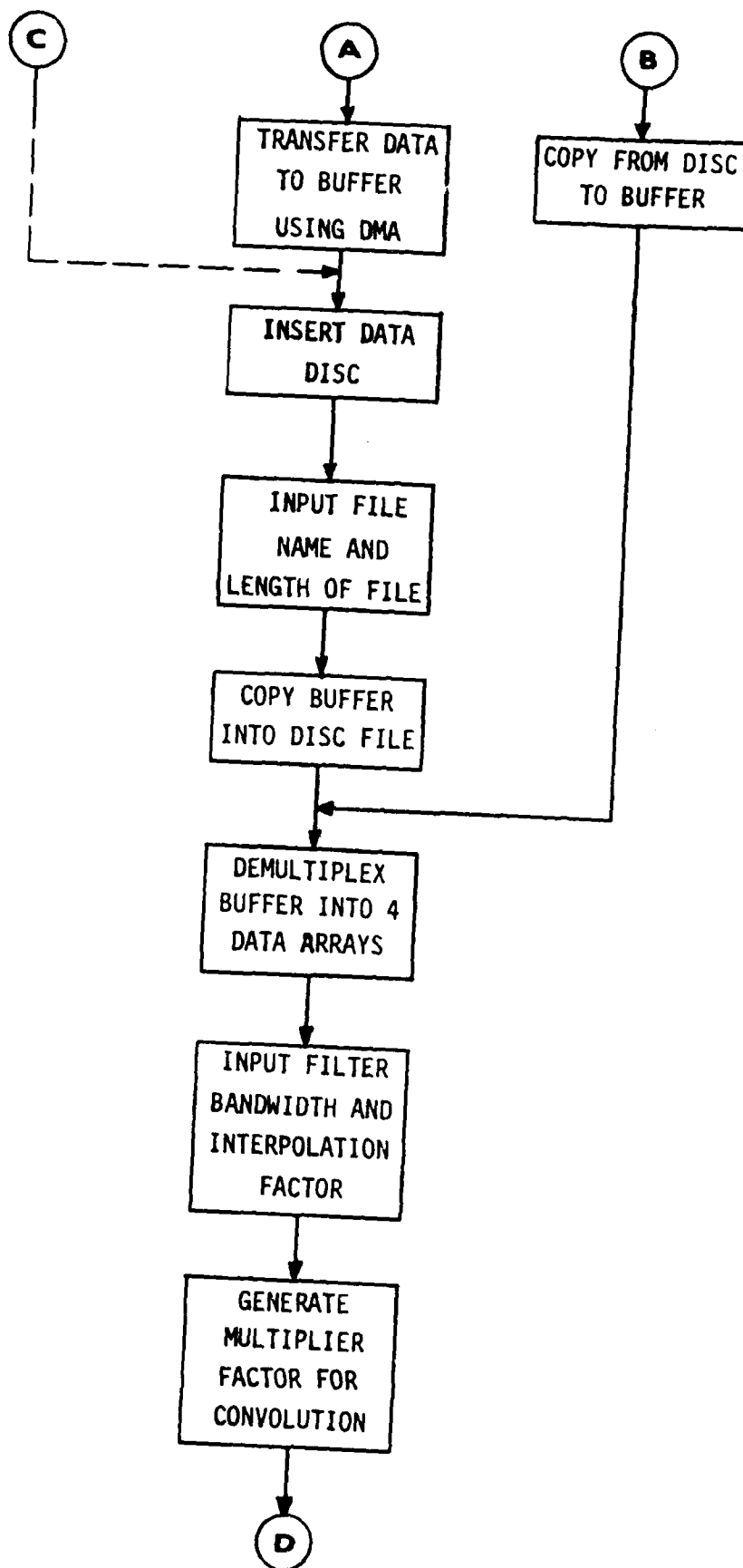


Figure 9. Computer Program Flow Chart

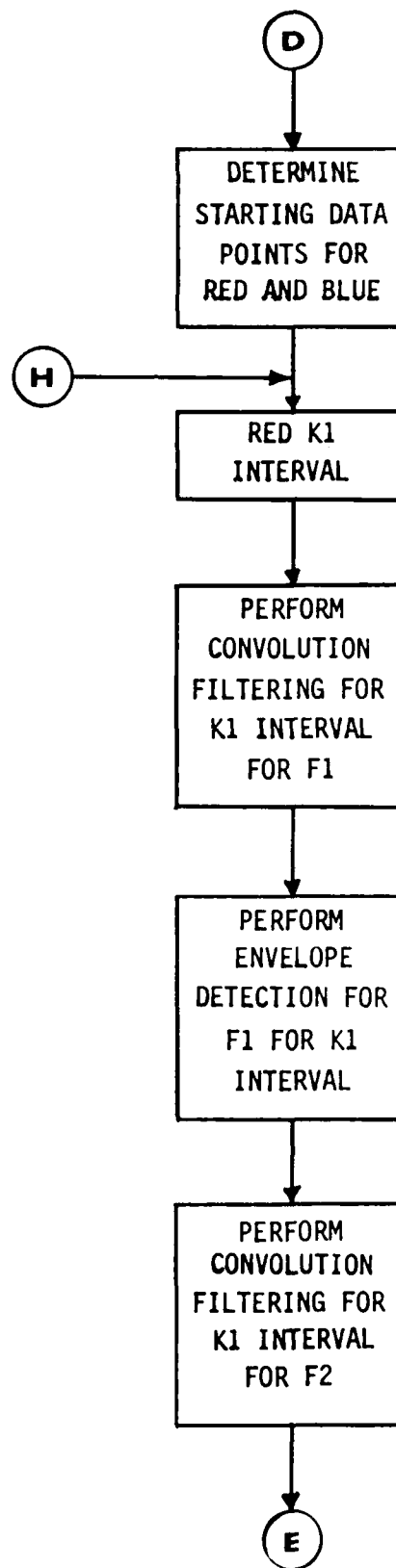


Figure 9. Computer Program Flow Chart (continued)

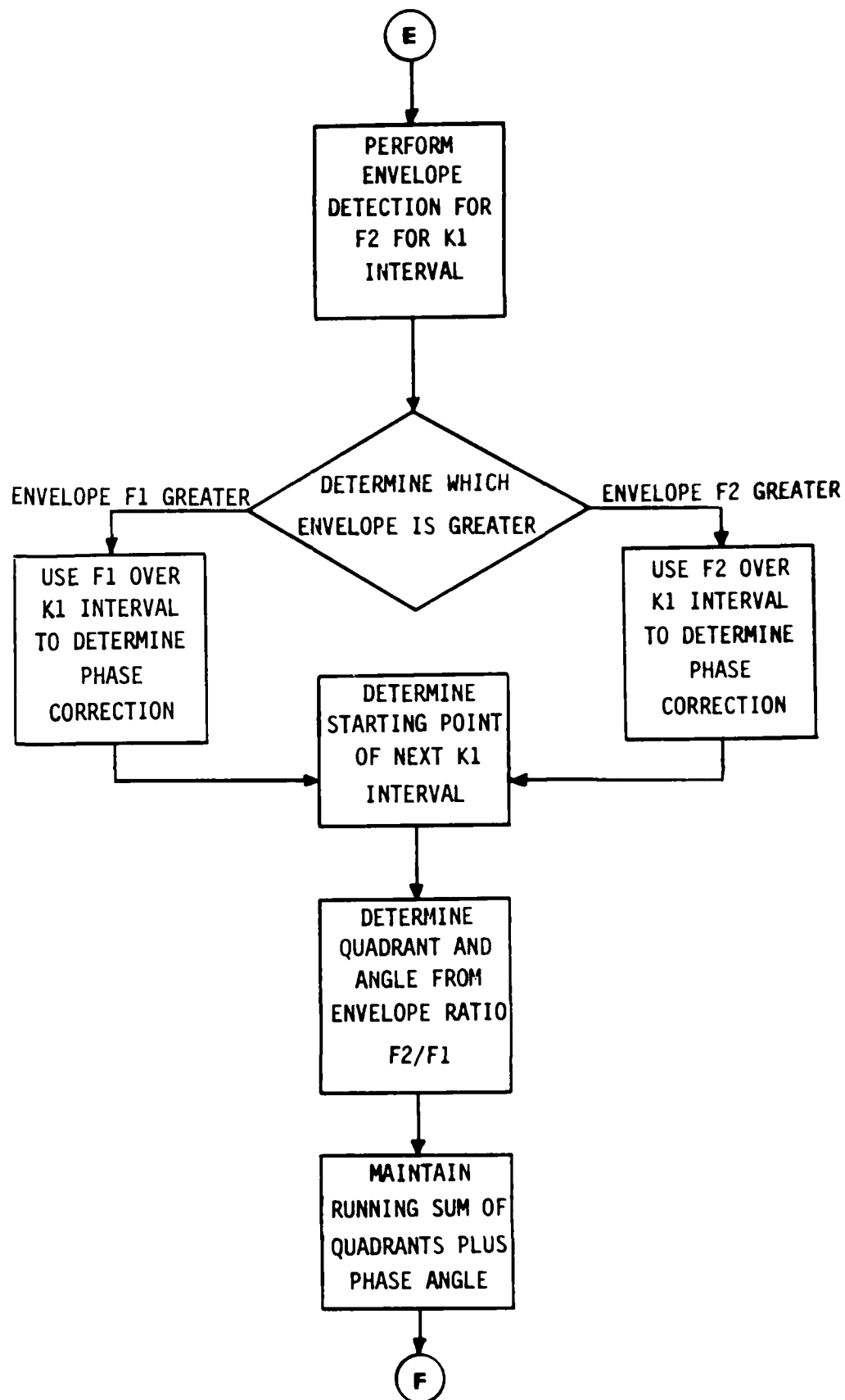


Figure 9. Computer Program Flow Chart (continued)

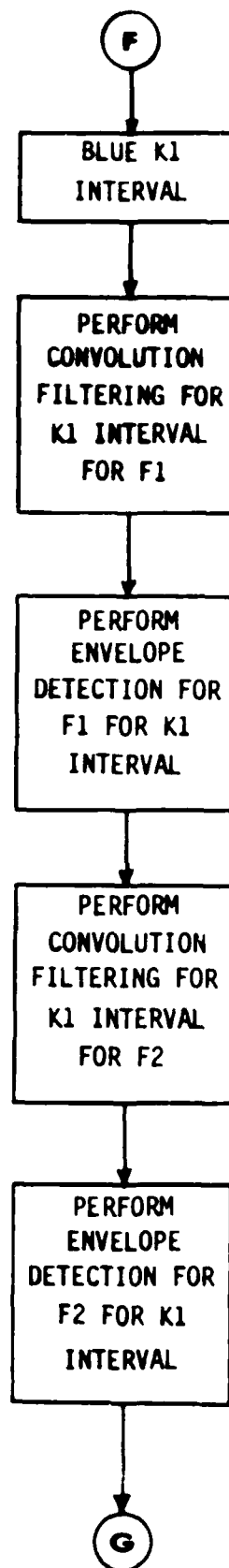


Figure 9. Computer Program Flow Chart (continued)

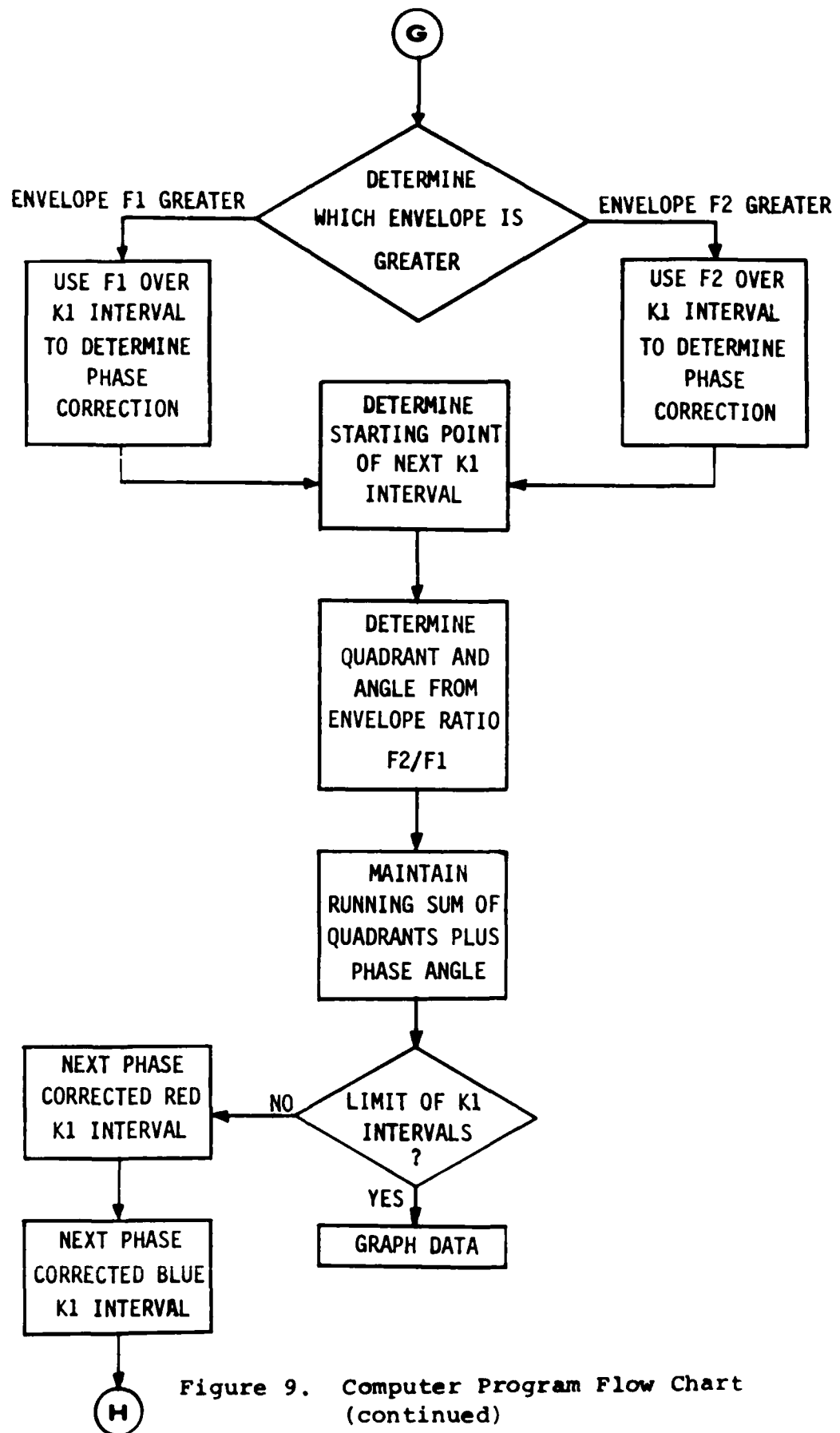


Figure 9. Computer Program Flow Chart (continued)

TEST RESULTS

The test configuration and results of tests performed jointly by BAC and BRL personnel are summarized in Table 1 below. Following the table is a discussion of the reduced data listed in the order of those rounds which produced reducible data.

TABLE 1 - ROUND FIRING SUMMARY

| NO. | ROUND TYPE | DATA SET | CONFIGURATION | RESULTS |
|-------|----------------|----------------|--|----------------------------------|
| 17450 | 30 mm spinning | | Scotchlite on nose, to check for obturation | No range trigger |
| 17452 | 30 mm spinning | Run-1 | Single 1 ⁰ hologram Red beam set for yaw Blue beam set for roll Flight path 30 m | Data on tape No range trigger |
| 17453 | 30 mm spinning | Run-2 | " | No data No range trigger |
| | | Run-3 | No retro, range trigger check | |
| | | Run-4 | No retro, range trigger check | |
| | | Run-5 | No retro, range trigger check | |
| 17458 | 30 mm spinning | Run-6 | Same as round 17452 | Early range trigger No data |
| | | Run-7 | No retro, range trigger check | |
| 17460 | 20 mm no spin | Run-8 | 2-axis, 1 ⁰ hologram Cal-3A Flight path 15 m | Data on tape and in computer |
| 17461 | 20 mm no spin | Run-9 Cal-4 | " | " |
| | | Run-10 | No retro, range trigger check | |
| | | Run-11 | No retro, range trigger check | |

TABLE 1 - ROUND FIRING SUMMARY

| NO. | ROUND TYPE | DATA SET | CONFIGURATION | RESULTS |
|-------|---------------|-----------------|---|---|
| 17464 | 30 mm | Run-12 Cal-5 | Set up for in-bore measurement. Folding mirror in blast room | No data due to obturation |
| 17465 | 25 mm | Run-13 Cal-6 | Muzzle angle and linear displacement setup. Reflector in collar on muzzle. | Collar moved No data |
| 17466 | 25 mm | Run-14 Cal-6 | " | Displacement and angle data Gun moved back 16mm in last two shots |
| 17467 | 25 mm | Run-15 Cal-6 | " | Muzzle pressure gauge failed |
| 17468 | 25 mm | Run-16 | " | |
| 17469 | 25 mm | Run-17 | " | |

Round 17452 - The purpose of this round was to test the feasibility of determining pitch, roll, and yaw angles of a spinning projectile with a fixed beam (no dither). The 30 mm round was fitted with a single 1° hologram and cube corner reflector. The calculated fringe angle is 6 milliradians, based on the hologram diffraction angle and the specified height of the retroreflector.

A retroreflector whose far-field pattern consisted of two spots was used to facilitate roll measurement. The optics for the blue beam were adjusted to form the far-field pattern of the return beam on a slit in front of the photomultiplier so an electrical pulse occurs each time the projectile rolls 180° . The reflection assembly on a spinning projectile generates a frequency modulated (fm) signal, where the number of cycles per half revolution is proportional to the total yaw angle (Reference 2). The polar angle is proportional to the time between the minimum frequency point of the fm signal (red beam) and the electrical pulse from the blue beam.

Figure 10 shows the laser measurement results of total yaw angle as a function of time. Data obtained from the analysis of a series of orthogonal spark shadowgraphs obtained for this round are shown in Figure 11. The laser measurement is quite similar in shape to the smooth curve fitted to the shadowgraph data. It is interesting that the minimum yaw points at 10m are not on the smooth curve, but they show up on both the raw shadowgraph data and the reduced laser data.

Figures 12a and b are laser and shadowgraph measurements of the horizontal and vertical yaw angle components. The square boxes on the laser curve mark the points at which the pair of far-field blue spots line up with the slit. The projectile rolls 180° between each pair of square boxes, thus the laser angle sensor can measure all three projectile angles.

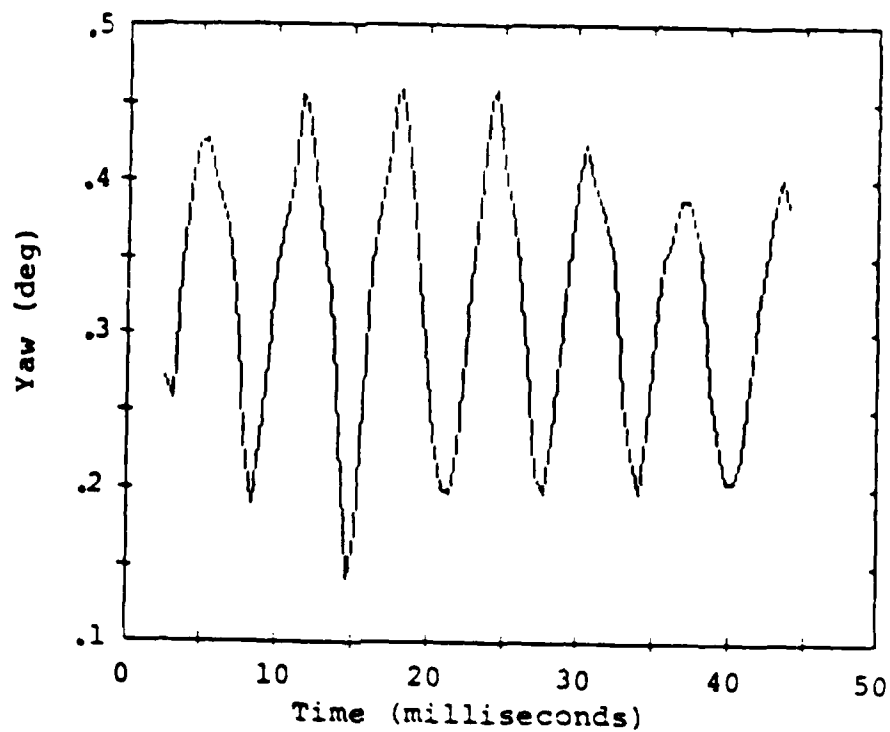


Figure 10. Total Yaw, Round 17452, Laser.

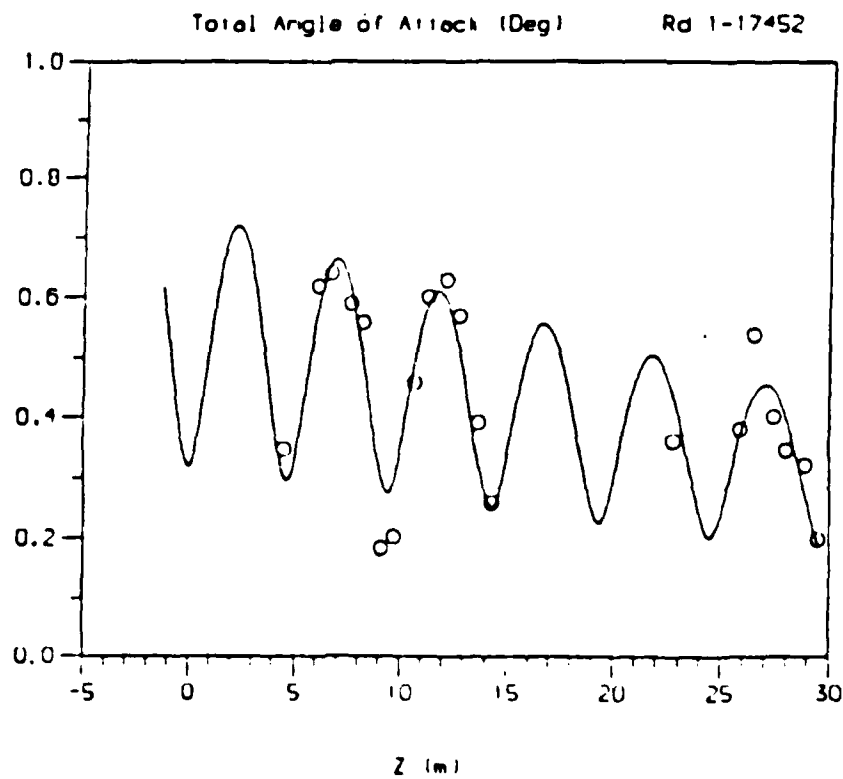
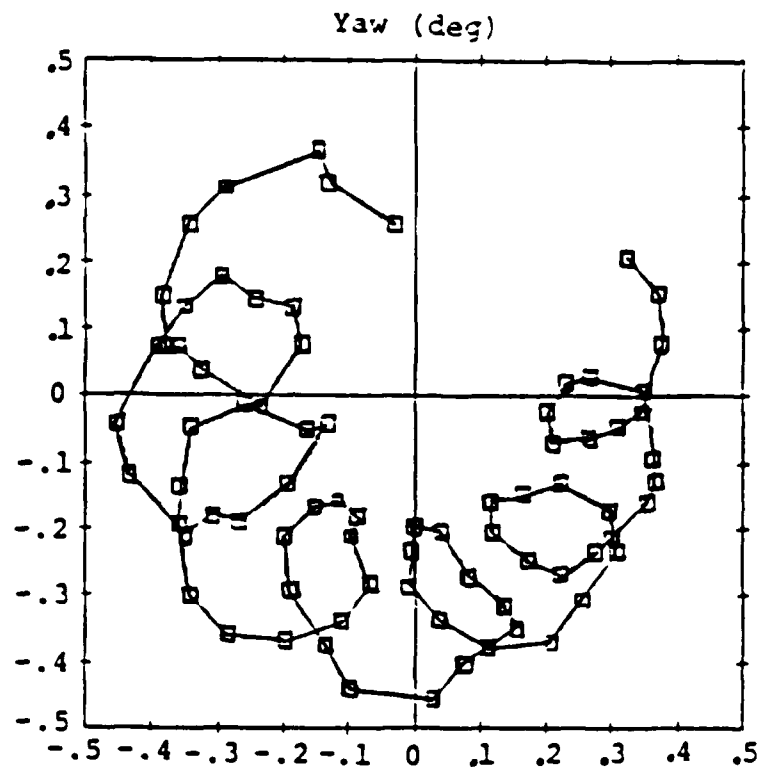
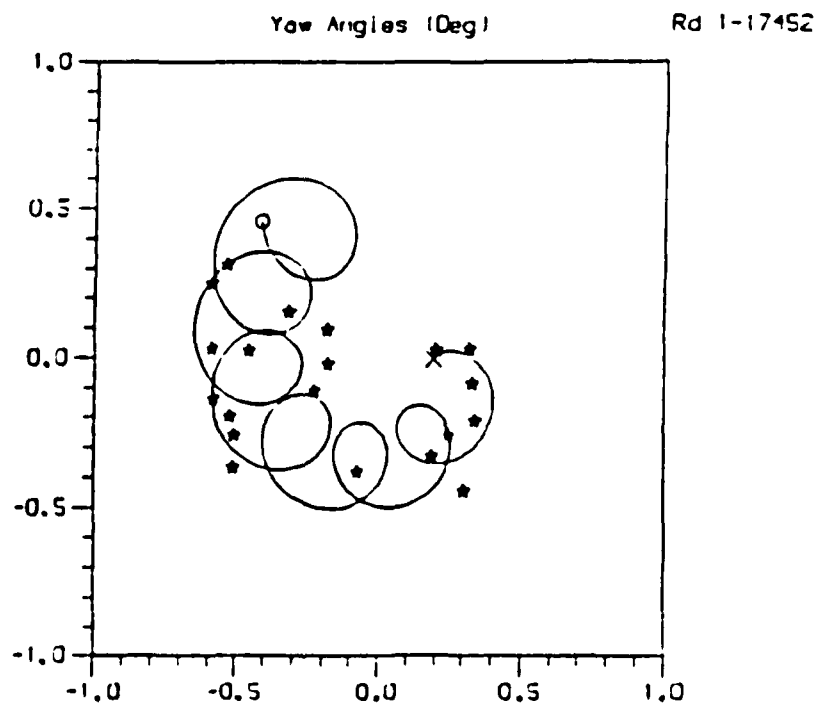


Figure 11. Total Yaw, Round 17452, Shadowgraph.



a) Laser Data

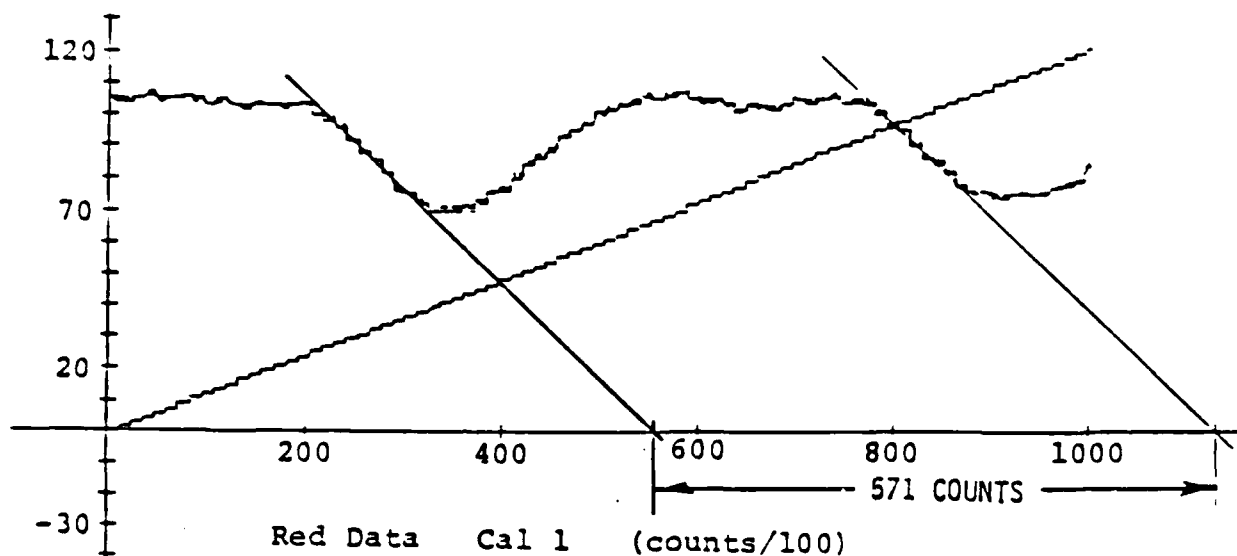


b) Shadowgraph

Figure 12. Yaw Angle, Looking Up-range, Round 17452.

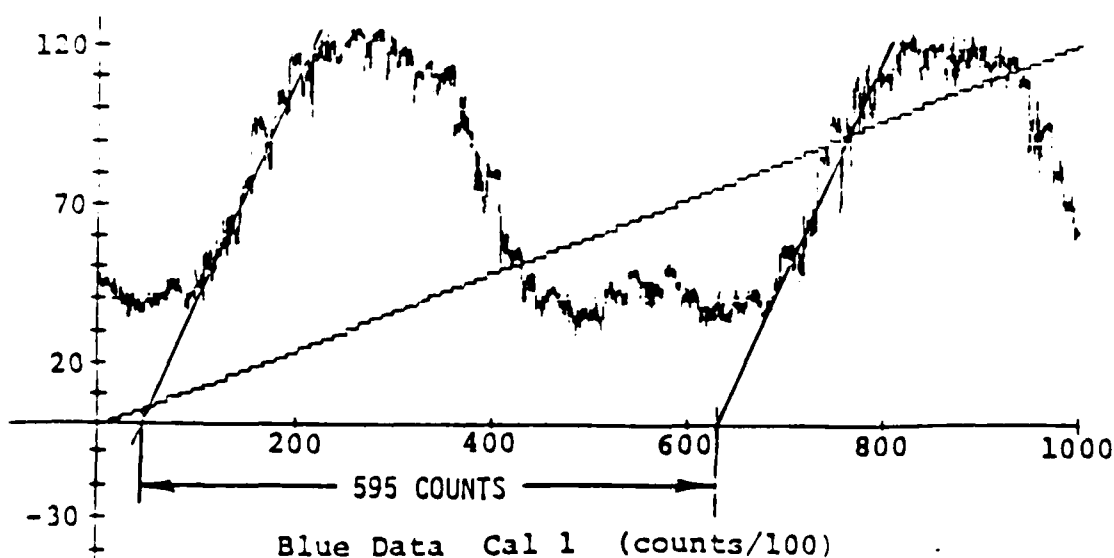
Round 17453 and 17458 - No data was acquired on these rounds due to problems with the range trigger. However, the rounds were calibrated prior to each firing. The calibration technique is described below.

The round is mounted in a precision two-axis motorized optical mount. The mount has a digital encoder that is coupled to the HP-9836 computer. The initial angle is adjusted so that the Fresnel reflection from the reflector assembly returns to the laser. The projectile is rolled so that motion on one axis produces no modulation of the fringes for the orthogonal axis. The mount is then driven slowly and the fringe pattern versus encoder counts is stored in the computer. Figured 13 and 14 show typical fringe patterns. The number of counts per fringe is found by the geometric construction shown on the figure.



1240 COUNTS = 10^{-2} rad
 FRINGE ANGLE = 4.61 mrad (0.264 deg.)

Figure 13. Fringe Angle Calibration, Red.



FRINGE ANGLE = 4.80 mrad (0.275 deg.)

Figure 14. Fringe Angle Calibration, Blue.

Rounds 17460 and 17462 - The purpose of these rounds was to evaluate the use of circular dither with two laser lines to measure all three angles of a non-spinning projectile. The reflector assemblies were calibrated in the precision 2-axis mount prior to firing the rounds. The hologram configuration was the orthogonal nested holographic grating as described in the preceding section. The hologram diffraction angle was one degree.

Fringe data was obtained for both of these rounds, and the signals are quite clean. Unfortunately, the rounds were very dynamic. The fringe rate is too high for processing with the algorithm that was developed on this program. The data reduction algorithm is designed for the case where the fringe rate is small compared to the dither rate. Figure 15 shows the output of the data reduction algorithm for these rounds. Figure 16 is the range data for round 17460.

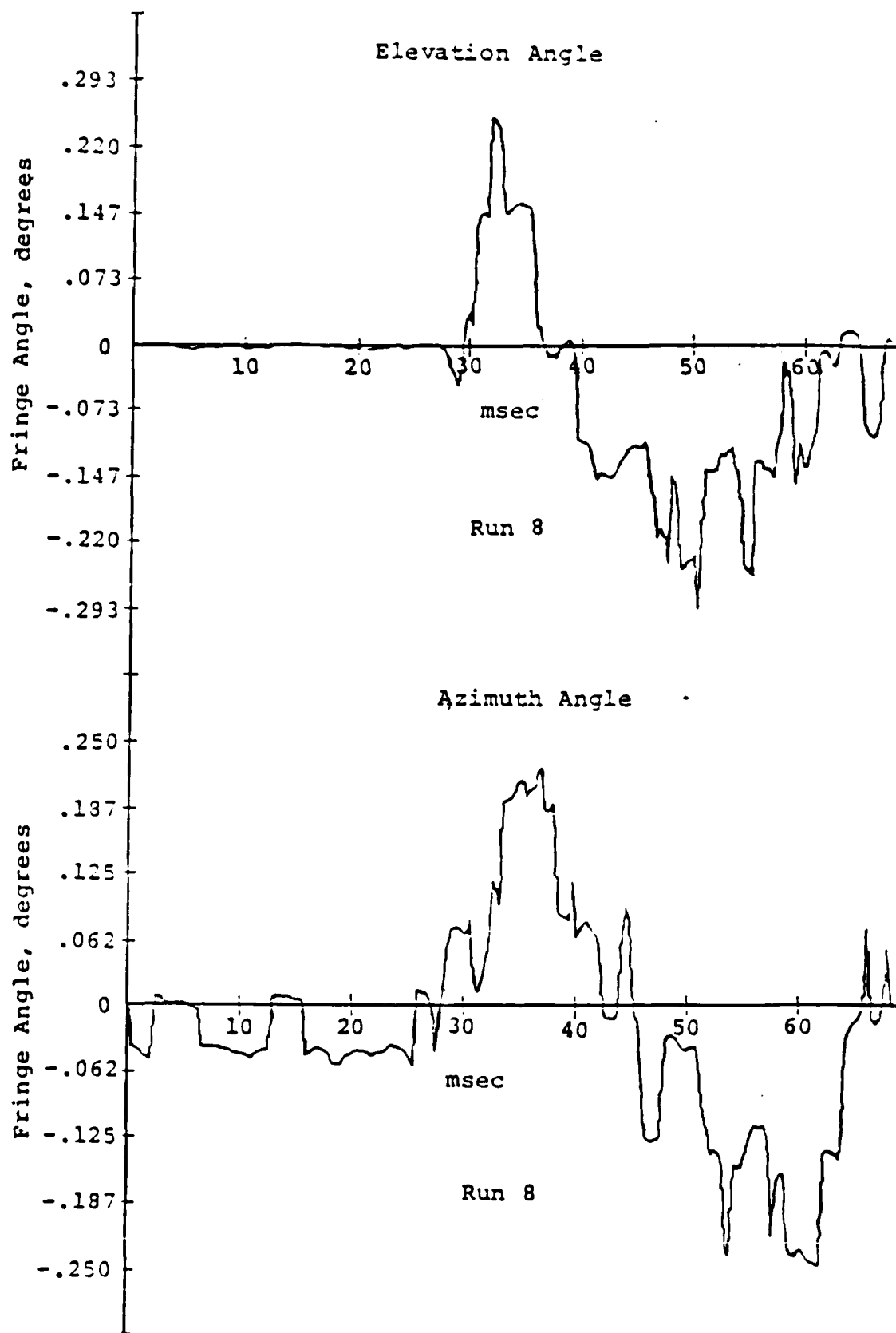
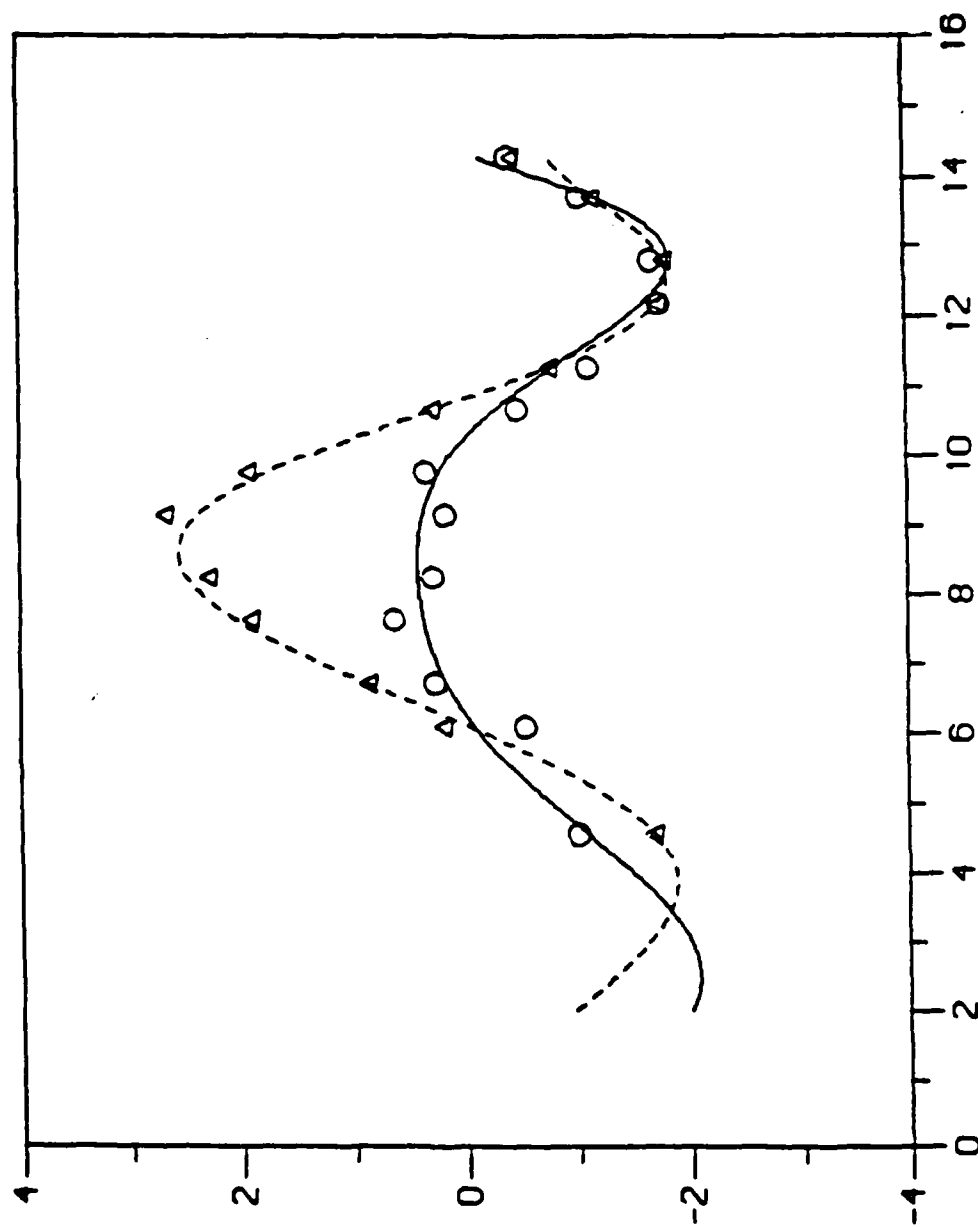


Figure 15. Computer Output, Round 17460.

Yaw Angles (Deg) vs. Z (m) Rd 1-17460



Solid: Beta, Dashed: Alpha
Figure 16. Yaw Angles versus Z, Round 17460.

Round 17464 - In-Bore. The optical unit was in the loading room. The laser beam passed through a 50 mm hole in the blast room wall and was directed by a pair of mirrors into the muzzle. The projectile base was fitted with a rubber seal in an attempt to prevent the passage of combustion gases. The seal failed, and no data was obtained from this round due to obscuration of the optical beam by leakage of hot combustion gases.

Rounds 17466 and 17467 - Muzzle Angle and Displacement. This setup is similar to the in-bore setup, except that the second folding mirror directed the laser beams to a reflector assembly mounted on a collar attached to the muzzle. In addition, the Boeing Dynamic Displacement Measurement System (DDMS) was used to measure the horizontal displacement of the muzzle.

Figures 17 and 18 are reduced angle data for these rounds. The blue laser measures the azimuth angle and the red laser measures the elevation angle.

DDMS counter output versus sample number is shown in Figure 19. Positive numbers correspond to displacement toward the right side of the range, looking down-range. The sample rate is 36 kHz.

The data for run 14, round 17466, indicates no motion for the first 5.6 milliseconds (200 samples), then a linear displacement of .280 millimeters in the following 2.5 milliseconds. The DDMS record then becomes flat, indicating that hot muzzle gases have passed through the beam. The hot gas destroys the lateral coherence of the signal beam, so the fringe count stops. The hot gas also affects the angle measurement system.

The displacement data can be used to estimate angular motion. Assume that the gun-tube bending follows a circular arc, then the angular change at the end of the tube is equal to the displacement divided by one-half the tube length. This results in an angle of 0.56 milliradian (0.032 degrees) for round 17466.

Proximity probes (used by BRL personnel following the above series of tests) registered a peak pointing angle of roughly 0.1 degree in the horizontal direction and a somewhat larger angle in the vertical direction.

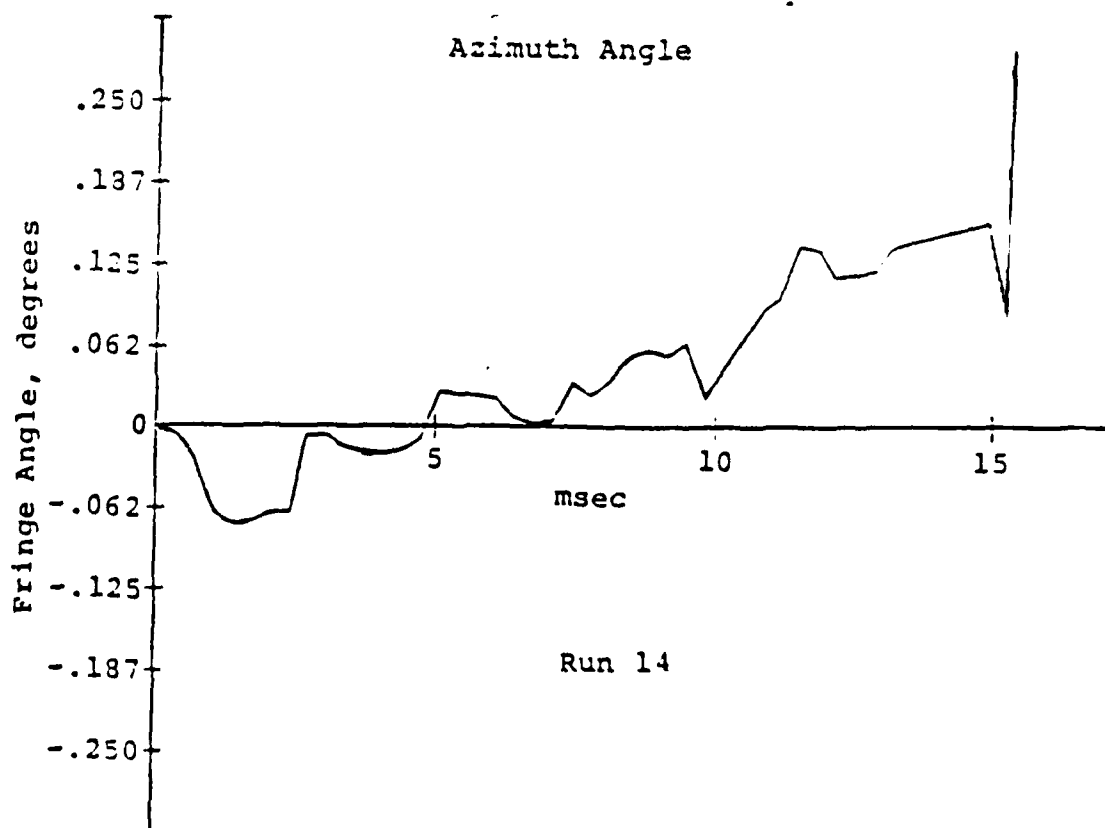
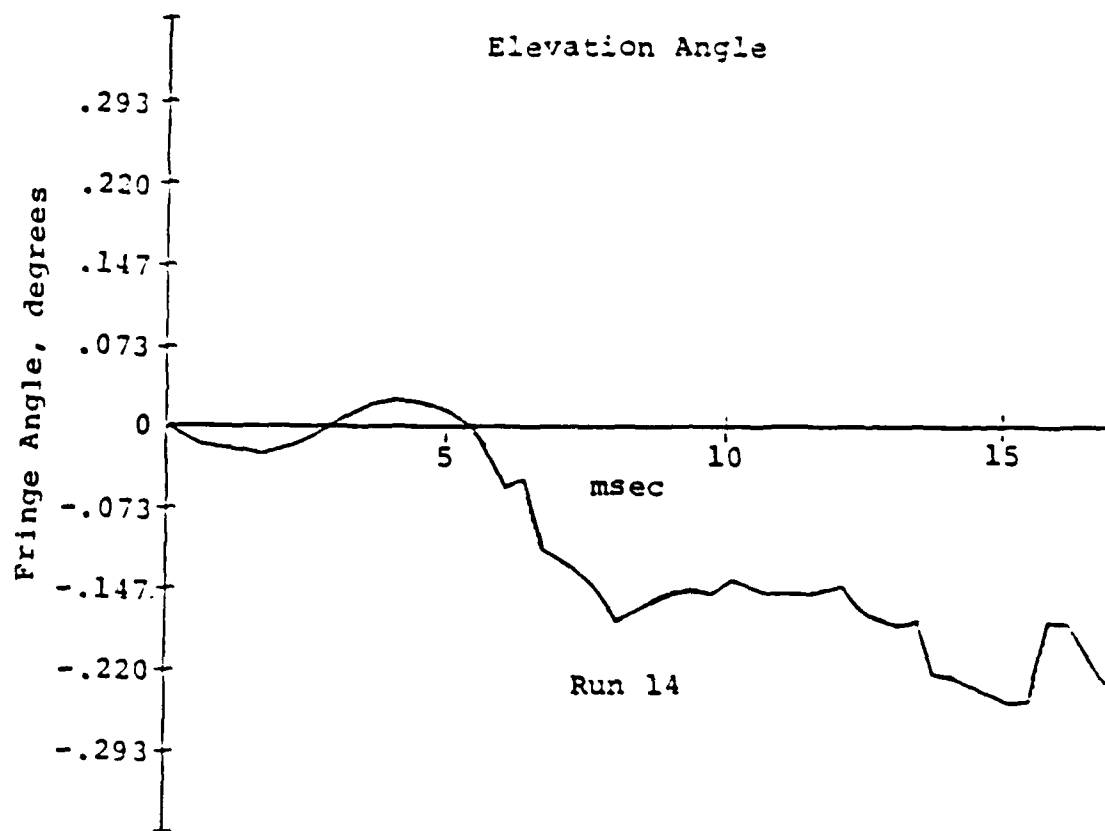


Figure 17. Muzzle Angle, Round 17466.

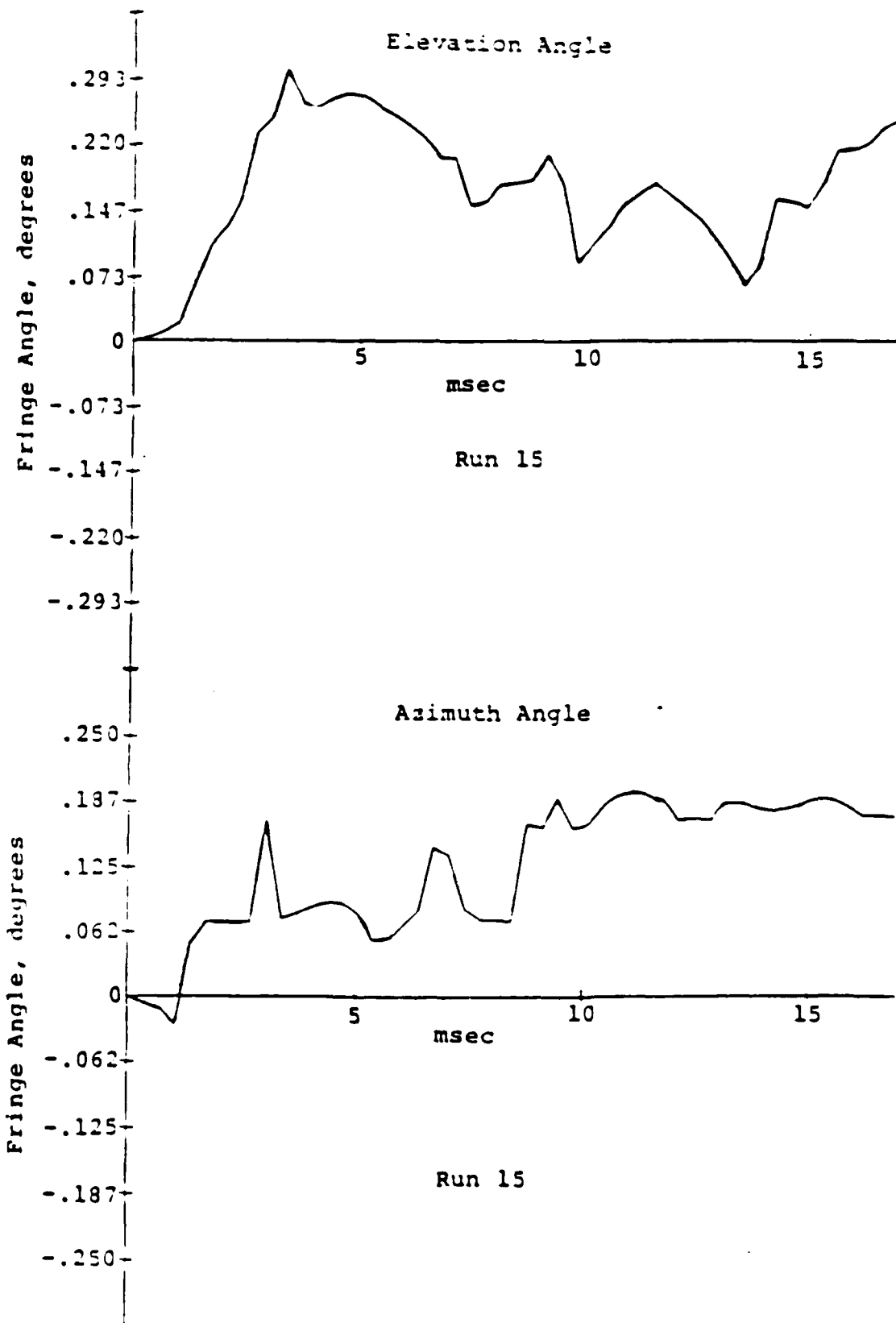
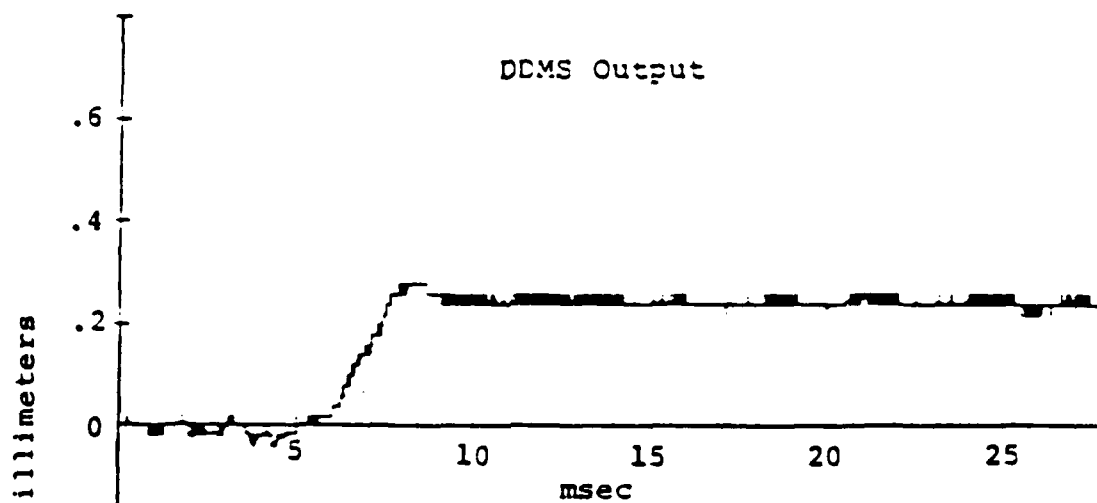
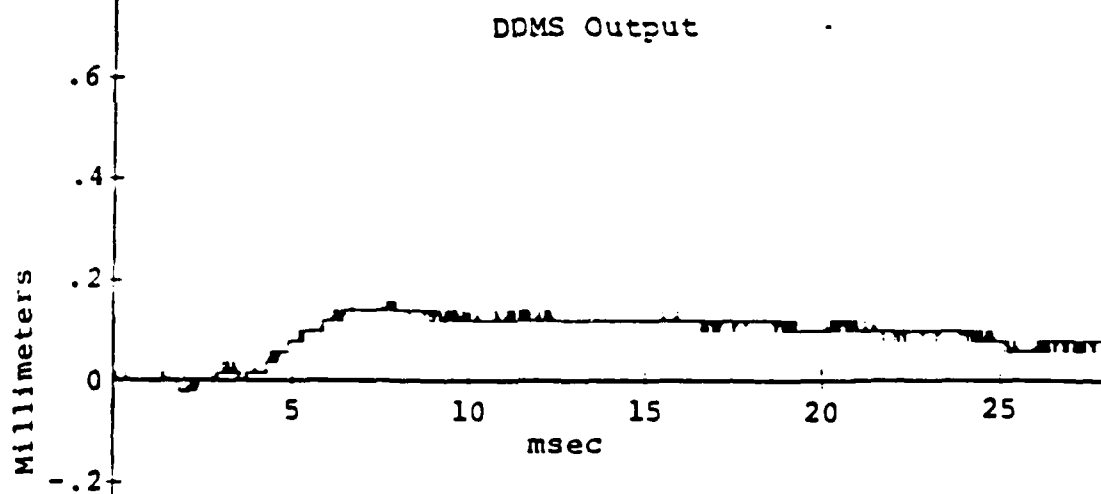


Figure 18. Muzzle Angle, Round 17467.



Run 14



Run 15

Figure 19. Muzzle Displacement, Round 17466 and 17467.

ANALYTIC MODEL

The equations relating the projectile angular motion with the output current from the photodetector will be developed in this section. The relationships for the general case (a spinning, yawing projectile observed by an optical system employing circular dither) are first derived. The simplifications that result when using a fixed laser beam or non-spinning projectile are then discussed.

A right-handed coordinate system is used in the analysis, with the X axis horizontal, the Y axis vertical and the Z axis along the range centerline. The optical centerline also coincides with the Z axis. The projectile undergoes pitching and yawing motions as a function of time, so the orientation of a projectile based coordinate system (x,y,z) may be specified by the direction cosines ($\cos\alpha_p$, $\cos\beta_p$) as shown in Figure 20-a.

The intensity of light reflected by the holographic grating/retroreflector combination placed at the nose of the projectile is a function of the angular orientation of the input laser beam relative to an axis normal to the hologram (Reference 1). It is therefore convenient to express the orientation of the body in terms of the angular coordinates

$$(u_p, v_p) = (\cos\alpha_p, \cos\beta_p)$$

where u_p and v_p may be thought of as the pitch and yaw of the projectile. The angles are chosen so that u_p is oriented perpendicular to the hologram grating lines (i.e. parallel to the direction in which light is diffracted). If the input angle is small, the reflection coefficient, $R[u_p(t)]$ is approximately a periodic function of the input angle as shown in Figure 20-b.

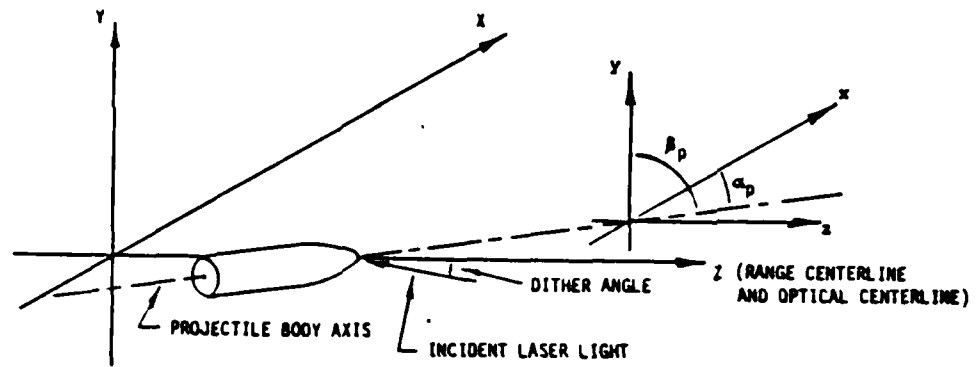
As part of the optical system, circular dither will be imparted to the laser light. Thus at any instant of time the angle between a ray representing the portion of the laser beam striking the retroreflector and the coordinate axes can be defined by the angular coordinates

$$(u_l, v_l) = (\cos\alpha_l, \cos\beta_l)$$

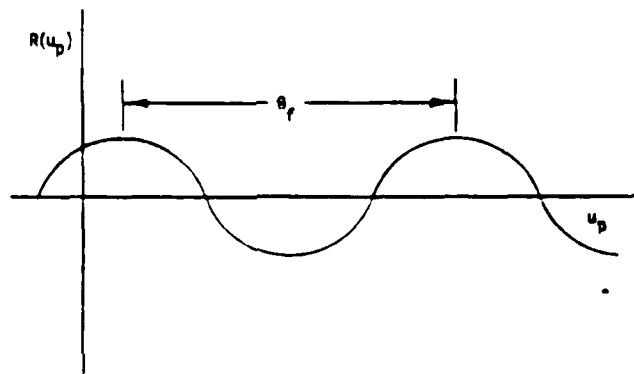
where α_l and β_l are the angles between the incident light and the X and Y axes respectively. Employing a trigonometric identity, the angle between the laser beam (or the body axis) and the Z axis may be expressed as

$$\gamma = \cos^{-1}(1 - \cos^2\alpha - \cos^2\beta)^{1/2}$$

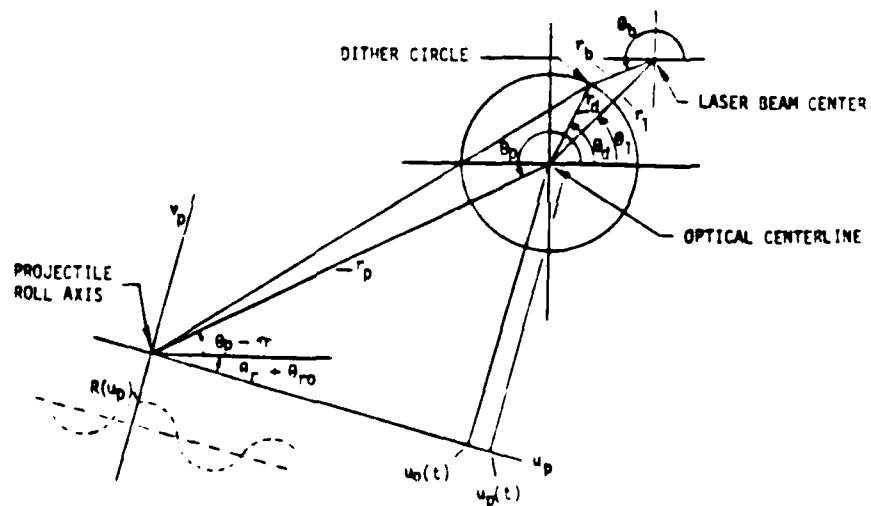
Figure 20-c illustrates the optical and projectile coordinate systems. The axes u_l and v_l are centered on the



a) Projectile Geometry



b) Reflector Reflection Coefficient versus Angle



c) Angular Geometry of Projectile and Laser Beam

Figure 20. Analytic Model Geometry.

optical centerline with u horizontal. The body fixed projectile axes, u_p and v_p , rotate as the projectile rolls. The circle represents a circular dither of the laser beam. The polar coordinates of the center of the laser beam are (r_1, θ_1) .

The circular dither can be resolved into time varying components in the projectile coordinate system

$$u_p(t) = r_d \cos(\omega_d t - \omega_r t + \theta_{d0} - \theta_{r0}) + u_0(t)$$

$$v_p(t) = r_d \sin(\omega_d t - \omega_r t + \theta_{d0} - \theta_{r0}) + u_0(t)$$

where r_d = angular radius of the circular dither
 ω_d = dither angular frequency
 ω_r = roll angular frequency
 θ_{d0} = initial dither angle
 θ_{r0} = initial roll angle
 $u_0(t), v_0(t)$ = position of optical axis in projectile coordinates.

The component of the dither in the u_p direction (perpendicular to the holographic grating lines) produces a modulated output, while the component in the v_p direction (parallel to the grating lines) does not.

The detector current is the product of two time varying functions. Since the intensity of the laser is not spatially uniform, as the beam is swept across the reflector (due either to the dither or the motion of the projectile) the light entering the retroreflector is amplitude modulated. The reflector further modulates the beam because the reflection coefficient, $R[u_p(t)]$, varies with the angle of incidence.

If it is initially assumed that the laser beam profile is not a function of z , then the intensity may be represented by

$$I(t) = (r_b, \theta_b)$$

where $r_b = (r_d^2 + r_1^2 - 2r_d r_1 \cos(\theta_{d0} + \omega_d t - \theta_1))^{1/2}$

$$\theta_b = \cos^{-1}[(r_d^2 - r_b^2 - r_1^2)/2r_d r_1]$$

The circular dither is actually a conical scan with a fixed point at a distance z from the initial projectile position. The linear displacement of the beam with respect to the retroreflector is a function of $z - z_0$ (when $z = z_0$ there is no translation of the beam). The z axis dependence can be placed into the expression for $I(t)$ by replacing r_1 with $r_1(t)$, where

$$r_1(t) = r_1(0) \cdot (z_0 - z(t))/z_0$$

The photodetector current is therefore

$$i(t) = \rho \Gamma I(t) R[u_p(t)]$$

where ρ = detector responsivity (amp/watt)
 Γ = optical efficiency
 I = laser beam intensity

Two special cases will be considered, both assuming that the laser beam is uniform, so that $I(t)$ is constant.

(1) The retroreflector reflection coefficient is a pure sinusoid with period θ_f .

(2) The reflection coefficient is periodic, but has higher order components.

Case 1: $R[u_p(t)] = a + b \sin(2\pi u_p(t)/\theta_f - c)$

where a = average value
 b = amplitude
 θ_f = angle between fringes
 c = phase at $t=0$

$$u_p(t) = r_d \cos(w_d t - w_r t + \theta_{d0} - \theta_{r0}) + u_o(t)$$

Let $d = \rho \Gamma I$, $k = 2\pi/\theta_f$ and $w = w_d t - w_r t + \theta_{d0} - \theta_{r0}$
then the detector current is

$$i(t) = ad + bd \sin[kr_d \cos w + k u_o(t) - c]$$

Express the sine term as a sum to get

$$i(t) = ad + bd \sin(kr_d \cos w) \cos(k u_o(t) - c) \\ + bd \cos(kr_d \cos w) \sin(k u_o(t) - c)$$

We can then expand the terms with sinusoidal argument using the Bessel function identities

$$\sin(z \cos \theta) = 2J_1(z) \cos \theta - 2J_3(z) \cos 3\theta + \dots$$

$$\cos(z \cos \theta) = J_0(z) - 2J_2(z) \cos 2\theta + 2J_4(z) \cos 4\theta - \dots$$

The first few terms are

$$i(t) = ad + bd J_0(kr_d) \sin(k u_o(t) - c) \\ + 2bd J_1(kr_d) \cos w \cos(k u_o(t) - c) \\ - 2bd J_2(kr_d) \cos 2w \sin(k u_o(t) - c) \\ + 2bd J_3(kr_d) \cos 3w \cos(k u_o(t) - c) + \dots$$

These harmonic components can be separated by filtering, then bidirectional changes in $u_0(t)$ can be determined by tracking the coefficients of $\cos[ku_0(t)]$ and $\sin[ku_0(t)]$ for a pair of even and odd harmonics. Note that the harmonic frequencies are multiples of the difference between the dither and roll frequencies. If the dither is high compared to the roll rate, roll produces a slow change in the phase of the dither.

Case 2: $R(u)$ is periodic but not a sinusoid.

The projectile reflector must be calibrated before firing. During calibration, $R(u)$ is measured over at least one period of θ_r . If the dither is on, the amplitudes of two adjacent harmonics are stored in a look-up table of harmonic amplitudes versus angle. If the dither is off, the table entries can be calculated as follows:

Write $R(u)$ as a Fourier series

$$R(u) = a_0/2 + a_1 \cos(ku) + a_2 \cos(2ku) + \dots \\ + b_1 \sin(ku) + b_2 \sin(2ku) + \dots$$

Replace u with $u + r_d \cos(w)$, then expand each term using the Bessel function identities and collect like harmonics. The first few term of the expansion and rearrangement are:

$$R_0(u) = a_0/2 + \sum_{j=1}^m [a_j \cos(jku) + b_j \sin(jku)] J_0(jk r_d)$$

$$R_1(u) = 2 \cos w \sum_{j=1}^m [a_j \sin(jku) - b_j \cos(jku)] J_1(jk r_d)$$

$$R_2(u) = 2 \cos 2w \sum_{j=1}^m [a_j \cos(jku) + b_j \sin(jku)] J_2(jk r_d)$$

$$R_3(u) = 2 \cos 3w \sum_{j=1}^m [a_j \sin(jku) - b_j \sin(jku)] J_3(jk r_d)$$

These are the harmonic amplitudes for the look-up table. The table is used to reduce the data after the round is fired.

The data is reduced in increments of the dither minus roll period. For each increment, harmonic component amplitudes can be found with the digital filtering routine in the data reduction algorithm. The look-up table is used to find the fractional angle u , where $u \leq \theta_r$. The total angle is incremented or decremented by θ_r each time u moves through the table.

PRELIMINARY DESIGN

This section contains preliminary design information for a three angle measurement system. The design addresses the limitations of the breadboard system tested at BRL, and defines the areas which require further development before a final design can be implemented.

The major limitation of the breadboard system was the dither rate. An available galvanometer deflection system was used to generate a conical scan at 3 KHz. However, in many cases the fringe rate was comparable to the dither rate. The dither rate must be much higher than the fringe rate, otherwise, the desired effect of the dither modulation is destroyed and angle information cannot be recovered. Spin stabilized rounds with fringe rates of 30 KHz have been observed in our test program. A dither rate of the order of 1 MHz would be appropriate. Oscillating mirrors (galvanometer type of resonant scanners) have a capability up to 20 KHz. Polygonal mirrors on turbine motors can generate higher scan rates, but produce linear rather than conical scans. The mechanical limit (reference 3) for a polygonal mirror is:

$$w = \frac{1}{2\pi r_o} \sqrt{\frac{8UTS}{p(3+n)}}$$

where r_o = Distance from center to edge
UTS = Ultimate tensile strength
 n = Poisson's ratio of material
 p = Volumetric density

Substituting the values for 6061 T6 aluminum, the mechanical limit for a 5 cm diameter scanner is about 10^4 revolutions/second. If the mirror had 20 facets, the scan rate would be 200 KHz. The practical limit, due to mirror deformation and safety, is well below the mechanical limit.

Acousto-optic beam deflectors (AOD) are an alternative method for generating a conical scan. Two-axis devices are available, so a circular dither is possible. The limit is set by the transit time of a sound wave across the aperture of the device. The transit time is the amount of time required to move the deflected beam from one position to another. Typical values for commercial deflectors are of the order of 10 microseconds. In the breadboard system the return beam was sampled at 12 positions around the dither circle. If an AOD is used with 10 microsecond transit time to sequentially deflect the beam to 12 points on a circular arc, the maximum rate would be 8.51 KHz. This rate is also inadequate for a spinning projectile.

A solution to the high dither rate requirement is to use a circular array of diode laser beams, where the diode lasers are sequentially modulated to produce a rotating optical source. Very high dither frequencies are possible with this technique. (We are currently modulating a Mitsubishi model ML 3001 diode laser at a frequency of 1 GHz.) The wavelength of laser-diodes are temperature sensitive. Diode lasers are available with fiber optic pigtails, so the lasers can be mounted on a temperature controlled heat sink and the fibers led out to form a circular array in the focal plane of the collimating lens. Two color operation is accomplished by selecting lasers so the even and odd fibers radiate in different wavelength bands, such as $830 \pm 5\text{nm}$ and $850 \pm 5\text{nm}$.

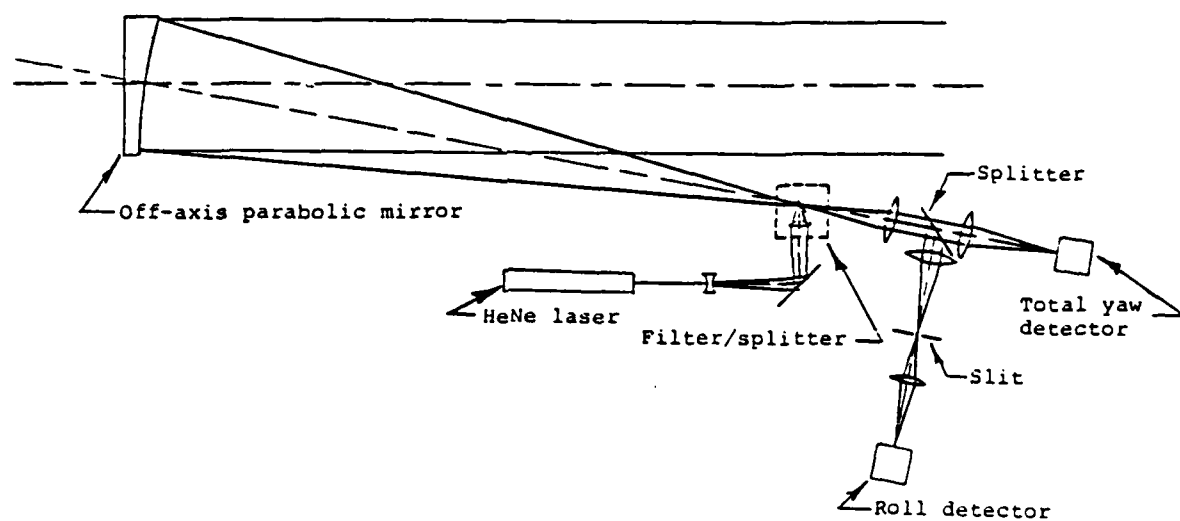
Figure 21 shows two optical configurations. Both configurations show an off-axis paraboloid mirror for collimating the beam. The beam diameter would be of the order of 30 cm, so the refracting optics as used in this program become more expensive than off-axis optics. Furthermore, the Fresnel reflection from the center of a refractor is avoided with reflecting optics.

Figure 21a is a three-axis system for measurements of spinning projectiles, using techniques verified on this program. The concept shown in Figure 21b is a circular dither system using diode laser, as described above. The first technique is lower risk and cost, since all the techniques have been demonstrated in this program.

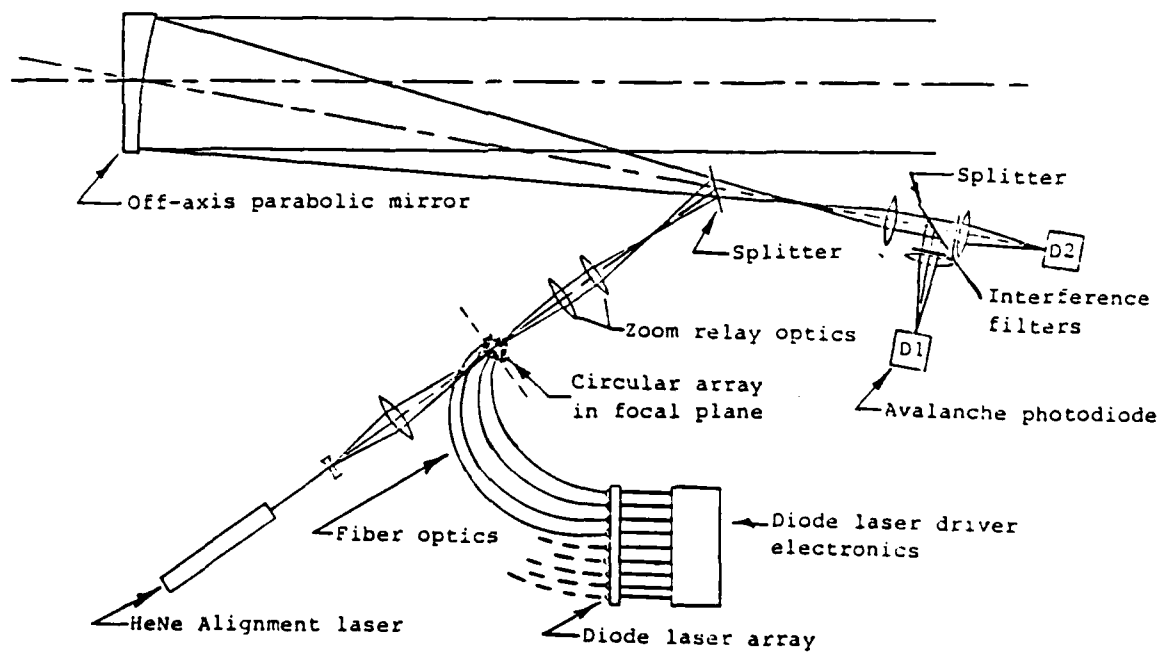
PHASE II PROGRAM

The major task in Phase II is the design and fabrication of a three-component angular measurement system. The results of the Phase I test program made a significant impact on the requirements for the Phase II system. The major requirement is for a much higher dither rate in order to follow the dynamics of some of the projectiles. Other requirements are increased laser beam diameter, improvements in the data reduction algorithms, and increased stability of the holograms in the humid environment at BRL. The Phase II tasks are listed below:

- 1) Design and fabricate a three-component angular measurement system. Two variations of the three-component system are possible.
 - a. Total yaw system for spinning projectiles. This technique was demonstrated on the prior contract, and improved to measure all three angles in the present contract.
 - b. Three-component system with circular dither for both spinning and non-spinning projectiles. The technical difficulties are greater and the risks are higher for this system. At this time it appears that the only method for generating the required high dither rate is to use a circular array of optical fibers, each driven by a diode laser, where the relative phases of the diode lasers is adjusted to produce a conical scan. The dither generation technique should be thoroughly evaluated with a breadboard device prior to fabrication of the three-component system.
- 2) Data reduction algorithms
 - a. No computer data reduction algorithm has been developed for 1a. Such a system would require the development of a suitable algorithm.
 - b. The algorithm for 1b has partially been developed in Phase I. Additional development is required based on the results of Phase I and the analytic model developed in Phase I. One of the important problems to be addressed is the magnitude of error caused by the deviation of the reflector reflection coefficient from a sinusoid. The algorithm already developed is for the sinusoidal case, but the reflection coefficient is approximately sinusoidal only when the diffraction efficiency is low. In order to maximize the return power, it is necessary to operate with a reflection coefficient versus angle that departs significantly from a sinusoid.



a) Fixed Beam Optics



b) High Speed Circular Dither Optics

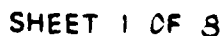
Figure 21. Laser Ballistic Sensor Configuration

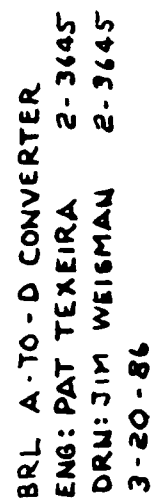
c. Algorithm development would be a joint Boeing and BRL task.

- 3) Investigate techniques to stabilize the holograms after fabrication. The diffraction efficiency of the holograms changed with time due to the change in relative humidity at BRL. Some of the holograms became cloudy and unusable.
- 4) For the system of 1b, design and fabricate signal conditioning, timing, and computer interface circuitry for an acceptable dither frequency in the range of one megahertz.

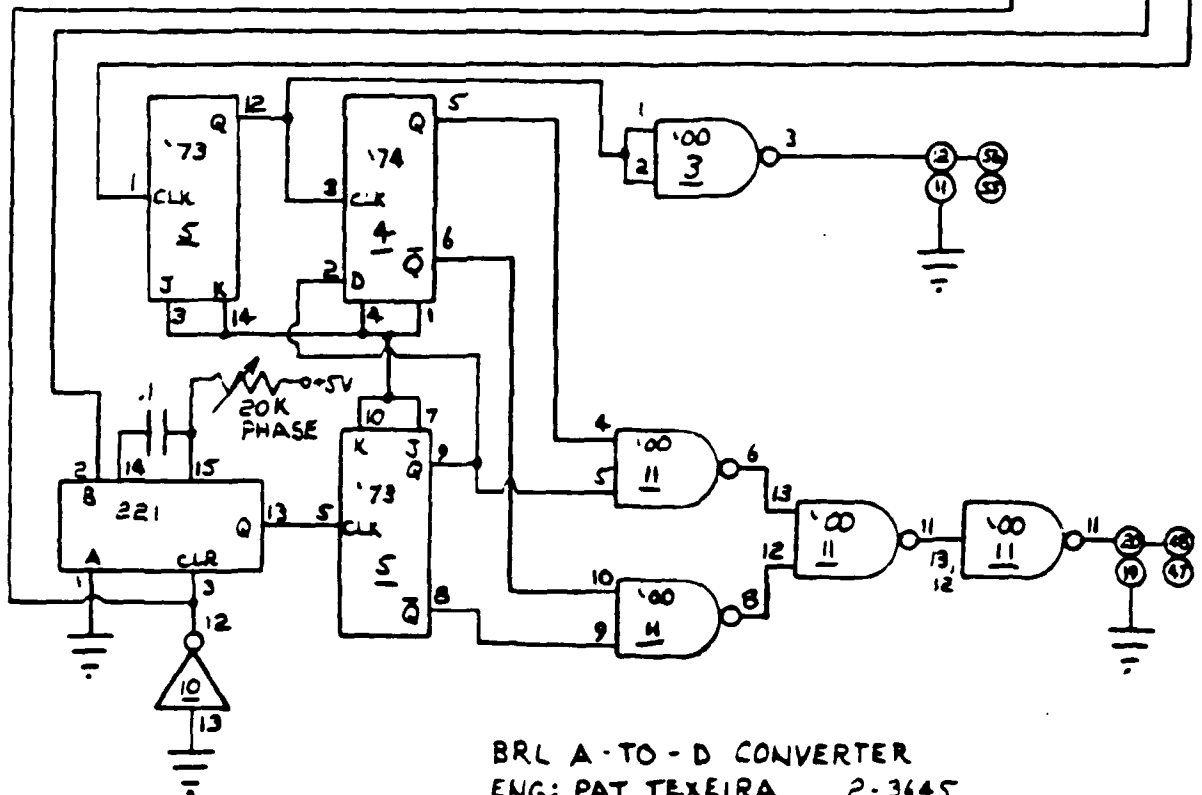
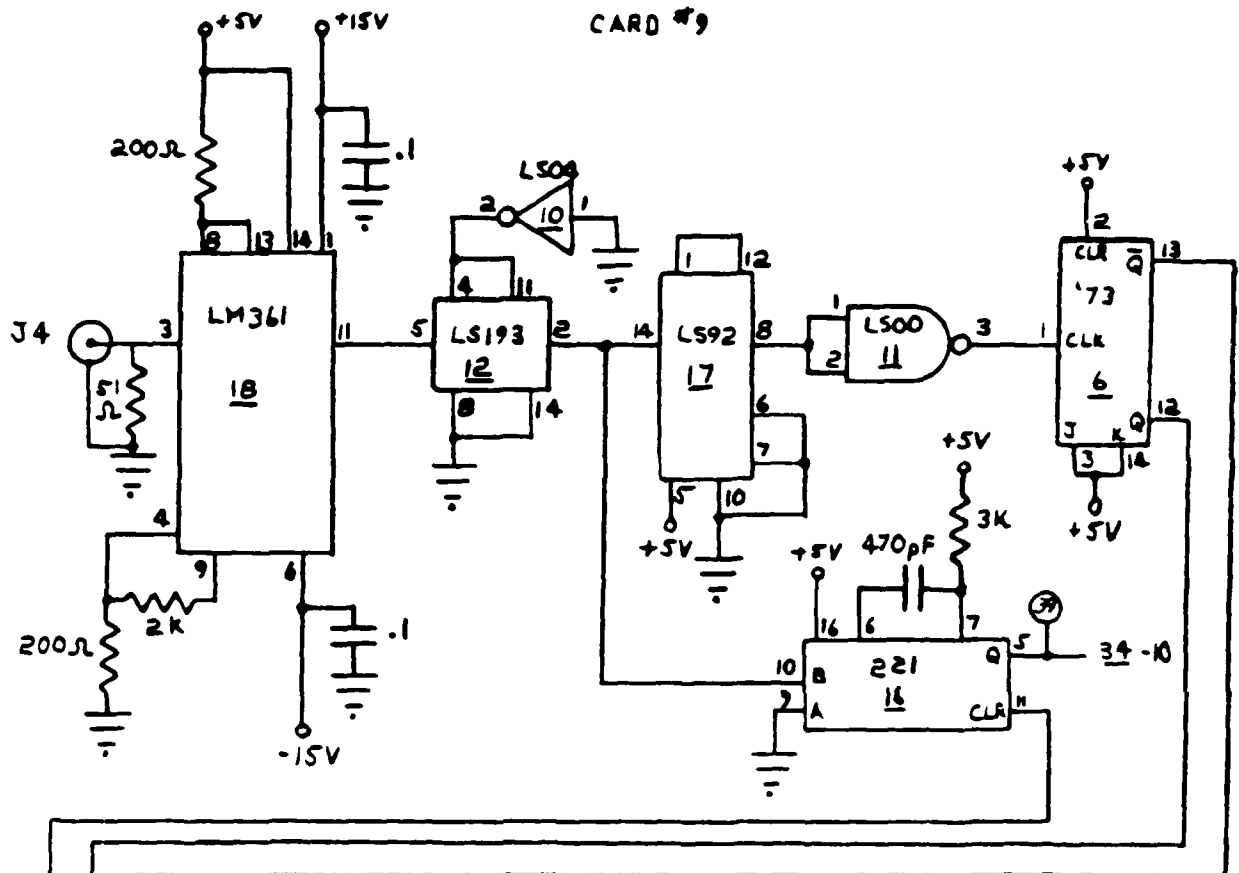
APPENDIX A

ELECTRONIC SCHEMATICS





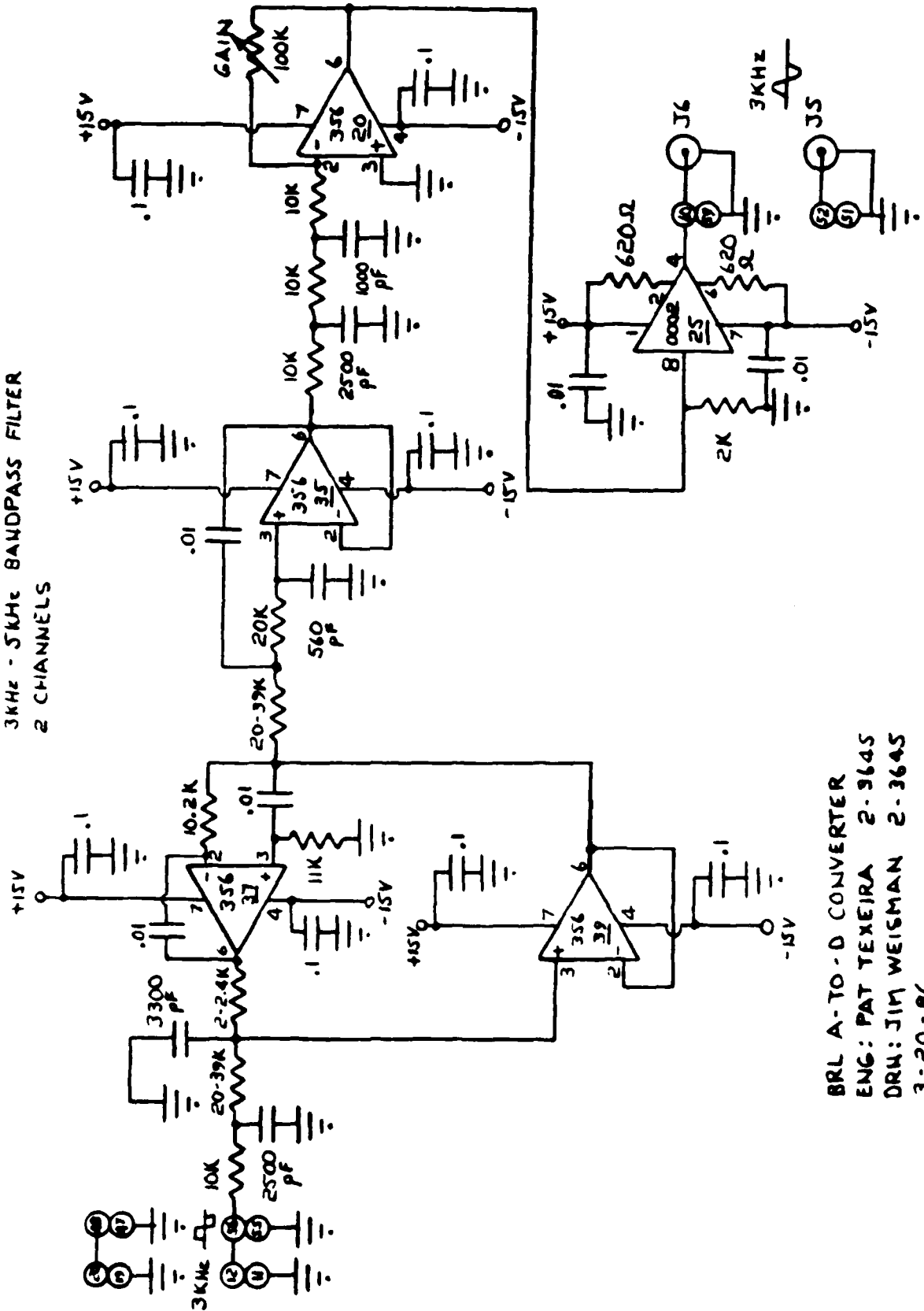
CARD #9



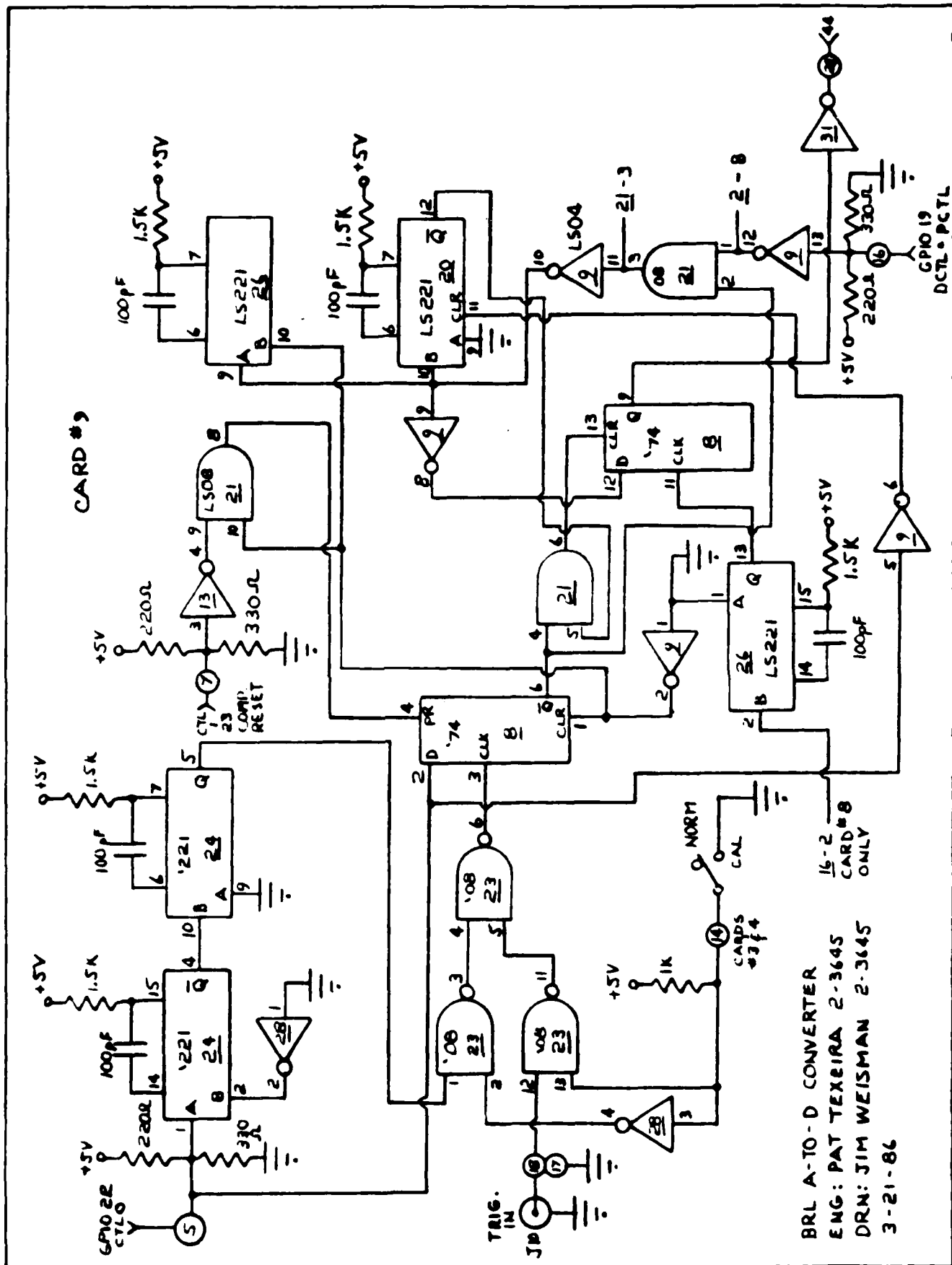
BRL A-TO-D CONVERTER
 ENG: PAT TEXEIRA 2-3645
 DRN: JIM WEISMAN 2-3645
 3-20-86

SHEET 3 OF 8

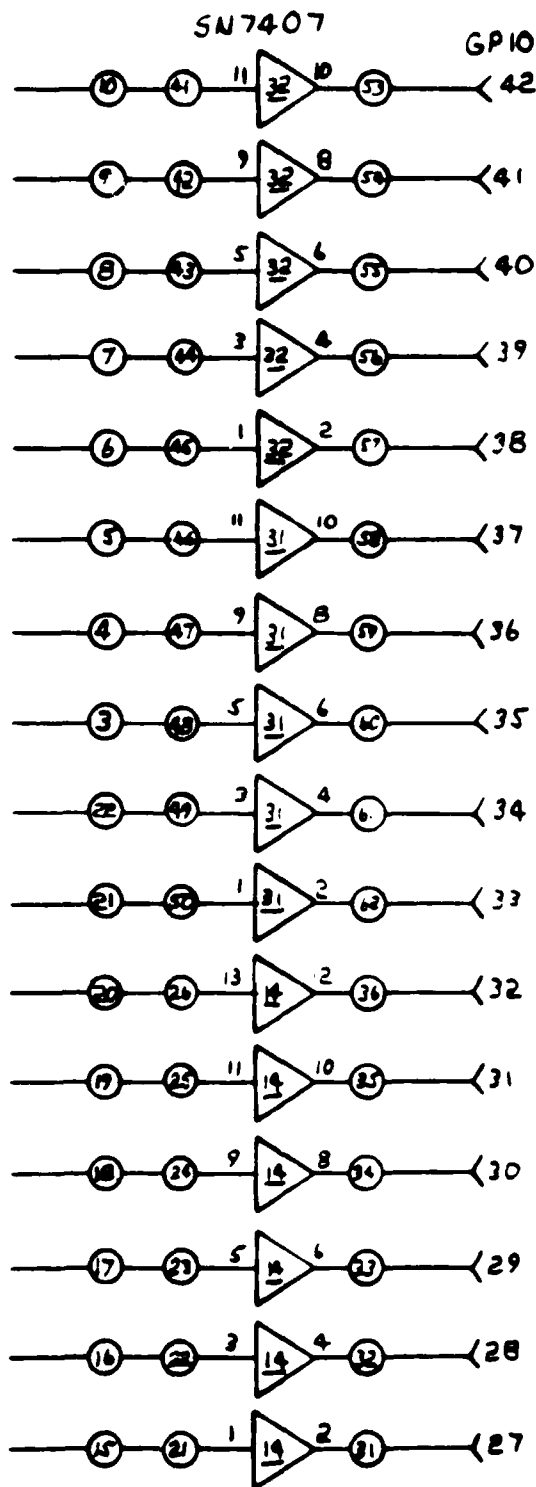
CARD #9
3KHz - 5KHz BANDPASS FILTER
2 CHANNELS



BRL A-TO-D CONVERTER
ENG: PAT TEXEIRA 2-3645
DRW: JIM WEISMAN 2-3645
3-20-86



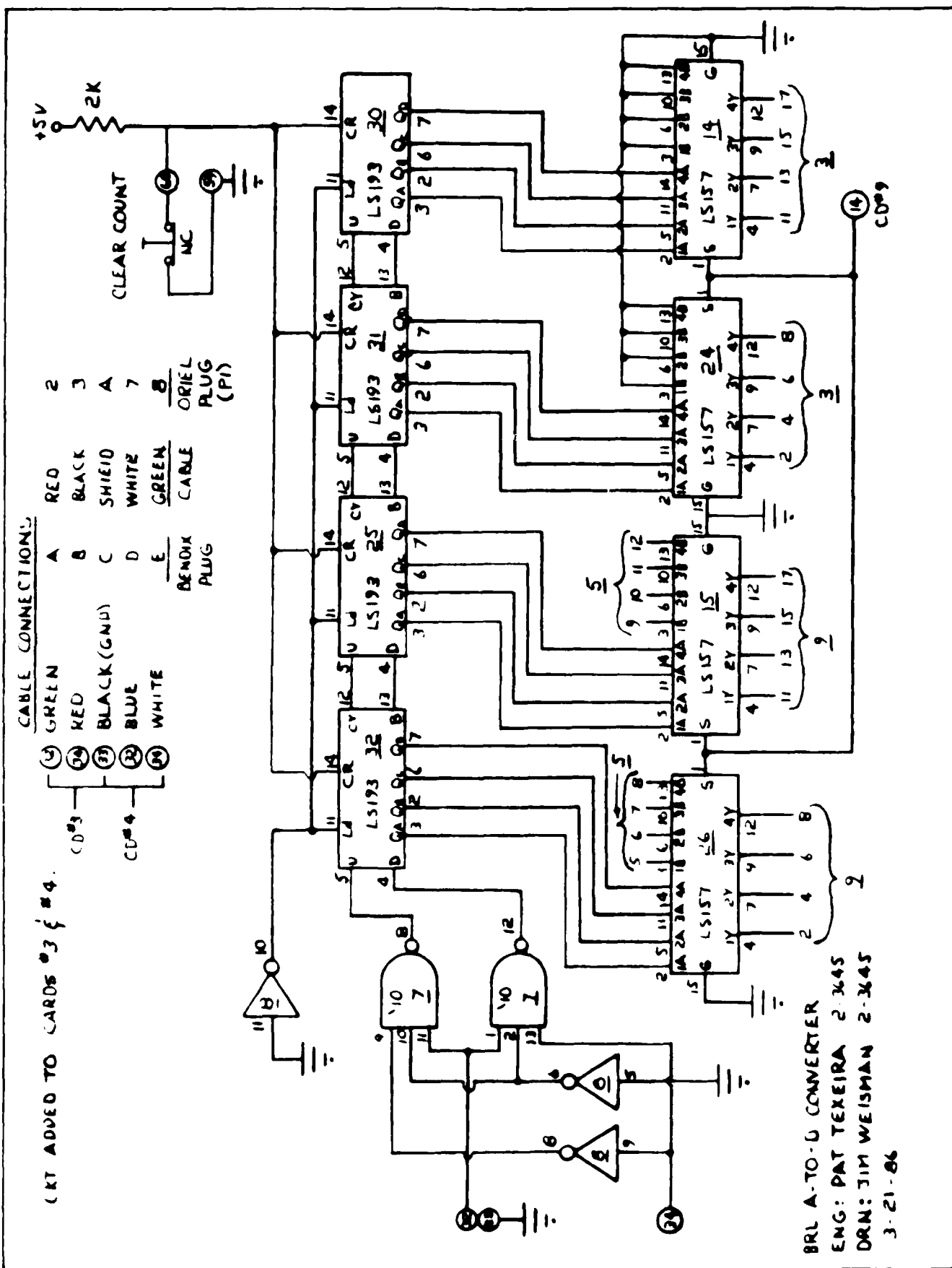
CARD # 9



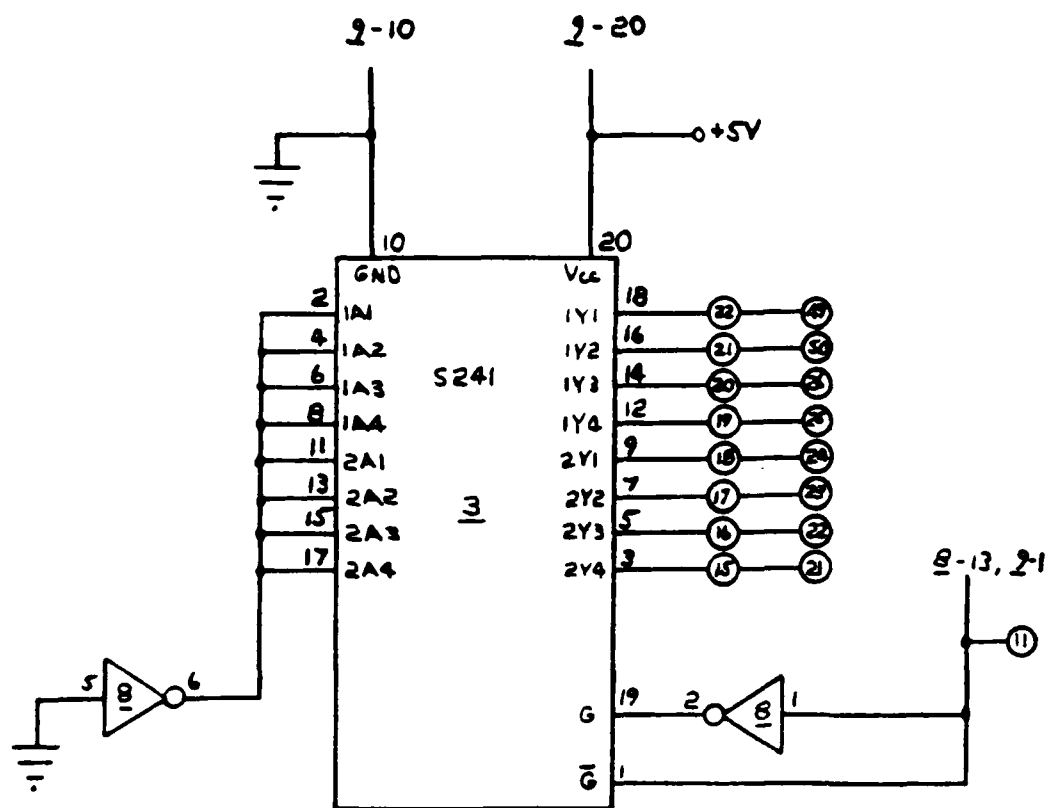
BRL A-TO-O CONVERTER
 ENG: PAT TEIXEIRA 2-3645
 DRN: JIM WEIGMAN 2-3645
 3-21-86

SHEET 6 OF 8

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MODIFICATION OF CARD #8



BRL A-TO-D CONVERTER
 ENG: PAT TEXEIRA 2-3645
 DRN: JIM WEISMAN 2-3645
 3-21-86

APPENDIX B

COMPUTER PROGRAM

```

10 !BRL_6_28
11 !-----
13 !PROGRAM FOR OBTAINING THE CALIBRATION CURVE FOR
14 !A GIVEN HOLOGRAM CORNER CUBE COMBINATION.
15 !-----
19 Comments: !This program uses the difference
20 !method for determining the amplitude
21 !for f2 & f1.
28 !The largest Num_samples is 32K.
31 !Graphing is included for 200 enve-
32 !lopes/graph. # of graphs depends on
33 !the number os Num_samples.
54 !Parameters of consideration: Ratios
55 !of f2/f1, degrees/count red(Dpcr),
56 !degrees/count blue(Dpcb)
57 !The doppler effect is included but
58 !not used(It was developed when this
59 !program was configured.). It uses
60 !the parabolic fit to determine the
61 !location of the peak using the greater
62 !of the amplitudes of f1 & f2. The
63 !doppler correction is indepentent for
64 !the red & blue.
66 !For a/d interface- reset for counter
67 !and normal cal switch, up for cal.
68 PRINTER IS 1
70 PRINT CHR$(12)
71 PRINT TABXY(30,15),"CALIBRATION PROGRAM"
72 WAIT 3
73 PRINT TABXY(30,15)," "
74 Initialize: !
75 PRINTER IS 1
76 PRINT CHR$(12) !CLEAR SCREEN
77 ON ERROR GOTO Err_rout
78 OPTION BASE 0
79 GRAPHICS OFF
80 RAD
81 ON KEY 4 LABEL "GRAPH".15 GOSUB Disp_out
82 ENABLE
83 Gpio=12
84 HpiB=704
85 !CONSTANTS
86 Sf=36000 !Sample frequency
87 Gf=3000 !galvo frequency set by clock
88 K1=12 !samples at f1
89 K2=6 !samples at f2
90 K3=4 !samples at f3
91 !Signal generator frequency 576.000Khz
92 !VARIABLE LIST
93 !Num_samp : number of samples to take at 36kz rate
94 !Fb : filter bandwidth
95 !N : INT(Sf/fb)
96 !Nnn : INT (Num_samples/48)*12, used in DMA
97 !Br( ) : array, sampled data, red
98 !Bl( ) : array, sampled data, blue
99 !If1 : interpolation factor for f1
100 !If2 : interpolation factor for f2
101 !F1( ) : convolution multipling function for f1
102 !F2( ) : convolution multipling function for f2
103 !Start : starting point, determined by N
104 !Qr : starting sampled data point for next K1

```

```

105      !Interval, including doppler correction, red.
106      !Qb : same as Qr except, blue
107      !Xx : Xx=1, loop for red;Xx=2 loop for blue
108      !B1( ) : Gimbal position for sample, red
109      !B2( ) : Gimbal position for sample, blue
110      !B1K1 : running sum of B1( ) in K1 interval
111      !B2K1 : running sum of B2( ) in K1 interval
112      !Filt_f1(J) : convolution elements in K1 interval, f1
113      !Filt_f2(J) : convolution elements in K1 interval, f2
114      !B1(P) : average of B1( ) in K1 interval
115      !B2(P) : average of B2( ) in k2 interval
116      !P : K1 interval of computation
117      !Ror : running sum of doppler correction, red
118      !Rob : running sum of doppler correction, blue
119      !Rotr(P) : running sum stored as a function of P
120      !Rotb(P) : running sum stored as a function of P
121      !Occr(P) : doppler correction for respective K1 interval, red
122      !Occb(P) : doppler correction for respective K1 interval, blue
123      !Qpsr : quadrant past, red
124      !Qpsb : quadrant past, blue
125      !Qptr : quadrant present, red
126      !Qptb : quadrant present, blue
127      !Qsr : running sum of quadrants, red
128      !Qsb : running sum of quadrants, blue
129      !Tar : total angle red, K1 interval
130      !Tab : total angle blue, K1 interval
131      !Ta_red(P) : total angle red, P K1 interval
132      !Ta_bl(P) : total angle blue. P K1 interval
133      !Whole quadrant update table
134      DIM Qt(5,4)
135      Qt(1,2)=1
136      Qt(1,4)=-1
137      Qt(2,1)=-1
138      Qt(2,3)=1
139      Qt(3,2)=-1
140      Qt(3,4)=1
141      Qt(4,1)=1
142      Qt(4,3)=-1
143      Qsr=10 !Quadrant sum red starts at 10
144      Qsb=10 !Quadrant sum blue starts at 10
145      ASSIGN @Gpio TO 12;WORD
146      !PRINT "      REGISTER STATUS BEFORE DMA"
147      !GOSUB Check_stat
148      Main:! Main program
149      PRINT CHR$(12)
150      GOSUB Num_samp      !# of samples to take
151      !GOSUB Test
152      !GOSUB File_data    ! Copy data from Buffer
153      !to Disk File.
154      !GOSUB File_array
155      GOSUB Filt_bw_int !Input filter bandwidth & interpolation factor
156      BEEP
157      INPUT "If calibration run, Type C. else, Type S, for stored",Aa$
158      IF Aa$="C" OR Aa$="c" THEN
159          GOSUB Dma_transfer
160          GOSUB File_data
161          GOTO 174
162      END IF
163      IF Aa$="S" OR Aa$="s" THEN
164          DISP "INSERT DATA DISK - PRESS CONT TO CONTINUE"
165          PAUSE

```

```

166     CAT
167     GOSUB File_array
168     GOTO 174
169 END IF
170     BEEP
171     BEEP
172     BEEP
173     GOTO 157
174     GOSUB Mult_fun      !Generate multiplier
175                         !functions.
176     GOSUB Comp_angle    !Convolution, envelop detection, doppler correction,
177                         !& Arctan for red and blue
178     DISP
179     GOSUB Disp_out
180     GOTO 1
181 !+++++
182 Num_samp: !Input the total number of samples
183           !to be taken. Maximum of 32K
184     BEEP
185     Num_samples=30000
186     GOTO 188
187     INPUT "TOTAL NUMBER OF SAMPLES TO BE TAKEN?".Num_samples
188     ALLOCATE REAL Ta_red(INT(Num_samples/K1)),Ta_bl(INT(Num_samples/K1)),Rot(I
NT(Num_samples/K1))
189     RETURN
190 Dma_transfer:
191     DISP "                COLLECTING DATA"
192     S=0
193     INTEGER Kk(0:47) BUFFER
194     ASSIGN @Buf TO BUFFER Kk(+)
195     !RESET DATA TRANSFER ELECTRONICS
196     CONTROL Gpio,2:2 !SET CTL0 & CTL1 TO 0
197     CONTROL Gpio,2:0 !SET CTL0 TO 1 & CTL1 TO 0
198     CONTROL Gpio,2:2 !SET CTL0 & CTL1 TO 1
199     !END RESET
200     CONTROL @Buf,4:0
201     CONTROL Gpio,0:2
202     CONTROL Gpio,3:0
203     !SET TRANSFER LOW
204     CONTROL Gpio,2:3 !SET CLT0 TO 0, XFER 0
205     !BEEP
206     TRANSFER @Gpio TO @Buf
207     FOR X=0 TO 47 STEP 4
208         Br(S)=Kk(X)-126
209         B1(S)=Kk(X+1)+128
210         B1(S)=Kk(X+2)+1
211         B2(S)=Kk(X+3)+1
212     ! PRINT S,Br(S),B1(S),B1(S),B2(S)
213     S=S+1
214 NEXT X
215 IF S=Num THEN
216     GOTO 220
217 ELSE
218     GOTO 196
219 END IF
220 CONTROL Gpio,2:2 !SET CLT0 & CLT1 TO 1
221 CONTROL Gpio,1:3
222 DISP
223 !PRINT "                REGISTER STATUS AFTER DMA"
224 !GOSUB Check_stat
225 BEEP

```

```

226 RETURN
246 Filt_bw_int:!!Input filter bandwidth in hertz
247 BEEP
250 INPUT "WHAT IS THE DESIRED FILTER BANDWIDTH IN HERTZ?",Fb
251 N=INT(Sf/Fb)
253 Nnn=INT(Num_samples/48)*12
254 ALLOCATE INTEGER Br(0:Nnn),INTEGER B1(0:Nnn)
255 ALLOCATE INTEGER B1(0:Nnn),INTEGER B2(0:Nnn)
256 ALLOCATE INTEGER Rotr(0:Nnn),Rotb(0:Nnn),Qccr(0:Nnn),Qccb(0:Nnn)
257 PRINT CHR$(12)
258 BEEP
262 INPUT "Real samples f1=12, interpolation factor?",If1
263 BEEP
264 INPUT "Real samples f2=6, interpolation factor?",If2
265 BEEP
266 ALLOCATE REAL Filt_f1(1:K1*If1),Filt_f2(1:K1*If2)
267 RETURN
268 Test:!! Generate data - needs to be modified for this
269 !program
272 BEEP
273 INPUT "What is the Envelop Peak Amplitude?",Epa
274 BEEP
275 DISP " GENERATING DATA"
276 Peak_value_data=0
277 Gal_cyc_frg=200
278 Ab=0
279 ALLOCATE INTEGER Mmm(0:Num_samples-1)
280 FOR X=0 TO Num_samples-1 STEP 4
281 Ac=Epa*SIN((PI*(+Ab))/(6*Gal_cyc_frg)+(2.7)*SIN(2*PI*Ab*Gf/Sf))
282 FOR Y=X TO X+3
283 Mmm(Y)=INT(Ac)+128
284 !PRINT Y,Mmm(Y)
285 OUTPUT @Buffer USING "#,B";Mmm(Y)
286 NEXT Y
287 Ab=Ab+1
288 NEXT X
289 DISP
290 GOSUB Check_stat
291 WAIT 5
292 RETURN
293 Mult_fun: !Generation of the multiplying function.
294 !Multiplying function will be (SIN(X)/(X))*COS(Y)
295 !PRINTER IS 705
296 !ALLOCATE REAL F1(0:(N*If1)),REAL F2(0:(N*If2))
297 DIM F1(0:2000),F2(0:2000)
298 RAD
299 BEEP
300 !Generate 1/2 multiplier function for F1
301 BEEP
302 DISP "Building multiplier function for F1"
303 F1(0)=1
304 !PRINT "F1(0)=";F1(0)
305 FOR P=1 TO N*If1
306 Aa=(PI*P)/(N*If1)
307 Bb=(Aa+2*Gf)/Fb
308 !PRINT "P=";P
309 !PRINT "N=";N*If1
310 !PRINT "(PI*P)/N=";Aa
311 !PRINT "(Aa+2*GF)/FB=";Bb
312 F1(P)=(SIN(Aa)/Aa)*COS(Bb)
313 !PRINT "F1(";P;")=";F1(P)

```

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314 NEXT P
315 DISP "Building multiplier function for F2"
316 !Generate 1/2 of multiplier function for F2
317 F2(0)=1
318 !PRINT "F2(0)=";F2(0)
319 FOR P=1 TO N*If2
320   Aa=(PI*P)/(N*If2)
321   Bb=(Aa*4*Gf)/Fb
322   !PRINT "P=";P
323   !PRINT "N=";N
324   !PRINT "(PI*P)/N=";Aa
325   !PRINT "(Aa*4*GF)/FB=";Bb
326   F2(P)=(SIN(Aa)/Aa)*COS(Bb)
327   !PRINT "F2(";P;")=";F2(P)
328 NEXT P
329 RETURN
330 !
331 !.....
332 !Convolution and envelop detection
333 !at K1 intervals for f1 and f2.
334 !.....
335 !
336 Comp_angle: !Filter, envelop detection,
337             !doppler correction and
338             !arctan for red & blue
339
340 RAD
341 Tot_angle_peak=0
342 P=0 !Initialization of array for the out-
343     !put angle for red and blue lasers.
344     !Determine first data point to start con-
345     !volution considering the value of N and
346     !the need of a + slope for f1 for synchr-
347     !onization.
348 Quotient=(N+1) DIV K1
349 Start=(Quotient+1)*12
350 BEEP
351 DISP "
352                                     C R U N C H I N G   D A T A "
353 PRINT "Start=";Start
354 Qq=0 !Initialize graph index. Increment
355     !Qq for each 200 computations
356
357 Qr=Start+2-3
358 Qb=Qr
359 FOR Xx=1 TO 2 !1=red,2=blue
360   IF Xx=1 THEN
361     Q=Qr
362   ELSE
363     Q=Qb
364   END IF
365   B1k1=0
366   B2k1=0
367 !.....
368 !.....
369 !.....
370 !.....
371 !.....
372 !.....
373 !.....
374 !.....
375 !.....
376 !.....
377 !.....
378 !.....
379 !.....
380 !.....
381 !.....
382 !.....
383 !.....

```

```

384      Sum_f1=0
385      IF Xx=1 THEN !Xx=1,Red:Xx=2,Blue
386          FOR M=X TO X+N
387              Product_f1=Br(M+1)*F1((M-X)*If1-C+If1)+Br(2*X-M)*F1((M-X)*If1+C)
388              Sum_f1=Sum_f1+Product_f1
389              ! PRINT Q:TAB(10);X:TAB(20);C:TAB(30);M:TAB(40);J:TAB(45);((M-X)*
If1+C);
390              ! PRINT TAB(50);((M-X)*If1-C+If1);
391              ! PRINT TAB(60);(2*X-M-1)
392          NEXT M
393      ELSE
394          FOR M=X TO X+N
395              Product_f1=B1(M+1)*F1((M-X)*If1-C+If1)+B1(2*X-M)*F1((M-X)*If1+C)
396              Sum_f1=Sum_f1+Product_f1
397              ! PRINT Q:TAB(10);X:TAB(20);C:TAB(30);M:TAB(40);J:TAB(45);((M-X)
*If1+C);
398              ! PRINT TAB(50);((M-X)*If1-C+If1);
399              ! PRINT TAB(60);(2*X-M-1)
400          NEXT M
401      END IF
402      Filt_f1(J)=Sum_f1
403      !PRINTER IS 705
404      !PRINT "Filt_f1(";J;"=");Filt_f1(J)
405      !PRINTER IS 1
406      IF ABS(Filt_f1(J))>Peak_value_f1 THEN
407          Peak_value_f1=ABS(Filt_f1(J))
408      END IF
409      J=J+1
410  NEXT C
411  NEXT X
412      IF Xx=1 THEN B1(P)=INT(B1k1/12)
413      IF Xx=2 THEN B2(P)=INT(B2k1/12)
414      !.....
420  Env_f1: !Envelop detection over k1
421      !interval for f1
422      !.....
423      Diff_plus_f1=0
424      Diff_minus_f1=0
425      FOR E=If1+1 TO 5*If1+1
426          Diff_f1=Filt_f1(E)-Filt_f1(E+INT(If1*P/2))
427          IF Diff_f1>Diff_plus_f1 THEN
428              Diff_plus_f1=Diff_f1
429          END IF
430          IF Diff_f1<Diff_minus_f1 THEN
431              Diff_minus_f1=Diff_f1
432          END IF
433          !PRINT "e=";E;"Diff_f1=";Diff_f1
434          !PRINT "Diff_plus_f1=";Diff_plus_f1
435          !PRINT "Diff_minus_f1=";Diff_minus_f1
436      NEXT E
437      !Value of Envelop at P interval for f1
438      IF Diff_plus_f1>ABS(Diff_minus_f1) THEN
439          Envelop_f1=Diff_plus_f1
440      ELSE
441          Envelop_f1=Diff_minus_f1
442      END IF
478  !.....
479  Con_f2: !
480      !Convolution over k1 interval for f1
481      !.....
483      J=1

```

```

484 Peak_value_f2=0
485 FOR X=Q TO Q+K1-1
486   FOR C=0 TO If2-1
487     Sum_f2=0
488     IF Xx=1 THEN !Xx=1,Red:Xx=2,Blue
489       FOR M=X TO X+N
490         Product_f2=Br(M+1)*F2((M-X)*If2-C+If2)+Br(2*X-M)*F2((M-X)*If2+C)
491         Sum_f2=Sum_f2+Product_f2
492         ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)
+If2+C);
493         ! PRINT TAB(50);((M-X)*If2-C+If2);
494         ! PRINT TAB(60);(2*X-M-1)
495       NEXT M
496     ELSE
497       FOR M=X TO X+N
498         Product_f2=Bl(M+1)*F2((M-X)*If2-C+If2)+Bl(2*X-M)*F2((M-X)*If2+C)
499         Sum_f2=Sum_f2+Product_f2
500         ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)
+If2+C);
501         ! PRINT TAB(50);((M-X)*If2-C+If2);
502         ! PRINT TAB(60);(2*X-M-1)
503       NEXT M
504     END IF
505     Filt_f2(J)=Sum_f2
506     !PRINTER IS 705
507     !PRINT "Filt_f2(";J;")=";Filt_f2(J)
508     !PRINTER IS 1
509     IF ABS(Filt_f2(J))>Peak_value_f2 THEN
510       Peak_value_f2=ABS(Filt_f2(J))
511     END IF
512     J=J+1
513   NEXT C
514 NEXT X
515
520 !.....
521 Env_f2: Envelop detection over fract-
522       tional interval of K1 for f2
523       using 1/2 cycle difference
524 !.....
525 Diff_plus_f2=0
526 Diff_minus_f2=0
527 FOR E=If2+1 TO 5*If2+1
528   Diff_f2=Filt_f2(E)-Filt_f2(E+3*If2)
529   IF Diff_f2>Diff_plus_f2 THEN
530     Diff_plus_f2=Diff_f2
531   END IF
532   IF Diff_f2<Diff_minus_f2 THEN
533     Diff_minus_f2=Diff_f2
534   END IF
535 NEXT E
536 !value of Envelop at P interval for f2
537 IF Diff_plus_f2>ABS(Diff_minus_f2) THEN
538   Envelop_f2=Diff_plus_f2
539 ELSE
540   Envelop_f2=-Diff_minus_f2
541 END IF
542 !PRINT "Envelop_f2";Envelop_f2
543 !.....
544 !.....
545 !.....
546 !.....
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999 !.....
1000 !.....

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558     FOR E=If1+1 TO 5*If1+1
560         D2=Filt_f1(E+2)-Filt_f1(E+1)
561         D1=Filt_f1(E+1)-Filt_f1(E)
562         !PRINT "D2=";D2,"D1=";D1,"(D2-D1)=";D2-D1,"E=";E
563         IF D2=0 OR SGN(D1)+SGN(D2)=0 THEN
564             GOTO 571
565         END IF
566     NEXT E
567     Dsq=D2-D1
568     Ap=Filt_f1(E)
569     Bp=D1-Dsq/2
570     Cp=Dsq/2
571     Envelop_f11=(Ap-(Bp^2)/(4*Cp))
572     !PRINT "Envelop_f11=";Envelop_f11
573     Ee=E-Bp/(2*Cp)
574     !PRINTER IS 705
575     PRINT "Ee_f1=";Ee
576     PRINTER IS 1
577     Qc=0
578     IF Ee>4*If1+1 THEN Qc=-1
579     IF Ee<2*If1+1 THEN Qc=+1
580 ELSE
581     FOR E=If2+1 TO 5*If2+1
582         D2=Filt_f2(E+2)-Filt_f2(E+1)
583         D1=Filt_f2(E+1)-Filt_f2(E)
584         !PRINT "D2=";D2,"D1=";D1,"(D2-D1)=";D2-D1,"E=";E
585         IF D2=0 OR SGN(D1)+SGN(D2)=0 THEN
586             GOTO 626
587         END IF
588     NEXT E
589     Dsq=D2-D1
590     Ap=Filt_f2(E)
591     Bp=D1-Dsq/2
592     Cp=Dsq/2
593     Envelop_f22=(Ap-(Bp^2)/(4*Cp))
594     !PRINT "Envelop_f22=";Envelop_f22
595     Ee=E-Bp/(2*Cp)
596     !PRINTER IS 705
597     PRINT "Ee_f2=";Ee
598     PRINTER IS 1
599     Qc=0
600     IF Ee>4*If2+1 THEN Qc=-1
601     IF Ee<2*If2+1 THEN Qc=+1
602 END IF
603 IF Xx=1 THEN
604     !PRINTER IS 705
605     Ror=Ror+Qc
606     PRINT "Ror=";Ror,
607     Rotr(P)=Ror
608     Qccr(P)=Qc
609     PRINT "Qc=";Qc
610     Or=Or+Qc+K1
611     PRINT "Or=";Or
612     PRINTER IS 1
613 ELSE
614     Rob=Rob+Qc
615     PRINT "Rob=";Rob
616     Rotb(P)=Rob
617     Qccb(P)=Qc
618     PRINT "Qc=";Qc,
619     Qb=Qb+Qc+K1

```

```

661      PRINT "Qb=";Qb
662    END IF
663    E_f1=Envelop_f1
664    E_f2=4*Envelop_f2
665    PRINT "Envelop_f2/Envelop_f1=";Envelop_f2/Envelop_f1
666    !.....
667    Arc_t: !Computation of the angle from the
668    !arctan(Envelop_f2/Envelop_f1) and
669    !counting of the whole quadrants
670    !.....
671    BEEP 500,.2
672    PRINTER IS 1
673    DEG
674    IF Xx=1 THEN ! Arc_t for Red laser
675      IF E_f1=0 THEN !Take care of divide by zero
676        IF Qpsr=1 OR Qpsr=3 THEN
677          Tar=(Qsr)*90+90
678          GOTO 716
679        ELSE
680          Tar=(Qsr)*90
681          GOTO 716
682        END IF
683      END IF
684      Ra=E_f2/E_f1
685      !Determine Quadrant & total_angle(P)_
686      IF ABS(Ra)>=60 THEN
687        IF SGN(Ra)=1 THEN
688          Ra=60
689        ELSE
690          Ra=-60
691        END IF
692      END IF
693      IF SGN(Ra)=1 THEN !Plus Ratio
694        IF SGN(E_f1)=1 THEN ! Deno plus
695          Qptr=1
696        ELSE
697          Qptr=3
698        END IF
699        Tar=(Qt(Qpsr,Qptr)+Qsr)*90+ATN(Ra)-900
700      ELSE
701        IF SGN(E_f1)=1 THEN
702          Qptr=4
703        ELSE
704          Qptr=2
705        END IF
706        Tar=(Qt(Qpsr,Qptr)+Qsr)*90+90+ATN(Ra)-900
707      END IF
708      Qsr=Qt(Qpsr,Qptr)+Qsr
709      IF P>1 THEN Qpsr=Qptr
710      PRINTER IS 705
711      PRINT P,PROUND(Tar,-2),Qpsr,Qptr,B1(Q),B1(P)
712      PRINTER IS 1
713      Ta_red(P)=Tar
714      PRINTER IS 705
715      PRINT P,"Ta_red=";PROUND(Ta_red(P),-2),"E_F1=";E_f1,"E_F2=";E_f2
716      PRINTER IS 1
717    ELSE !Arc_t for blue laser
718      IF E_f1=0 THEN !Take care of divide by zero
719        IF Qpsb=1 OR Qpsb=3 THEN
720          Tab=(Qsb)*90+90
721          GOTO 763

```

```

733     ELSE
734         Tab=(Qsb)*90
735         GOTO 763
736     END IF
737 END IF
738 Ra=E_f2/E_f1
739 !Determine Quadrant & Total_angle(P)_
740 IF ABS(Ra)>=60 THEN
741     IF SGN(Ra)=1 THEN
742         Ra=60
743     ELSE
744         Ra=-60
745     END IF
746 END IF
747 IF SGN(Ra)=1 THEN !Plus Ratio
748     IF SGN(E_f1)=1 THEN ! Deno plus
749         Optb=1
750     ELSE
751         Optb=3
752     END IF
753     Tab=(Qt(Qpsb,Optb)+Qsb)*90+ATN(Ra)-900
754 ELSE
755     IF SGN(E_f1)=1 THEN
756         Optb=4
757     ELSE
758         Optb=2
759     END IF
760     Tab=(Qt(Qpsb,Optb)+Qsb)*90+90+ATN(Ra)-900
761 END IF
762 Qsb=Qt(Qpsb,Optb)+Qsb
763 IF P>1 THEN Qpsb=Qptb
764 PRINTER IS 705
765 PRINT P,PROUND(Tab,-2),Qpsb,Optb,B1(Q),B1(P)
766 Ta_b1(P)=Tab
767 PRINT P,"Ta_b1=";PROUND(Ta_b1(P),-2),"E_F1=";E_f1,"E_F2=";E_f2
768 END IF!For red or blue
769 PRINTER IS 1
770 NEXT Xx
771 RAD
772 PRINT
773 !This completes the angle computation
774 !for the red and blue for this K1 interval P
775 P=P+1
776 !Test to increment Qq or not
777 IF P=0 THEN
778     Qq=0
779 ELSE
780     Rmdr=P MOD 199      !P starts at 0
781     IF Rmdr=0 THEN Qq=Qq+1
782 END IF
783 INEXT Q
784 IF Q>Nnn-N-25-Start THEN RETURN
785 GOTO 363
786 Disp_out:!!Graph data
787 Dpcr=.064          !Degrees/count gimbal red
788 Dpcb=.064          !Degrees/count gimbal blue
789 DUMP DEVICE IS 705
790 PRINTER IS 705
791 PEN 2
792 PRINT CHR$(12);    !Clear alpha & form feed
793 DISP

```

```

796 GINIT !Initialize
797 GRAPHICS ON !Raster on
798 FOR Qqq=0 TO Qq !Graph index
799 FOR R=1 TO 2 !red=1, blue=2
800 PEN 2
801 WINDOW -25,200,-500,500!Set window
802 AXES 20,100
803 LORG 6
804 FOR I=0 TO 180 STEP 40
805 MOVE I,0
806 LABEL Qqq*200+I
807 NEXT I
808 !
809 LORG 8
810 FOR I=-400 TO 400 STEP 100
811 MOVE 0,I
812 LABEL I
813 NEXT I
814 !
815 LORG 5
816 MOVE 100,475
817 PEN 6
818 LABEL "ANGLE AS A FUNCTION OF ENVELOP SAMPLES"
819 MOVE 180,425
820 IF R=1 THEN
821 PEN 2
822 LABEL "RED"
823 MOVE 180,400
824 LABEL Name$
825 PEN 6
826 END IF
827 IF R=2 THEN LABEL "BLUE"
828 MOVE 180,-400
829 LABEL Name$
830 PEN 2
831 MOVE -13,0
832 LORG 6
833 LABEL "Samp"
834 LORG 5
835 MOVE 100,-475
836 LABEL Title$:"ENVELOP SAMPLE RATE, 3KHZ"
837 PEN 0
838 MOVE 0,0
839 PEN 2
840 FOR X=0 TO 199 !graph data
841 IF R=1 THEN
842 Tb=Tb_red(Qqq*200+X+4)-Ta_red+4)
843 ELSE
844 Tb=Ta_b1(Qqq*200+X+4)-Ta_b1(4)
845 END IF
846 DRAW X,Tb
847 NEXT X
848 PEN 0
849 MOVE 0,0
850 PEN 6
851 FOR X=0 TO 199 !Graph dimbal
852 IF R=1 THEN
853 Bbb=B1(Qqq*200+X+4)*Dpcc-B1(4)*Dpcc
854 ELSE
855 Bbb=B2(Qqq*200+X+4)*Dpcc-B2(4)*Dpcc
856 END IF

```

```

860      DRAW X,-Bbb
861      NEXT X
862      PEN 0
863      PAUSE
864      !DUMP GRAPHICS
865      !PRINT CHR$(12)
866      GCLEAR
867      NEXT K
868      NEXT Qqq
869      ALPHA ON
870      PRINTER IS 1
871      DISP "STOP"
872      PAUSE
873      RETURN
874      Check_stat: !Status check on Buffer.
875      PRINT "      REGISTER STATUS @Buffer"
876      FOR X=1 TO 5
877          STATUS @Buffer,X:A
878          PRINT "REGISTER ":X,A
879      NEXT X
880          !DISP "PAUSE - PRESS CONTINUE TO CONTINUE"
881      BEEP
882          !PAUSE
883      WAIT 5
884      PRINT CHR$(12)
885      RETURN
886      Err_rout: !Routine for error recovery
887      BEEP
888      BEEP
889      BEEP
890      DISP ERRMS
891      PAUSE
892      GOSUB Disp_out
893      PAUSE
894      !
895      File_data: !Transfer of data from Buffer
896                  !to disk
897      !
898      BEEP
899      DISP "INSERT A FORMATTED DATA DISK!!!!!! -- PRESS CONT TO CONT"
900      PAUSE
901      !Create a BDAT file for the data.
902      INPUT "NAME OF DATA FILE TO BE CREATED?".Names
903      IF Names="" THEN 901
904      BEEP
905      DISP "CREATING A DATA FILE"
906      !Record length is 1
907      CREATE BDAT Names$:INTERNAL".8*Nnn+4,2
908      !Assign an I/O path
909      ASSIGN @File TO Names$:INTERNAL"
910      GOSUB Check_stat_r
911      PRINT CHR$(12)
912      DISP "TRANSFERRING DATA FROM BUFFER TO FILE"
913      OUTPUT @File;B1(*),B1(*),B1(*),B2(*)
914      ASSIGN @File TO *
915      WAIT 5
916      DISP "TRANSFER COMPLETE"
917      WAIT 5
918      DISP
919      PRINT CHR$(12)
920      RETURN

```

```

921      !
922 File_array: !Transfer from file to array
923      !
924      INPUT "NAME OF DATA FILE? ",Name$
925      ASSIGN @File TO Name$:INTERNAL"
926      BEEP
929      DISP "TRANSFERRING FROM DATA FILE TO AN ARRAY"
930      ENTER @File;Br(*),B1(*),B1(*),B2(*)
931      PRINT CHR$(12)
933      DISP "TRANSFER COMPLETE"
934      WAIT 3
938      !GOSUB Check_stat
939      !GOSUB Check_stat_r
940      !GOSUB Check_stat
941      DISP
942      RETURN
943 Check_stat_r: !Status check on File.
944      BEEP
945      PRINT "          REGISTER STATUS @File"
946      FOR R=1 TO 5
947          STATUS @File,R:A
948          PRINT "REGISTER ";R,A
949      NEXT R
950          !DISP "PAUSE - PRESS CONTINUE TO CONTINUE"
951      BEEP
952          !PAUSE
953      WAIT 5
954      PRINT CHR$(12)
955      RETURN
956      END

```

Cross Reference <<<<

* Numeric Variables

| | | | | | | | | | |
|---------------|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| A | 877 | 878 | 947 | 948 | | | | | |
| A_to_d_reset | | | | | | | | | |
| Aa | 306 | 307 | 312 | 320 | 321 | 326 | | | |
| Ab | 278 | 281 | 287 | | | | | | |
| Ac | 281 | 283 | | | | | | | |
| Ap | 572 | 577 | 627 | 630 | | | | | |
| B1 | 210 | 255 | 381 | 412 | 719 | 766 | 856 | 913 | 930 |
| B1_peak | | | | | | | | | |
| B1k1 | 369 | 381 | 412 | | | | | | |
| B2 | 211 | 255 | 382 | 413 | 858 | 913 | 930 | | |
| B2_peak | | | | | | | | | |
| B2k1 | 370 | 382 | 413 | | | | | | |
| Bb | 307 | 312 | 321 | 326 | | | | | |
| Bb2 | | | | | | | | | |
| Bbb | 856 | 858 | 860 | | | | | | |
| Bbb2 | | | | | | | | | |
| B1 | 209 | 254 | 395 | 498 | 913 | 930 | | | |
| b1 | 574 | 577 | 580 | 628 | 630 | 631 | | | |
| b1 | 208 | 254 | 387 | 490 | 913 | 930 | | | |
| butter array | | | | | | | | | |
| bu_label | | | | | | | | | |
| | 383 | 387 | 395 | 410 | 486 | 490 | 498 | 513 | |
| check_stat | | | | | | | | | |
| check_stat_r | | | | | | | | | |
| Comp_and_r | | | | | | | | | |
| comp_t1 | | | | | | | | | |
| c | 575 | 577 | 580 | 629 | 630 | 631 | | | |
| c | | | | | | | | | |
| c | | | | | | | | | |
| c | | | | | | | | | |
| c | | | | | | | | | |
| c | 561 | 562 | 571 | 574 | 619 | 621 | 626 | 637 | |
| c | 561 | 563 | 571 | 618 | 621 | 626 | | | |
| c | 426 | 427 | 428 | 430 | 431 | | | | |
| c | 529 | 530 | 531 | 533 | 534 | | | | |
| Diff t1 | 424 | 430 | 431 | 438 | 441 | | | | |
| Diff minus t1 | 527 | 533 | 534 | 541 | 544 | | | | |
| Diff minus t1 | 423 | 427 | 428 | 438 | 439 | | | | |
| Diff plus t1 | 526 | 530 | 531 | 541 | 542 | | | | |
| Disp con | | | | | | | | | |
| Disp data | | | | | | | | | |
| Disp data 1 | | | | | | | | | |
| Disp data 2 | | | | | | | | | |
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|-----------------|-----|-------|-----|-----|-----|-----|-----|-----|-----|-----|
| Envelop_f22 | 630 | | | | | | | | | |
| Epa | 273 | 281 | | | | | | | | |
| Err_rout | | | | | | | | | | |
| F1 | 297 | <-DEF | 303 | 312 | 387 | 395 | | | | |
| F2 | 297 | <-DEF | 317 | 326 | 490 | 498 | | | | |
| Fb | 250 | 251 | 307 | 321 | | | | | | |
| File_array | | | | | | | | | | |
| File_data | | | | | | | | | | |
| Filt_bw_int | | | | | | | | | | |
| Filt_f1 | 266 | 402 | 406 | 407 | 426 | 560 | 561 | 572 | | |
| Filt_f11b | | | | | | | | | | |
| Filt_f11r | | | | | | | | | | |
| Filt_f1b | | | | | | | | | | |
| Filt_f1r | | | | | | | | | | |
| Filt_f2 | 266 | 505 | 509 | 510 | 529 | 618 | 619 | 627 | | |
| Filt_f22b | | | | | | | | | | |
| Filt_f22r | | | | | | | | | | |
| Filt_f2b | | | | | | | | | | |
| Filt_f2r | | | | | | | | | | |
| Gal_cyc_frg | 277 | 281 | | | | | | | | |
| Gf | 87 | 281 | 307 | 321 | | | | | | |
| Gpio | 83 | 196 | 197 | 198 | 201 | 202 | 204 | 220 | 221 | |
| Hpi0 | 84 | | | | | | | | | |
| I | 804 | 805 | 806 | 807 | 810 | 811 | 812 | 813 | | |
| I1 | 262 | 266 | 305 | 306 | 383 | 387 | 395 | 425 | 426 | 55 |
| I2 | 594 | 595 | | | | | | | | |
| I2 | 264 | 266 | 319 | 320 | 486 | 490 | 498 | 528 | 529 | 617 |
| I3 | 639 | 640 | | | | | | | | |
| I4 | 378 | 402 | 406 | 407 | 409 | 483 | 505 | 509 | 510 | 512 |
| I5 | 88 | 188 | 266 | 348 | 380 | 426 | 485 | 650 | 660 | |
| I6 | 89 | | | | | | | | | |
| I7 | 90 | | | | | | | | | |
| I8 | 192 | <-DEF | 194 | 208 | 209 | 210 | 211 | | | |
| I9 | 386 | 387 | 392 | 394 | 395 | 400 | 489 | 490 | 495 | 497 |
| I10 | 498 | 503 | | | | | | | | |
| Max_x_axes | | | | | | | | | | |
| Mnn | 279 | 283 | 285 | | | | | | | |
| Mn | | | | | | | | | | |
| Mult_fun | | | | | | | | | | |
| N | 251 | 305 | 306 | 319 | 320 | 348 | 386 | 394 | 489 | 497 |
| Nnn | 785 | | | | | | | | | |
| Ns | 215 | 253 | 254 | 255 | 256 | 785 | 907 | | | |
| Num_samp | | | | | | | | | | |
| Num_samples | 185 | 187 | 188 | 253 | 279 | 280 | | | | |
| Op | | | | | | | | | | |
| O1 | 205 | 206 | 312 | 314 | 319 | 320 | 326 | 328 | 347 | 41 |
| O2 | 413 | 647 | 648 | 657 | 658 | 716 | 719 | 722 | 725 | 72 |
| O3 | 766 | 767 | 768 | 776 | 778 | 781 | | | | |
| Peak_value_data | 276 | | | | | | | | | |
| Peak_value_f1 | 379 | 406 | 407 | | | | | | | |
| Peak_value_f11b | | | | | | | | | | |
| Peak_value_f11r | | | | | | | | | | |
| Peak_value_f1b | | | | | | | | | | |
| Peak_value_f1r | | | | | | | | | | |
| Peak_value_f2 | 484 | 509 | 510 | | | | | | | |
| Peak_value_f22b | | | | | | | | | | |
| Peak_value_f22r | | | | | | | | | | |
| Peak_value_f2b | | | | | | | | | | |
| Peak_value_f2r | | | | | | | | | | |
| Product_f1 | 387 | 388 | 395 | 396 | | | | | | |

* I/O Path Names

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| @Buf | 194 | 200 | 206 | | | |
| @Buffer | 285 | 877 | | | | |
| @File | 909 | 913 | 914 | 925 | 930 | 947 |
| @Gpio | 145 | 206 | | | | |

* Line Labels

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|--------------|-----|-------|-------|-------|--|-----|
| A_to_d_reset | | | | | | |
| Arc_t | 672 | <-DEF | | | | |
| Buffer_array | | | | | | |
| Bw_label | | | | | | |
| Check_stat | 290 | 874 | <-DEF | | | |
| Check_stat_r | 910 | 943 | <-DEF | | | |
| Comments | 19 | <-DEF | | | | |
| Comp_angle | 176 | 336 | <-DEF | | | |
| Con_f1 | 374 | <-DEF | | | | |
| Con_f2 | 479 | <-DEF | | | | |
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| Disp_con | | | | | | |
| Disp_data | | | | | | |
| Disp_data_1 | | | | | | |
| Disp_data_2 | | | | | | |
| Disp_out | 81 | 179 | 788 | <-DEF | | 892 |
| Dma_transfer | 159 | 190 | <-DEF | | | |
| Doppler_cor | | | | | | |
| Env_f1 | 420 | <-DEF | | | | |
| Env_f2 | 521 | <-DEF | | | | |
| Err_rout | 77 | 886 | <-DEF | | | |
| File_array | 167 | 922 | <-DEF | | | |
| File_data | 160 | 895 | <-DEF | | | |
| Filt_buf_int | 155 | 246 | <-DEF | | | |
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| Initialize | 74 | <-DEF | | | | |
| Main | 148 | <-DEF | | | | |
| Mult_fun | 174 | 293 | <-DEF | | | |
| Num_samp | 150 | 180 | <-DEF | | | |
| Range_stored | | | | | | |
| Test | 268 | <-DEF | | | | |
| Test_data | | | | | | |
| Var_list | | | | | | |

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116 RAD
117 ON KEY 4 LABEL "GRAPH",15 GOSUB Disp_Glt
118 ENABLE
119 Gpio=12
120 HpiB=704
121 !::::::::::::::::::::::::::::::::::::::::::::::::::
122 Constants:
123 !
124 !-----
125 Sf=36000 !Sample frequency
126 Gf=3000 !galvo frequency set by clock
127 K1=12 !samples at f1
128 K2=6 !samples at f2
129 K3=4 !samples at f3
130 Sys_del_red=-0!-3 for data acquisition
131 !+0 for Test_data
132 Sys_del_bl=-0 !-3 for data acquisition
133 !+0 for Test_data
134 !Signal generator frequency 576.000kHz
135 !::::::::::::::::::::::::::::::::::::::::::::::::::
136 Var_list: !Variable list
137 !-----
138 !Num_samp : number of samples to take at 36kHz rate
139 !Fb : filter bandwidth
140 !N : INT(Sf/fb)
141 !Ns : INT (Num_samples/4)
142 !Br( ) : array, sampled data, red
143 !Bl( ) : array, sampled data, blue
144 !If1 : interpolation factor for f1
145 !If2 : interpolation factor for f2
146 !F1( ) : convolution multiplying function for f1
147 !F2( ) : convolution multiplying function for f2
148 !Start : starting point, determined by N
149 !Qr : starting sampled data point for next K1
150 !interval, including doppler correction, red.
151 !Qb : same as Qr except, blue
152 !Xx : Xx=1, loop for red;Xx=2 loop for blue
153 !B1( ) : spare input
154 !B2( ) : spare input
155 !Filt_f1(J) : convolution elements in K1 interval, f1
156 !Filt_f2(J) : convolution elements in K1 interval, f2
157 !B1(P) : average of B1( ) in K1 interval
158 !B2(P) : average of B2( ) in K2 interval
159 !P : K1 interval of computation
160 !Ror : running sum of doppler correction, red
161 !Rob : running sum of doppler correction, blue
162 !Rotr(P) : running sum stored as a function of P
163 !Rotb(P) : running sum stored as a function of P
164 !Occr(P) : doppler correction for respective K1 interval, red
165 !Occb(P) : doppler correction for respective K1 interval, blue
166 !Qpsr : quadrant past, red
167 !Qpsb : quadrant past, blue
168 !Qptr : quadrant present, red
169 !Qptb : quadrant present, blue
170 !Qsr : running sum of quadrants, red
171 !Qsb : running sum of quadrants, blue
172 !Tar : total angle red, K1 interval
173 !Tab : total angle blue, K1 interval
174 !Ta_red(P) : total angle red, P K1 interval
175 !Ta_bl(P) : total angle blue, P K1 interval
176 !Whole quadrant update table

```

```

180 DIM Q(5,4)
181 Q(1,2)=1
182 Q(1,4)=1
183 Q(2,1)=1
184 Q(2,3)=1
185 Q(3,2)=1
186 Q(3,4)=1
187 Q(4,1)=1
188 Q(4,3)=1
189 Qsr=10 !Quadrant sum red starts at 10
190 Qsb=10 !Quadrant sum blue starts at 10
191 ASSIGN @Gpio TO 12
192 !PRINT "REGISTER STATUS BEFORE DMA"
193 !GOSUB Check_stat
194 !::::::::::::::::::::::::::::::::::::::::::::::::::
195 Main: ! Main program
196 !
197 PRINT CHR$(12)
198 GOSUB A_to_d_reset
199 GOSUB Range_stored
200 BEEP
201 GOSUB Mult_fun !Generate multiplier
202 !functions.
203 BEEP
204 GOSUB Comp_angle !Convolution, envelop detection, doppler correction,
205 !& Arctan for red and blue
206 DISP
207 GOSUB Disp_out
208 GOTO 1
209 !::::::::::::::::::::::::::::::::::::::::::::::::::
210 Range_stored: !
211 !
212 INPUT "RANGE DATA, Type R/r : STORED DATA?, Type S/s",Aa$
213 IF Aa$="R" OR Aa$="r" THEN
214     GOSUB Num_samp
215     !GOSUB Dma_transfer
216     GOSUB Test_data
217     GOSUB File_data
218     GOSUB Buffer_array
219     GOSUB Filt_bw_int
220     RETURN
221 END IF
222 IF Aa$="S" OR Aa$="s" THEN
223     GOSUB File_array
224     GOSUB Buffer_array
225     GOSUB Filt_bw_int
226     RETURN
227 END IF
228 BEEP
229 BEEP
230 BEEP
231 GOTO 213
232 !::::::::::::::::::::::::::::::::::::::::::::::::::
233 Num_samp: !Input the total number of samples
234 !
235 !to be taken. Maximum of 32K
236 BEEP
237 INPUT "TOTAL NUMBER OF SAMPLES TO BE TAKEN?",Num_samples
238 Ns=INT(Num_samples/4)
239 ALLOCATE INTEGER B1(0:Ns),INTEGER B2(0:Ns)

```

```

243 ALLOCATE INTEGER Rotr(0:Ns),Rotb(0:Ns),Qccr(0:Ns),Qccb(0:Ns)
245 ALLOCATE REAL Ta_red(INT(Num_samples/K1)),Ta_bl(INT(Num_samples/K1)),Rot(I
NT(Num_samples/K1))
246 !Create an un-named buffer so greater than
247 !32k can be collected.
248 ASSIGN @Buffer TO BUFFER [Num_samples]
249 RETURN
250 Dma_transfer:
251 CONTROL @Buffer,4;0
252 CONTROL Gpio,0;2
253 CONTROL Gpio,3;0
254 ISET TRANSFER LOW
255 CONTROL Gpio,2;3 !SET CLT0 TO 0, XFER 0
256 BEEP
257 DISP "          WAITING FOR RANGE TRIGGER"
258 TRANSFER @Gpio TO @Buffer;COUNT Num_samples
259 CONTROL Gpio,2;2 !SET CLT0 & CLT1 TO 1
260 CONTROL Gpio,1;3
261 !PRINT "          REGISTER STATUS AFTER DMA"
262 !GOSUB Check_stat
263 BEEP
264 RETURN
265 Filt_bw_int: !Input filter bandwidth in hertz
266 BEEP
267 INPUT "WHAT IS THE DESIRED FILTER BANDWIDTH IN HERTZ?",Fb
268 N=INT(Sf/Fb)
269 PRINT CHR$(12)
270 BEEP
271 INPUT "Real samples f1=12. interpolation factor?",If1
272 BEEP
273 INPUT "Real samples f2=6. interpolation factor?",If2
274 BEEP
275 ALLOCATE REAL Filt_f1(1:K1*If1),Filt_f2(1:K1*If2)
276 RETURN
277 Test: !Generate data - needs to be modified for this
278 !program
279 BEEP
280 INPUT "What is the Envelop Peak Amplitude?",Epa
281 BEEP
282 DISP "          GENERATING DATA"
283 Peak_value_data=0
284 Gal_cyc_frg=200
285 Ab=0
286 ALLOCATE INTEGER Mmm(0:Num_samples-1)
287 FOR X=0 TO Num_samples-1 STEP 4
288   Ac=Epa*SIN((PI*(+Ab))/(6*Gal_cyc_frg)+(2.7)*SIN(2*PI*Ab*Gf/Sf))
289   FOR Y=X TO X+3
290     Mmm(Y)=INT(Ac)+128
291     !PRINT Y,Mmm(Y)
292     OUTPUT @Buffer USING "#,B";Mmm(Y)
293   NEXT Y
294   Ab=Ab+1
295 NEXT X
296 DISP
298 WAIT 5
299 RETURN
300 !::::::::::::::::::::::::::::::::::::::::::::::::::
301 Mult_fun: !Generation of the multiplying function.
302 !
303 !Multiplying function will be (SIN(X)/(X))*COS(Y)
304 !PRINTER IS 705

```

```

305 !ALLOCATE REAL F1(0:(N*If1)),REAL F2(0:(N*If2))
306 DIM F1(0:2000),F2(0:2000)
307 RAD
308 BEEP
309 !Generate 1/2 multiplier function for F1
310 BEEP
311 DISP "Building multiplier function for F1"
312 F1(0)=1
313 !PRINT "F1(0)=";F1(0)
314 FOR P=1 TO N*If1
315 Aa=(PI*P)/(N*If1)
316 Bb=(Aa*2*Gf)/Fb
317 !PRINT "P=";P
318 !PRINT "N=";N*If1
319 !PRINT "(PI*P)/N=";Aa
320 !PRINT "(Aa*2*GF)/FB=";Bb
321 F1(P)=(SIN(Aa)/Aa)*COS(Bb)
322 !PRINT "F1(";P;")=";F1(P)
323 NEXT P
324 DISP "Building multiplier function for F2"
325 !Generate 1/2 of multiplier function for F2
326 F2(0)=1
327 !PRINT "F2(0)=";F2(0)
328 FOR P=1 TO N*If2
329 Aa=(PI*P)/(N*If2)
330 Bb=(Aa*4*Gf)/Fb
331 !PRINT "P=";P
332 !PRINT "N=";N
333 !PRINT "(PI*P)/N=";Aa
334 !PRINT "(Aa*4*GF)/FB=";Bb
335 F2(P)=(SIN(Aa)/Aa)*COS(Bb)
336 !PRINT "F2(";P;")=";F2(P)
337 NEXT P
338 RETURN
339 !
340 !.....
341 !Convolution and envelop detection
342 !at K1 intervals for f1 and f2.
343 !.....
344 !.....
345 Comp_angle: !Filter, envelop detection,
346 !doppler correction and
347 !arctan for red & blue
348 !
349 !
350 !
351 RAD
352 Tot_angle_peak=0
353 P=0 !Initialization of array for the out-
354 !put angle for red and blue lasers.
355 !Determine first data point to start con-
356 !volution considering the value of N and
357 !the need of a + slope for f1 for synchr-
358 !onization.
359 Quotient=(N+1) DIV K1
360 Start=(Quotient+1)*12
361 BEEP
362 DISP " CRUNCHING DATA"
363 PRINT "Start=";Start
364 Qq=0 !Initialize graph index. Increment
365 Qr=Start+2+Sys_del_red !Qq for each 200 computations
366 Qb=Start+2+Sys_del_bl

```



```

372   FOR Xx=1 TO 2 !1=red,2=blue
374   IF Xx=1 THEN
375       Q=Qr
376   ELSE
377       Q=Qb
378   END IF
379   !::::::::::::::::::::::::::::::::::::::::::::::::::
380   Con_f1:!!
381       !Convolution over K1 interval for
382       !f1
383   !
384       J=1
385       Peak_value_f1=0
388       FOR X=Q TO Q+K1-1
392           FOR C=0 TO If1-1
393               Sum_f1=0
394               IF Xx=1 THEN !Xx=1,Red:Xx=2,Blue
395                   FOR M=X TO X+N
396                       Product_f1=Br(M+1)*F1((M-X)*If1-C+If1)+Br(2*X-M)*F1((M-X)*If1+C)
397                       Sum_f1=Sum_f1+Product_f1
398                       ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)*
399                       ! PRINT TAB(50);((M-X)*If1-C+If1);
400                       ! PRINT TAB(60);(2*X-M-1)
401                   NEXT M
402               ELSE
403                   FOR M=X TO X+N
404                       Product_f1=B1(M+1)*F1((M-X)*If1-C+If1)+B1(2*X-M)*F1((M-X)*If1+C)
405                       Sum_f1=Sum_f1+Product_f1
406                       ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)
407                       ! PRINT TAB(50);((M-X)*If1-C+If1);
408                       ! PRINT TAB(60);(2*X-M-1)
409                   NEXT M
410               END IF
411               Filt_f1(J)=Sum_f1
413               !PRINTER IS 705
414               !PRINT "Filt_f1(";J;"=");Filt_f1(J)
417               !PRINTER IS 1
418               IF ABS(Filt_f1(J))>Peak_value_f1 THEN
419                   Peak_value_f1=ABS(Filt_f1(J))
420               END IF
421               J=J+1
422           NEXT C
423       NEXT X
424   !::::::::::::::::::::::::::::::::::::::::::::::::::
425   Env_f1:!!Envelop detection over K1
426       !interval for f1
427   !
428       Diff_plus_f1=0
429       Diff_minus_f1=0
430       FOR E=If1+1 TO 5*If1+1
431           Diff_f1=Filt_f1(E)-Filt_f1(E+INT(If1*K1/2))
432           IF Diff_f1>Diff_plus_f1 THEN
433               Diff_plus_f1=Diff_f1
434           END IF
435           IF Diff_f1<Diff_minus_f1 THEN
436               Diff_minus_f1=Diff_f1
437           END IF
438           !PRINT "e=";E;"Diff_f1=";Diff_f1
439           !PRINT "Diff_plus_f1=";Diff_plus_f1

```

```

440      !PRINT "Diff_minus_f1=";Diff_minus_f1
441      NEXT E
442      !Value of Envelop at P interval for f1
443      IF Diff_plus_f1>ABS(Diff_minus_f1) THEN
444          Envelop_f1=Diff_plus_f1
445      ELSE
446          Envelop_f1=Diff_minus_f1
447      END IF
448      !::::::::::::::::::::::::::::::::::::::::::
449      Con_f2:!
450      !Convolution over K1 interval for F2
451      !-----
452      J=1
453      Peak_value_f2=0
454      FOR X=Q TO Q+K2-1
455          FOR C=0 TO If2-1
456              Sum_f2=0
457              IF X=1 THEN !Xx=1,Red:Xx=2,Blue
458                  FOR M=X TO X+N
459                      Product_f2=Br(M+1)*F2((M-X)*If2-C+If2)+Br(2*X-M)*F2((M-X)*If2+C)
460                      Sum_f2=Sum_f2+Product_f2
461                      ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)
462                      ! PRINT TAB(50);((M-X)*If2-C+If2);
463                      ! PRINT TAB(60);(2*X-M-1)
464                  NEXT M
465              ELSE
466                  FOR M=X TO X+N
467                      Product_f2=Bl(M+1)*F2((M-X)*If2-C+If2)+Bl(2*X-M)*F2((M-X)*If2+C)
468                      Sum_f2=Sum_f2+Product_f2
469                      ! PRINT Q;TAB(10);X;TAB(20);C;TAB(30);M;TAB(40);J;TAB(45);((M-X)
470                      ! PRINT TAB(50);((M-X)*If2-C+If2);
471                      ! PRINT TAB(60);(2*X-M-1)
472                  NEXT M
473              END IF
474              Filt_f2(J)=Sum_f2
475              !PRINTER IS 705
476              !PRINT "Filt_f2(";J;"=");Filt_f2(J)
477              !PRINTER IS 1
478              IF ABS(Filt_f2(J))>Peak_value_f2 THEN
479                  Peak_value_f2=ABS(Filt_f2(J))
480              END IF
481              J=J+1
482          NEXT C
483      NEXT X
484      !::::::::::::::::::::::::::::::::::::::::::
485      Env_f2:!Envelop detection over fract-
486      !ional interval of K1 for f2
487      !using 1/2 cycle difference
488      !-----
489      Diff_plus_f2=0
490      Diff_minus_f2=0
491      FOR E=If2+1 TO 5*If2+1
492          Diff_f2=Filt_f2(E)-Filt_f2(E+3*If2)
493          IF Diff_f2>Diff_plus_f2 THEN
494              Diff_plus_f2=Diff_f2
495          END IF
496          IF Diff_f2<Diff_minus_f2 THEN
497              Diff_minus_f2=Diff_f2
498          END IF

```

```

500     NEXT E
501     !Value of Envelop at P interval for f2
502     IF Diff_plus_f2>ABS(Diff_minus_f2) THEN
503         Envelop_f2=Diff_plus_f2
504     ELSE
505         Envelop_f2=Diff_minus_f2
506     END IF
507     !PRINT "Envelop_f2";Envelop_f2
508     !.....
509     Doppler_cor:
510     !.....
511     !.....
512     !.....
513     !.....
514     ! CORRECTION FOR THE DOPPLER EFFECT FOR RED
515     ! AND BLUE
516     !.....
517     IF ABS(Envelop_f1)>ABS(Envelop_f2) THEN
518         FOR E=If1+1 TO 5*If1+1
519             D2=Filt_f1(E+2)-Filt_f1(E+1)
520             D1=Filt_f1(E+1)-Filt_f1(E)
521             !PRINT "D2=";D2,"D1=";D1,"(D2-D1)=";D2-D1,"E=";E
522             IF D2=0 OR SGN(D1)+SGN(D2)=0 THEN
523                 GOTO 526
524             END IF
525         NEXT E
526         Dsq=D2-D1
527         Ap=Filt_f1(E)
528         Bp=D1-Dsq/2
529         Cp=Dsq/2
530         Envelop_f11=(Ap-(Bp^2)/(4*Cp))
531         !PRINT "Envelop_f11=";Envelop_f11
532         Ee=E-Bp/(2*Cp)
533         !PRINTER IS 705
534         !!!! PRINT "Ee_f1=";Ee
535         PRINTER IS 1
536         Qc=0
537         IF Ee>4*If1+1 THEN Qc=-1
538         IF Ee<2*If1-1 THEN Qc=+1
539     ELSE
540         FOR E=If2+1 TO 5*If2+1
541             D2=Filt_f2(E+2)-Filt_f2(E+1)
542             D1=Filt_f2(E+1)-Filt_f2(E)
543             !PRINT "D2=";D2,"D1=";D1,"(D2-D1)=";D2-D1,"E=";E
544             IF D2=0 OR SGN(D1)+SGN(D2)=0 THEN
545                 GOTO 548
546             END IF
547         NEXT E
548         Dsq=D2-D1
549         Ap=Filt_f2(E)
550         Bp=D1-Dsq/2
551         Cp=Dsq/2
552         Envelop_f22=(Ap-(Bp^2)/(4*Cp))
553         !PRINT "Envelop_f22=";Envelop_f22
554         Ee=E-Bp/(2*Cp)
555         !PRINTER IS 705
556         !!!! PRINT "Ee_f2=";Ee
557         PRINTER IS 1
558         Qc=0
559         IF Ee>4*If2+1 THEN Qc=-1
560         IF Ee<2*If2-1 THEN Qc=+1
561     END IF
562     IF Xx=1 THEN

```

```

563      Ror=Ror+Qc
564      !!!! PRINT "Ror=";Ror,
565      Rotr(P)=Ror
566      Qccr(P)=Qc
567      !PRINTER IS 705
568      !!!! PRINT "Qc=";Qc
569      PRINTER IS 1
570      Qr=Qr+Qc+K1
571      !!!! PRINT "Qr=";Qr
572      ELSE
573      Rob=Rob+Qc
574      !!!! PRINT "Rob=";Rob
575      Rotb(P)=Rob
576      Qccb(P)=Qc
577      !PRINTER IS 705
578      !!!! PRINT "Qc=";Qc
579      PRINTER IS 1
580      Qb=Qb+Qc+K1
581      !!!! PRINT "Qb=";Qb
582      PRINT
583      END IF
584      E_f1=Envelop_f1
585      E_f2=Envelop_f2
586      !!!!PRINT "Envelop_f2/Envelop_f1=";Envelop_f2/Envelop_f1
587      !::::::::::::::::::::::::::::::::::::::::::::::::::
588      Arc_t: !Computation of the angle from the
589      !arctan(Envelop_f2/Envelop_f1) and
590      !counting of the whole quadrants
591      !
592      BEEP 500,,2
593      PRINTER IS 1
594      DEG
595      IF Xx=1 THEN ! Arc_t for Red laser
596      IF E_f1=0 THEN !Take care of divide by zero
597      IF Qpsr=1 OR Qpsr=3 THEN
598      Tar=(Qsr)*90+90
599      GOTO 634
600      ELSE
601      Tar=(Qsr)*90
602      GOTO 634
603      END IF
604      END IF
605      Ra=E_f2/E_f1
606      !PRINTER IS 705
607      !!!! PRINT Ra
608      PRINTER IS 1
609      !Determine Quadrant & Total_angle(P)_
610      IF ABS(Ra)>=60 THEN
611      IF SGN(Ra)=1 THEN
612      Ra=60
613      ELSE
614      Ra=-60
615      END IF
616      END IF
617      IF SGN(Ra)=1 THEN !Plus Ratio
618      IF SGN(E_f1)=1 THEN !Deno plus
619      Qptr=1
620      ELSE
621      Qptr=3
622      END IF
623      Tar=(Qt(Qpsr,Qptr)+Qsr)*90+ATN(Ra)-900

```

```

624     ELSE
625         IF SGN(E_f1)=1 THEN
626             Qptr=4
627         ELSE
628             Qptr=2
629         END IF
630         Tar=(Qt(Qpsr,Qptr)+Qsr)*90+90+ATN(Ra)-900
631     END IF
632     Qsr=Qt(Qpsr,Qptr)+Qsr
633     IF P>1 THEN Qpsr=Qptr
634     !PRINTER IS 705
635     !!!! PRINT P,PROUND(Tar,-2),Qpsr,Qptr,B1(Q),B1(P)
636     PRINTER IS 1
637     PRINT
638     Ta_red(P)=Tar
639     !!!! PRINT P,"Ta_red=";PROUND(Ta_red(P),-2),"E_F1=";E_f1,"E_F2=";E_f2
640     PRINTER IS 1
641     ELSE !Arc_t for blue laser
642     IF E_f1=0 THEN !Take care of divide by zero
643         IF Qpsb=1 OR Qpsb=3 THEN
644             Tab=(Qsb)*90+90
645             GOTO 679
646         ELSE
647             Tab=(Qsb)*90
648             GOTO 679
649         END IF
650     END IF
651     Ra=E_f2/E_f1
652     !Determine Quadrant & Total_angle(P)_
653     IF ABS(Ra)>=60 THEN
654         IF SGN(Ra)=1 THEN
655             Ra=60
656         ELSE
657             Ra=-60
658         END IF
659     END IF
660     IF SGN(Ra)=1 THEN !Plus Ratio
661         IF SGN(E_f1)=1 THEN !Deno plus
662             Qptb=1
663         ELSE
664             Qptb=3
665         END IF
666         Tab=(Qt(Qpsb,Qptb)+Qsb)*90+ATN(Ra)-900
667     ELSE
668         IF SGN(E_f1)=1 THEN
669             Qptb=4
670         ELSE
671             Qptb=2
672         END IF
673         Tab=(Qt(Qpsb,Qptb)+Qsb)*90+90+ATN(Ra)-900
674     END IF
675     Qsb=Qt(Qpsb,Qptb)+Qsb
676     IF P>1 THEN Qpsb=Qptb
677     !PRINTER IS 705
678     PRINT
679     !!!! PRINT P,PROUND(Tab,-2),Qpsb,Qptb,B1(Q),B1(P)
680     PRINTER IS 1
681     Ta_bl(P)=Tab
682     !!!! PRINT P,"Ta_bl=";PROUND(Ta_bl(P),-2),"E_F1=";E_f1,"E_F2=";E_f2
683     END IF !For red or blue
684     PRINTER IS 1

```

```

689 RAD
690 NEXT Xx
691 RAD
692 PRINT
693 !This completes the angle computation
694 !for the red and blue for this K1 interval P
695 P=P+1
696 PRINT P
698 !Test to increment Qq or not
699 IF P=0 THEN
700 Qq=0
701 ELSE
702 Rmdr=P MOD 199 !P starts at 0
703 IF Rmdr=0 THEN Qq=Qq+1
704 END IF
705 !Q>Ns-48, either red/blue will exit to Main
706 IF Q>Ns-48 THEN RETURN
707 GOTO 372
708 Disp_out: !Graph data
709 DUMP DEVICE IS 705
710 PRINTER IS 705
711 PEN 2
712 !!PRINT CHR$(12); !Clear alpha & form feed
713 DISP
714 GINIT !Initialize
715 GRAPHICS ON !Raster on
716 FOR Qqq=0 TO Qq !Graph index
717 FOR R=1 TO 2 !red=1, blue=2
718 PEN 2
719 WINDOW -7.50,-500,500 !Set window
720 AXES 10,100
721 LOG 6
722 FOR I=0 TO 50 STEP 10
723 MOVE I,0
724 LABEL Qqq*200+I
725 NEXT I
726 !
727 LOG 8
728 FOR I=-400 TO 400 STEP 100
729 MOVE 0,I
730 LABEL I
731 NEXT I
732 !
733 LOG 5
734 MOVE 25,475
735 PEN 6
736 LABEL "ANGLE AS A FUNCTION OF ENVELOP SAMPLES"
737 MOVE 18,425
738 IF R=1 THEN
739 PEN 2
740 LABEL "RED"
741 MOVE 18,-400
742 LABEL Name$
743 PEN 6
744 END IF
745 IF R=2 THEN LABEL "BLUE"
746 MOVE 18,-400
747 LABEL Name$
748 MOVE -4,0
749 LOG 6
750

```

```

751 LABEL "Samp"
752 LONG 5
753 MOVE 25,-475
754 LABEL Title$:"ENVELOP SAMPLE RATE, 3KHZ"
755 PEN 0
756 MOVE 0,0
757 PEN 2
758 FOR X=0 TO 50 !graph data
759 IF R=1 THEN
760 Tb=Ta_red(Qqq*200+X+4)-Ta_red(4)
761 ELSE
762 Tb=Ta_bl(Qqq*200+X+4)-Ta_bl(4)
763 END IF
764 DRAW X,Tb
765 NEXT X
766 PEN 0
767 MOVE 0,0
768 PEN 0
769 PAUSE
770 !DUMP GRAPHICS
771 !PRINT CHR$(12)
772 GCLEAR
773 NEXT R
774 NEXT Qqq
775 ALPHA ON
776 PRINTER IS 1
777 DISP "PAUSE"
778 PAUSE
779 RETURN
780 !::::::::::::::::::::::::::::::::::::::::::
781 Err_rout: !Routines for error recovery
782 !
783 BEEP
784 BEEP
785 BEEP
786 DISP ERRMS
787 PAUSE
788 GOSUB Disp_out
789 PAUSE
790 !::::::::::::::::::::::::::::::::::::::::::
791 Buffer_array: !Transfer of data from buffer to array.
792 !
793 PRINT CHR$(12)
794 DISP "TRANSFERRING DATA FROM BUFFER TO 4 ARRAYS"
795 CONTROL @Buffer,5;1
796 FOR X=1 TO Ns-1
797 ENTER @Buffer USING "#.B";Zzz
798 Br(X-1)=Zzz-128
799 ENTER @Buffer USING "#.B";Zzz
800 B1(X-1)=Zzz-128
801 ENTER @Buffer USING "#.B";Zzz
802 B1(X-1)=Zzz-128
803 ENTER @Buffer USING "#.B";Zzz
804 B2(X-1)=Zzz-128
805 !PRINT X-1,Br(X-1),B1(X-1),B1(X-1),B2(X-1)
806 NEXT X
807 DISP "TRANSFER COMPLETE"
808 WAIT 2
809 PRINT CHR$(12)
810 RETURN
811 !::::::::::::::::::::::::::::::::::::::::::

```

```

812 File_data: !Transfer of data from Buffer
813             !to disk
814 !
815     BEEP
816     DISP "INSERT A FORMATTED DATA DISK! - PRESS CONT TO CONT"
817     PAUSE
818     INPUT "NAME OF DATA FILE TO BE CREATED?".Name$
819     IF Name$="" THEN 818
820     BEEP
821     DISP "CREATING A DATA FILE"
822     !Record length is 1
823     CREATE BDAT Name$":INTERNAL",Num_samples,1
824     !Assign an I/O path
825     ASSIGN @File TO Name$":INTERNAL"
826     PRINT CHR$(12)
827     DISP "TRANSFERRING DATA FROM BUFFER TO FILE"
828     TRANSFER @Buffer TO @File;COUNT Num_samples
829     WAIT 5
830     DISP "TRANSFER COMPLETE"
831     WAIT 5
832     DISP
833     RETURN
834 !::::::::::::::::::::::::::::::::::::::::::::::::::
835 File_array: !Transfer from file to array
836 !
837     BEEP
838     PRINT CHR$(12)
839     DISP "INSERT DATA DISK -- PRESS CONT TO CONT"
840     PAUSE
841     CAT
842     BEEP
843     INPUT "WHAT FILE?".Name$
844     INPUT "Rec/File for file?".Num_samples
845     GOSUB 240
846     ASSIGN @File TO Name$":INTERNAL"
847     !Reset file pointer to 1
848     CONTROL @File,5;1
849     !Reset buffer pointer to 1
850     CONTROL @Buffer,5;1
851     DISP "TRANSFERRING FROM DATA FILE TO BUFFER ARRAY"
852     TRANSFER @File TO @Buffer;COUNT Num_samples
853     DISP "TRANSFER COMPLETED"
854     !Reset buffer pointer to 1
855     CONTROL @Buffer,5;1
856     !Close @file
857     ASSIGN @File TO *
858     WAIT 2
859     PRINT CHR$(12)
860     DISP
861     RETURN
862 !::::::::::::::::::::::::::::::::::::::::::::::::::
863 A_to_d_reset: !
864 !
865     !RESET DATA TRANSFER ELECTRONICS
866     CONTROL Gpio,2;2 !SET CTLO & CTL1 TO 0
867     CONTROL Gpio,2;0 !SET CTLO TO 1 & CTL1 TO 0
868     CONTROL Gpio,2;2 !SET CTLO & CTL1 TO 1
869     !END RESET
870     RETURN
871 !::::::::::::::::::::::::::::::::::::::::::::::::::
1180 Test_data: ! Generate test data

```



```

1183
1184      !PRINTER IS 705
1185      PRINT CHR$(12)
1186      DISP
1187      BEEP
1188      INPUT "What is the Envelop Peak Amplitude?".Epa
1189      BEEP
1190      DISP "                GENERATING DATA"
1191      Peak_value_data=0
1192      Gal_cyc_frg=200
1193      Ab=0
1194      ALLOCATE INTEGER Mmm(0:Ns*4-1)
1195      FOR X=0 TO Ns*4-1 STEP 4
1196          Ac=Epa*SIN((PI*(1+Ab))/(6*Gal_cyc_frg)+2.7*SIN(2*PI*Ab*Gf/S))
1197          FOR Y=X TO X+3
1198              Mmm(Y)=INT(Ac)+128
1199              !PRINT Y,Mmm(Y)
1200              OUTPUT @Buffer USING "e,B";Mmm(Y)
1201          NEXT Y
1202          Ab=Ab+1
1203      NEXT X
1204      DISP
1205      RETURN
1206      END

```

Cross Reference ****

| Numerical Variables | | | | | | | | | |
|---------------------|-----|-----|------|------|------|------|-----|-----|---|
| h_to_d_reset | | | | | | | | | |
| ha | 315 | 316 | 321 | 329 | 330 | 335 | | | |
| ht | 285 | 288 | 294 | 1193 | 1196 | 1204 | | | |
| h | 288 | 290 | 1196 | 1200 | | | | | |
| h | 527 | 530 | 549 | 552 | | | | | |
| h | 242 | 802 | | | | | | | |
| h_pea | | | | | | | | | |
| h | 242 | 804 | | | | | | | |
| h_pea | | | | | | | | | |
| ht | 316 | 321 | 330 | 335 | | | | | |
| hbt | | | | | | | | | |
| hbt | | | | | | | | | |
| h | 241 | 404 | 467 | 800 | | | | | |
| h | 526 | 530 | 532 | 550 | 552 | 554 | | | |
| h | 241 | 396 | 459 | 798 | | | | | |
| butter_array | | | | | | | | | |
| bu_label | | | | | | | | | |
| Check_stat | 392 | 396 | 404 | 422 | 455 | 459 | 467 | 482 | |
| Check_stat_r | | | | | | | | | |
| Comp_angle | | | | | | | | | |
| con_f2 | | | | | | | | | |
| cp | 529 | 530 | 532 | 551 | 552 | 554 | | | |
| csbf1 | | | | | | | | | |
| csbf2 | | | | | | | | | |
| csrf1 | | | | | | | | | |
| csrf2 | | | | | | | | | |
| d1 | 520 | 522 | 526 | 528 | 542 | 544 | 548 | 550 | |
| d2 | 519 | 522 | 526 | 541 | 544 | 546 | | | |
| Diff_f1 | 431 | 432 | 433 | 435 | 436 | | | | |
| Diff_f2 | 493 | 494 | 495 | 497 | 498 | | | | |
| Diff_minus_f1 | 429 | 435 | 436 | 443 | 446 | | | | |
| Diff_minus_f2 | 490 | 497 | 498 | 502 | 505 | | | | |
| Diff_plus_f1 | 428 | 432 | 433 | 443 | 444 | | | | |
| Diff_plus_f2 | 489 | 494 | 495 | 502 | 503 | | | | |
| Disp_con | | | | | | | | | |
| Disp_data | | | | | | | | | |
| Disp_data_1 | | | | | | | | | |
| Disp_data_2 | | | | | | | | | |
| Disp_out | | | | | | | | | |
| Dma_transfer | | | | | | | | | |
| Dpcb | | | | | | | | | |
| Dpcr | | | | | | | | | |
| ds | 550 | | | | | | | | |
| dsq | 526 | 528 | 529 | 540 | 551 | | | | |
| e | 430 | 431 | 441 | 492 | 493 | 500 | 518 | 519 | 5 |
| | 527 | 532 | 540 | 541 | 542 | 547 | 549 | 554 | 6 |
| e_f1 | 584 | 596 | 605 | 618 | 625 | 644 | 653 | 663 | 6 |
| e_f2 | 585 | 605 | 653 | | | | | | |
| ee | 532 | 537 | 538 | 554 | 559 | 560 | | | |
| Endcat | | | | | | | | | |
| Envelop_f1 | 444 | 446 | 517 | 584 | | | | | |
| Envelop_f11 | 530 | | | | | | | | |
| Envelop_f2 | 503 | 505 | 517 | 585 | | | | | |

AD-A179 742

LASER BALLISTIC SENSOR DEVELOPMENT(U) BOEING AEROSPACE
CO SEATTLE WA C R POND ET AL JAN 87 BRL-CR-563
DAAK11-84-C-0095

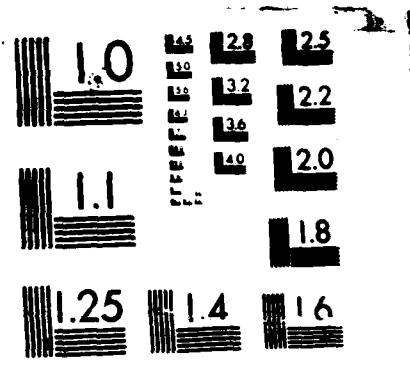
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NL





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS 1963 A

| | | | | | | | | | | |
|-----------------|-----|-------|------|------|------|------|------|------|-----|-----|
| Envelop_f22 | 552 | | | | | | | | | |
| Epa | 280 | 288 | 1188 | 1196 | | | | | | |
| Err_rout | | | | | | | | | | |
| F1 | 306 | <-DEF | 312 | 321 | 396 | 404 | | | | |
| F2 | 306 | <-DEF | 326 | 335 | 459 | 467 | | | | |
| Fb | 267 | 268 | 316 | 330 | | | | | | |
| File_array | | | | | | | | | | |
| File_data | | | | | | | | | | |
| Filt_bw_int | | | | | | | | | | |
| Filt_f1 | 275 | 411 | 418 | 419 | 431 | 519 | 520 | 527 | | |
| Filt_f11b | | | | | | | | | | |
| Filt_f11r | | | | | | | | | | |
| Filt_f1b | | | | | | | | | | |
| Filt_f1r | | | | | | | | | | |
| Filt_f2 | 275 | 474 | 478 | 479 | 493 | 541 | 542 | 549 | | |
| Filt_f22b | | | | | | | | | | |
| Filt_f22r | | | | | | | | | | |
| Filt_f2b | | | | | | | | | | |
| Filt_f2r | | | | | | | | | | |
| Gal_cyc_frg | 284 | 288 | 1192 | 1196 | | | | | | |
| Gf | 127 | 288 | 316 | 330 | 1196 | | | | | |
| Gpio | 119 | 252 | 253 | 255 | 259 | 260 | 866 | 867 | 868 | |
| Hpiib | 120 | | | | | | | | | |
| I | 722 | 723 | 724 | 725 | 728 | 729 | 730 | 731 | | |
| If1 | 271 | 275 | 314 | 315 | 392 | 396 | 404 | 430 | 431 | 518 |
| | 537 | 538 | | | | | | | | |
| If2 | 273 | 275 | 328 | 329 | 455 | 459 | 467 | 492 | 493 | 540 |
| | 559 | 560 | | | | | | | | |
| J | 384 | 411 | 418 | 419 | 421 | 452 | 474 | 478 | 479 | 481 |
| K1 | 128 | 245 | 275 | 360 | 388 | 431 | 570 | 580 | | |
| K2 | 129 | 454 | | | | | | | | |
| K3 | 130 | | | | | | | | | |
| Kk | | | | | | | | | | |
| M | 395 | 396 | 401 | 403 | 404 | 409 | 458 | 459 | 464 | 466 |
| | 467 | 472 | | | | | | | | |
| Max_x_axes | | | | | | | | | | |
| Mmm | 286 | 290 | 292 | 1194 | 1200 | 1202 | | | | |
| Mn | | | | | | | | | | |
| Mult_fun | | | | | | | | | | |
| N | 268 | 314 | 315 | 328 | 329 | 360 | 395 | 403 | 458 | 466 |
| Nnn | | | | | | | | | | |
| Ns | 240 | 241 | 242 | 243 | 706 | 796 | 1194 | 1195 | | |
| Num_samp | | | | | | | | | | |
| Num_samples | 239 | 240 | 245 | 248 | 258 | 286 | 287 | 823 | 828 | 844 |
| | 852 | | | | | | | | | |
| Op | | | | | | | | | | |
| | 314 | 315 | 321 | 323 | 328 | 329 | 335 | 337 | 354 | 565 |
| | 566 | 575 | 576 | 633 | 639 | 678 | 685 | 695 | 696 | 699 |
| | 702 | | | | | | | | | |
| Peak_value_data | 283 | 1191 | | | | | | | | |
| Peak_value_f1 | 385 | 418 | 419 | | | | | | | |
| Peak_value_f11b | | | | | | | | | | |
| Peak_value_f11r | | | | | | | | | | |
| Peak_value_f1b | | | | | | | | | | |
| Peak_value_f1r | | | | | | | | | | |
| Peak_value_f2 | 453 | 478 | 479 | | | | | | | |
| Peak_value_f22b | | | | | | | | | | |
| Peak_value_f22r | | | | | | | | | | |
| Peak_value_f2b | | | | | | | | | | |
| Peak_value_f2r | | | | | | | | | | |
| Product_f1 | 396 | 397 | 404 | 405 | | | | | | |

| | | | | | | | | | | |
|--------------------|-----|-----|-----|------|------|------|------|------|------|-----|
| Product_f2 | 459 | 460 | 467 | 468 | | | | | | |
| Q | 375 | 377 | 388 | 454 | 706 | | | | | |
| Qb | 371 | 377 | 580 | | | | | | | |
| Qc | 536 | 537 | 538 | 558 | 559 | 560 | 563 | 566 | 570 | 573 |
| | 576 | 580 | | | | | | | | |
| Qccb | 243 | 576 | | | | | | | | |
| Qccr | 243 | 566 | | | | | | | | |
| Qj | | | | | | | | | | |
| Qpsb | 645 | 668 | 675 | 677 | 678 | | | | | |
| Qpsr | 597 | 623 | 630 | 632 | 633 | | | | | |
| Qptb | 664 | 666 | 668 | 671 | 673 | 675 | 677 | 678 | | |
| Qptr | 619 | 621 | 623 | 626 | 628 | 630 | 632 | 633 | | |
| Qq | 365 | 700 | 703 | 716 | | | | | | |
| Qqq | 716 | 724 | 760 | 762 | 774 | | | | | |
| Qr | 370 | 375 | 570 | | | | | | | |
| Qsb | 190 | 646 | 649 | 668 | 675 | 677 | | | | |
| Qsr | 189 | 598 | 601 | 623 | 630 | 632 | | | | |
| Qt | 180 | DEF | 181 | 182 | 183 | 184 | 185 | 186 | 187 | 188 |
| | 623 | 630 | 632 | 668 | 675 | 677 | | | | |
| Quotient | 360 | 361 | | | | | | | | |
| R | 717 | 738 | 745 | 759 | 773 | | | | | |
| Ra | 605 | 610 | 611 | 612 | 614 | 617 | 623 | 630 | 653 | 655 |
| | 656 | 657 | 659 | 662 | 668 | 675 | | | | |
| Range_stored | | | | | | | | | | |
| Rmdr | 702 | 703 | | | | | | | | |
| Rob | 573 | 575 | | | | | | | | |
| Ror | 563 | 565 | | | | | | | | |
| Rot | 245 | | | | | | | | | |
| Rotb | 243 | 575 | | | | | | | | |
| Rotr | 243 | 565 | | | | | | | | |
| S | | | | | | | | | | |
| Sf | 126 | 268 | 288 | 1196 | | | | | | |
| Ssp | | | | | | | | | | |
| Start | 361 | 364 | 370 | 371 | | | | | | |
| Sum_f1 | 393 | 397 | 405 | 411 | | | | | | |
| Sum_f2 | 456 | 460 | 468 | 474 | | | | | | |
| Sys_del_bl | 134 | 371 | | | | | | | | |
| Sys_del_red | 131 | 370 | | | | | | | | |
| Ta_bl | 245 | 685 | 762 | | | | | | | |
| Ta_red | 245 | 639 | 760 | | | | | | | |
| Tab | 646 | 649 | 668 | 675 | 685 | | | | | |
| Tar | 598 | 601 | 623 | 630 | 639 | | | | | |
| Tb | 760 | 762 | 764 | | | | | | | |
| Test_data | | | | | | | | | | |
| Tic_m_x | | | | | | | | | | |
| Tot_angle_peak | 353 | | | | | | | | | |
| Tt | | | | | | | | | | |
| X | 287 | 289 | 295 | 388 | 395 | 396 | 403 | 404 | 423 | 454 |
| | 458 | 459 | 466 | 467 | 483 | 758 | 760 | 762 | 764 | 765 |
| | 796 | 798 | 800 | 802 | 804 | 806 | 1195 | 1199 | 1205 | |
| Xx | 372 | 374 | 394 | 457 | 562 | 595 | 690 | | | |
| Y | 289 | 290 | 292 | 293 | 1199 | 1200 | 1202 | 1203 | | |
| Z | | | | | | | | | | |
| Zzz | 797 | 798 | 799 | 800 | 801 | 802 | 803 | 804 | | |
| + String Variables | | | | | | | | | | |
| Aa\$ | 213 | 214 | 224 | | | | | | | |
| Names | 742 | 747 | 818 | 819 | 823 | 825 | 843 | 846 | | |
| Title\$ | 754 | | | | | | | | | |
| + I/O Path Names | | | | | | | | | | |

| | | | | | | | | | | |
|---------|-----|-----|-----|------|-----|-----|-----|-----|-----|-----|
| @Buf | | | | | | | | | | |
| @Buffer | 248 | 251 | 258 | 292 | 795 | 797 | 799 | 801 | 803 | 828 |
| | 850 | 852 | 855 | 1202 | | | | | | |
| @File | 825 | 828 | 846 | 848 | 852 | 857 | | | | |
| @Gpio | 191 | 258 | | | | | | | | |

* Line Labels

| | | | | |
|--------------|-----|-------|-------|-----------|
| A_to_d_reset | 198 | 863 | <-DEF | |
| Arc_t | 588 | <-DEF | | |
| Buffer_array | 220 | 226 | 791 | <-DEF |
| Bw_label | | | | |
| Check_stat | | | | |
| Check_stat_r | | | | |
| Comments | 11 | <-DEF | | |
| Comp_angle | 204 | 345 | <-DEF | |
| Con_f1 | 380 | <-DEF | | |
| Con_f2 | 449 | <-DEF | | |
| Constants | 123 | <-DEF | | |
| Disp_con | | | | |
| Disp_data | | | | |
| Disp_data_1 | | | | |
| Disp_data_2 | | | | |
| Disp_out | 117 | 207 | 708 | <-DEF 788 |
| Dma_transfer | 250 | <-DEF | | |
| Doppler_cor | 511 | <-DEF | | |
| Env_f1 | 425 | <-DEF | | |
| Env_f2 | 485 | <-DEF | | |
| Err_rout | 113 | 781 | <-DEF | |
| File_array | 225 | 835 | <-DEF | |
| File_data | 219 | 812 | <-DEF | |
| Filt_bw_int | 221 | 227 | 265 | <-DEF |
| Index | 81 | <-DEF | | |
| Initialize | 109 | <-DEF | | |
| Main | 195 | <-DEF | | |
| Mult_fun | 201 | 301 | <-DEF | |
| Num_samp | 215 | 235 | <-DEF | |
| Range_stored | 199 | 210 | <-DEF | |
| Test | 277 | <-DEF | | |
| Test_data | 217 | 1180 | <-DEF | |
| Var_list | 139 | <-DEF | | |

* Line Numbers

| | |
|-----|-----|
| 1 | 208 |
| 4 | |
| 5 | |
| 9 | |
| 11 | |
| 13 | |
| 17 | |
| 24 | |
| 157 | |
| 174 | |
| 180 | |
| 188 | |
| 194 | |
| 196 | |
| 202 | |
| 213 | 233 |
| 214 | |
| 220 | |
| 221 | |

| | | |
|------|-----|-----|
| 226 | | |
| 240 | 845 | |
| 245 | | |
| 249 | | |
| 363 | | |
| 372 | 707 | |
| 390 | | |
| 524 | | |
| 526 | 523 | |
| 540 | | |
| 546 | | |
| 548 | 545 | |
| 562 | | |
| 571 | | |
| 578 | | |
| 600 | | |
| 626 | | |
| 634 | 599 | 602 |
| 635 | | |
| 648 | | |
| 679 | 647 | 650 |
| 686 | | |
| 691 | | |
| 716 | | |
| 729 | | |
| 736 | | |
| 763 | | |
| 816 | | |
| 818 | 819 | |
| 820 | | |
| 827 | | |
| 830 | | |
| 858 | | |
| 884 | | |
| 895 | | |
| 902 | | |
| 907 | | |
| 911 | | |
| 951 | | |
| 1254 | | |
| 1274 | | |

* SUB Subprograms
Endcat

Unused entries = 3

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